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REQUIREMENTS STUDY FOR A
BIOTECHNOLOGY LABORATORY FOR
MANNED EARTH ORBITING MISSIONS

Volume I - Results, Recommendations and Requirements

by L. T. Kail

Prepared by
MCDONNELL DOUGLAS CORPORATION
Huntington Beach, Calif.
for



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION - WASHINGTON, D. C. - AUGUST 1969

REQUIREMENTS STUDY FOR A BIOTECHNOLOGY LABORATORY FOR MANNED EARTH ORBITING MISSIONS

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for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

PREFACE

This document contains the results of a study of the requirements for a biotechnology laboratory for manned earth orbiting missions in the 1970's. The results are presented in two volumes. Volume I contains descriptions of the life sciences research programs and the associated laboratory equipment requirements. The objectives and corresponding information needs of these research programs were prepared on the basis of the results of past studies and in collaboration with the NASA and the scientific community. While these objectives are not to be construed as an official statement by NASA of life sciences space research goals in the 1970's, they do represent a consensus developed under the cognizance of NASA OART, OSSA, and OMSF*. These goals served as a reliable basis for the selection of representative experiments in each research category and the subsequent establishment of laboratory equipment requirements. Volume II contains brief descriptions of the experiments upon which the requirements analysis was based.

The study has resulted in the identification of more than 100 major items of laboratory equipment which will be required in manned Earth-orbiting laboratory life science research programs. Many of these items were found to have multiple applications, while others are required only by a single experiment in a narrow research subcategory. In addition, an examination of the availability of the equipment showed that development programs of some magnitude will be required before the equipment

^{*}National Aeronautics and Space Administration Office of Advanced Research and Technology, Office of Space Science Applications, and Office of Manned Space Flight.

will be suitable for space applications. The results of this study provide information to support the planning of the necessary equipment development programs as well as the life science research programs for individual missions throughout the 1970's.

The study was performed by the McDonnell Douglas Astronautics Company—Western Division (MDAC-WD) for the Biotechnology and Human Research Division of the Office of Advanced Research and Technology (OART). The work began in January, 1968 and was performed in accordance with contract number NAS7-518.

The work on this study was performed by MDAC-WD personnel from two departments: Advance Biotechnology and Power Systems, Dr. K. H. Houghton, Chief Engineer; and Advance Space Stations, F. C. Runge, Program Manager. The study was managed for NASA (OART) by R. W. Dunning, and for MDAC-WD by L. T. Kail.

Throughout the study, the MDAC-WD study team worked closely with NASA Headquarters personnel. The study team members and their NASA coordinators were:

Study Team	NASA Coordinator
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Biomedicine - Dr. W. F. Arndt, Jr.	Dr. E. J. McLaughlin (MM)
Behavior - J. D. Brower	Dr. S. Deutsch (RBM)
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Other MDAC-WD personnel who made significant contributions were J.S. Seeman in the area of Behavioral Research and M.M. Yakut in the area of Life Support and Protective Systems Research.

Questions or comments concerning this report will be welcomed by either of the individuals listed below:

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Section 1 INTRODUCTION

NASA has performed or sponsored many studies, the objective of which was the definition of life sciences research that must be performed before man can venture confidently into space. Other studies have explored that research which could take advantage of the unique weightless environment available in space missions to increase our understanding of life processes. This activity has included many studies in the various fields of life science and has resulted in the identification of various research objectives and the equipment needs of associated experiments. An examination of the results of these studies has indicated that extensive research and development would be required to provide the laboratory equipment needed to perform such life science research in Earth-orbiting laboratories. Since the individual studies performed to date have been limited to either a single life science field, or a part of that field, a need arose to develop a plan for an integrated program which includes all of the required research and takes maximum advantage of equipment commonality. The objective of the study presented here was to expand and incorporate the results of past studies, into an all-encompassing life science research program (gap-filling as required) to identify manned Earth-orbital life science laboratory equipment requirements for the 1970's. Another objective of the study was to identify and integrate requirements that are common to the four life science research fields of biomedicine and human research, behavior, bioscience, and life support systems.

To orient the reader, an elementary sketch of a concept for a biotechnology laboratory as part of a multipurpose space station is shown in Figure 1-1. The individual equipment items required in the "laboratory area" are described. The centrifuge shown relates to requirements of experiments considered in this report and represents another program currently under study by NASA for life sciences research. This study has addressed itself to the total research activities and equipment encompassed by such a laboratory. The results of the study provide a basis for further definition of a facility.

SPACE STATION BIOTECH LAB CONCEPT BIOMEDICINE - BEHAVIORAL BIOSCIENCE - LIFE SUPPORT

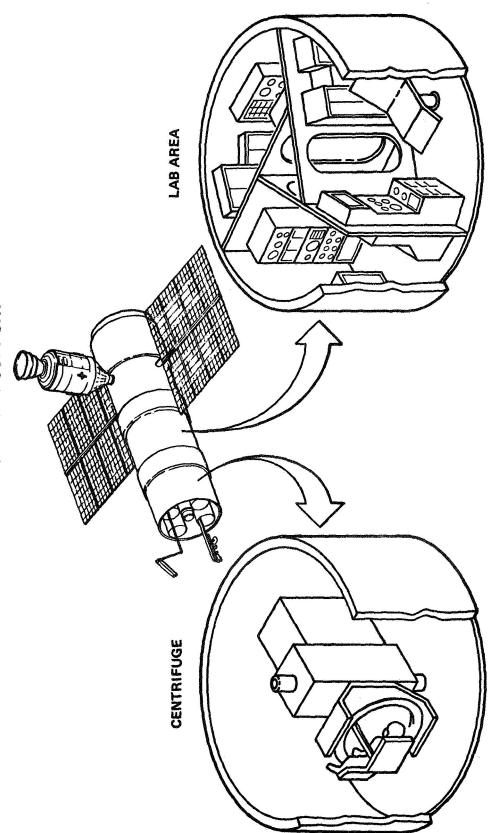


Figure 1-1. Space Station Biotech Lab Concept

In the preparation of this report MDAC has endeavored to provide a document which will be used as a reliable reference by personnel engaged in life sciences activities in space in particular, and by those involved in space program planning in general.

Program managers and mission planners are referred to the biotechnology research goals and the program recommendations in Section 3. The research and development requirements and the program planning suggestions in this section will be of particular value.

Research scientists who are interested in the detailed description of the research programs, and the methods by which they were assembled will find such information in Section 4. The manner in which a special field of research fits into the total research goals is also presented. The measurements associated with the experiments in each research subcategory are also discussed. Further details concerning each of the experiments is presented in the experiment summaries in Volume II.

Space laboratory design engineers and others concerned with planning and studying the accommodations of life sciences experiment programs in such laboratories will find complete lists of the required experiment equipment in the matrix charts in Section 5. The commonality of equipment which is graphically displayed in these charts will permit experiment program planners to modify the experiment programs to take advantage of equipment which has broad application. The power, weight, and volume requirements of each item of equipment, as far as they could be determined, are summarized in Section 7.

Supporting research and technology planners are referred to Section 7, in which each item of equipment requiring development is listed. Current development programs which include these items are identified where applicable. The matrix charts, showing the commonality of equipment throughout the experiment programs will assist in the selection of equipment which requires development emphasis.

The space station for the mid 1970's, in which a large portion of the research outlined herein will be performed, is currently in the candidate configurations study phase within NASA. Therefore, this document will provide a coordinated (OART, OMSF, OSSA)* reference for the forthcoming Phase B spacestation studies and will ensure the proper integration and support of the biotechnology laboratory in the space-station configuration.

^{*}Office of Advanced Research and Technology; Office of Manned Space Flight; Office of Space Science and Applications.

Section 2 STUDY APPROACH

The objectives of this study were (1) to correlate space-based life sciences research goals for the next decade and (2) to identify the laboratory requirements for conducting, in a manned orbiting space station, the experiments which are representative of that research.

In reaching these objectives, MDAC followed a modified version of the "top-down analysis" approach which was successfully used by the Company in recent NASA studies in astronomy, oceanography, and meteorology. The initial step in these previous studies involved the identification of the overall research objectives. This was followed by an examination of the objectives to define those which require or use space-based facilities. The final step was the development of an orbital experiment program which included the required instrumentation, facilities, and operational support.

The first objective of this study was accomplished by performing the first two steps during which we relied to a considerable extent on past NASA and contractor work. Since the reports dealt almost exclusively with space-oriented research, it was not necessary to review research objectives which were not space related. Further, the development of a representative orbital experiment program was accomplished by selecting, again from existing literature, the experiments which were considered to be responsive to the information needs of each research category. The analysis also indicated research requirements for which there were no proposed experiments. In these instances, the study team prepared possible experiments so that the required measurement equipment would not be overlooked.

In reaching the second objective of the study, MDAC analyzed the selected experiments to identify the necessary measurements, laboratory equipment, and support equipment and to ensure that the equipment requirements were indeed representative and responsive to the research needs.

An Experiment Requirements Summary was prepared for each experiment analyzed. Matrix charts which illustrate the commonality of measurements and the commonality of equipment were prepared for the representative experiments in each of the four research areas of biomedicine, behavior, bioscience and life support, and protective systems.

Each major item of required laboratory equipment was described briefly. In cases where equipment designed for space use could not be identified, the study team listed ground-based equipment in order to indicate the nature of the requirement.

At the end of the study, the MDAC study team made suggestions as to how the identified research programs might be carried out. Because the making of these suggestions involved the assignment of research priorities, the study team realizes that these suggestions are open to criticism. However, the intent of these recommendations is to provide a reference upon which future research in manned space missions may be based.

In summary, the approach to the study was based on four basic tasks:

- 1. Extracting research goals from existing life-science literature.
- 2. Structuring research programs.
- 3. Selecting experiments which are representative of these programs.
- 4. Identifying laboratory equipment required by the representative experiments.

The following sections present the detailed output of this effort.

Section 3 RESULTS AND RECOMMENDATIONS

The life sciences research program evolving from this study is summarized in this section to provide an overview of the total program scope before proceeding to the details of the individual areas (Section 4). In each individual area, consideration of the desirable sequence for performing this research has led to recommendations concerning the implementation of the biotechnology laboratory program. These recommendations are discussed in the second half of this section.

3.1 LIFE SCIENCES RESEARCH PROGRAM SUMMARIES

Briefly, the Life Sciences research goals defined for manned space missions are as follows:

- 1. Obtain an understanding of man's reaction to long-term exposure to the space environment.
- 2. Study the effects of the space environment on living organisms and increase our understanding of life on Earth.
- 3. Advance the technology for support of man in extended exposure to the space environment.

Research necessary to achieve the first goal is discussed under the separate headings "Biomedical and Human Research" and "Behavioral Research". Despite the intimate relationship between these two areas, they were treated separately to provide equal emphasis during the study and to permit ease of reference by specialists in each area. Under "Bioscience Research," the activities necessary to achieve the second goal are described. In "Life Support and Protective Systems Research," the research necessary to achieve the third goal is described.

The research programs prepared in this study form a firm base for planning space research in life sciences for at least the next decade. Results from the experiments described should answer most current pressing questions and raise others which, in turn, will give rise to further research.

There is no doubt that one of the most important current questions concern man's reaction to long periods in space; therefore, research in the areas of biomedical and behavioral research will receive the highest priority in mission assignment. Bioscience research which will study life processes in the weightless environment of an Earth-orbiting laboratory holds significant potential value in contributing to a fuller understanding of those life processes on Earth. The development of effective life support systems is, of course, the key factor in all space research programs involving man. This development requires not only system and component testing but also research in the associated basic technology. The ability of the crew to maintain this "hardware" system in the space environment, the reaction of the crew to the atmosphere provided, and the analysis of microbial products which may develop in the space station ecology require the facilities of an integrated laboratory if the life support system research is to be effective.

It is recommended that the biotechnology laboratory provide facilities for research from each of the areas of biomedicine, behavior, bioscience, and life support and protective systems.

3.1.1 Biomedicine and Human Research Program Summary

The biomedicine and human research program presented in this report is composed of 55 representative experiments on both human and animal subjects. The established program goal is to qualify man for long-duration space flight. The specific objectives within this goal are those stated by NASA and discussed in Section 4. By using these objectives as guidelines, it was possible to organize critical research areas or "information needs" which were identified in a variety of previous studies into 10 different functional body systems. Experiments which appeared to satisty the information needs were then selected and reviewed, and then were consolidated to achieve economy of operation. In a few cases where no experiments were located in the literature, protocols were specifically designed to fulfill the needs. Finally, each experiment was analyzed to determine the required measurements and equipment. Table 3-1 presents the basic organization of this biomedical program and shows the distribution of experiments.

was analyzed to determine the required measurements and equipment.

Table 3-1 presents the basic organization of this biomedical program and shows the distribution of experiments.

Table 3-1
BIOMEDICAL-HUMAN RESEARCH STUDY PROGRAM OUTLINE

			ber of iments Animal
Neurologic function	Vestibular function Circadian rhythms	2	1 1
Cardiovascular function	Cardiovascular deconditioning Deconditioning countermeasures Homeostatic mechanisms Blood volume Verification of measurement techniques	5 3 2 1	- - 1 -
Pulmonary function	Vascular collapse Ventilatory mechanics Pulmonary efficiency Pulmonary disease	2 1	1 3
Gastrointestinal function	Function and motility	2	-
Renal function	Renal physiology Renal pathology	1_	2
Nutritionmetabolism	General metabolism Specific metabolites	1 4	-
Musculoskeletal function	Decalcification Work capacity Bone pathology	1 2 -	- - 2
Endocrine function	Stress effects Thermal regulation	1 1	2
Hematological function	Cytogenetics Cell dynamics Coagulation Tissue pathology	1 3 1	- 2 - 1
Microbilogy and immunology	Ecology	4	-
	Totals	38	17

3. 1. 2 Behavioral Research Program Summary

The behavioral research program defined in this study was based on current and future NASA program requirements and a critical review and evaluation of the information needs, research categories, and experiments cited in the literature. In general, the literature concerned with measurement and evaluation of man's reaction and adaptation to the orbital environment and his ability to perform meaningful tasks for extended periods of time.

During the study, a structured approach was developed which consisted of the following:

- 1. Determination of information needs.
- 2. Establishment of research categories and description of experiments responsive to the information needs.
- 3. Development of a plan for implementing the research program.

The program is summarized in Table 3-2.

3.1.3 Bioscience Research Program Summary

The goal of the bioscience research program described here is to increase our understanding of biological mechanisms through the study of various life processes in the unique environment of a weightless, orbiting space laboratory. The program is composed of 50 experiments utilizing 19 different plant and animal subjects that vary in biological complexity from single-cell protozoa and bacteria to primates. This program was developed by selecting representative experiments from a variety of sources. The American Institute of Biological Science (AIBS) Report of December 1967 provided guidance in the matters of priority research areas and effective utilization of experimental subjects.

The entire program is designed for the performance of basic rather than applied research and is structured around nine research objectives. Of these objectives, seven were recommended by the AIBS report and two were added. For each research objective, several research categories were defined, based on the classes or organisms which the AIBS suggested as most suitable for research related to a particular objective. For example, with respect to

Table 3-2
BEHAVIORAL RESEARCH PROGRAM SUMMARY

Information Needs	Research Category	Experiments
Man's Capability to	Mission activities	Accessibility
change his environment		Communication and Recording
environment		Mass Translation
		Monitoring and Observation
		Orientation, Stability, and Restraint
		Personnel Translation
		Restraint and Gross Force Generation
		Restraint and Fine Force Generation
Effects of the	Basic behavioral	Auditory Function
environment on man	integrity	Chemical Sense Function
		Higher Mental Function
		Orientation Senses
		Psychomotor Function
		Somesthetic Function
		Visual Function
Effects of man -	Habitability	Clothing
hardware - environ-	•	Food Management
ment interactions		Hygiene System
		Interior Design
		Interpersonal Factors
		Intrapersonal Factors
		Work-rest-sleep Cycles
		Recreation
		Volume and Layout
		Waste Management
		Water Management

the biorhythm research objective, three research categories are defined: vertebrate, invertebrate, and plant. Representative experiments were then found which were responsive to each of the research categories. The total program is summarized in Table 3-3, which shows the number of representative experiments in the applicable research category and the related research objective.

The resulting program provides broad capability for bioscience research by including the necessary equipment and environmental control systems to permit a wide variety of investigations beyond those specifically programmed. In the formulation of this program, it was recognized that exobiology and remote sensing of terrestrial life were additional significant aims of the NASA bioscience program. However, these areas were eliminated as inappropriate to a manned Earth-orbiting laboratory because of the constraints imposed by Earth orbit and because current or projected technology makes them unrealistic goals within the general time-frame of this laboratory program.

Table 3-3
BIOSCIENCE RESEARCH PROGRAM OUTLINE

	Research Categories					
Research Objectives	Vertebrate	Invertebrate	Plant	Microbial	Cellular	
Behavior	3	5	_	. -	-	
Biorhythms	2	2	2	-	-	
Genetics	1	2	-	3	-	
Geosensitivity	1	1	2	-	1	
Hemodynamics	1	-	-	, -	-	
Metabolism	2	3	2	1		
Morphogenesis	2	4	4	.		
Parasitism	 -	1		.2		
Reproduction	. 2	1	.	-	<u>-</u>	

3.1.4 Life Support and Protective System Research Program Summary

Flight research in life support and protective systems (LS/PS) has as its goal the advancement of the technology that will support man in long-duration space flight. As a basis for structuring and analysing this research, the study team established three research objectives: (1) flight qualification and/or verification of integrated life support systems, (2) flight qualification and/or verification of subsystems and components, and (3) research in the basic technology which will lead to system development.

The inclusion of equipment-development programs in a life sciences research program may be considered by some as an unusually liberal interpretation of the word "research." These activities, however, are essential to the accomplishment of the goal and therefore are included in the research program structure. The research and/or development activities associated with systems, subsystems, and components were identified by examining the related functional requirements. The resulting research program is shown in Table 3-4. At least one representative experiment associated with each functional requirement has been described. The analysis of these experiments has provided the details of equipment and support requirements which are described in later sections. In many instances, several approaches to a functional requirement are known to be under consideration. Each of these has been identified in Section 4.4.

Table 3-4 (page 1 of 3)
BIOTECHNOLOGY LABORATORY LIFE SUPPORT AND
PROTECTIVE SYSTEM INFORMATION NEED

Research Objective	System or Technology	Functional Requirement
Integrated system flight	Advanced integrated EC/LS	Life support
verification and/or qualification	Integrated EC/LS and power systems	Life support and power
	Animal EC/LS	Experiment support
	All EC/LS systems	Maintenance

Table 3-4 (page 2 of 3)

Research Objective	System or Technology	Functional Requirement
Subsystem or component Flight verification and/ or qualification	Atmosphere supply and pressurization	Two-gas control
		Gaging
		Nitrogen and oxygen supply
		Oxygen supply by electrolysis
		Oxygen recovery
	Atmosphere purification and control	Carbon-dioxide control
		Trace contaminant control
		Microbial control and monitoring
	Thermal control	Humidity control
		Process cooling
		Process heating
	Water management	Water recovery
		Potability verification
	Waste management	Feces and urine collection
		Feces processing
	Food management	Food supply
	Crew protection	Clothingspace suit
		Back pack
		Gravity compensation
		Personal hygiene
		Fire protection
		Leakage
		Airlock
Basic technology	Atmosphere supply	Heat transfer
	Atmosphere supply/thermal control/water management	Liquid gas separation

Table 3-4 (page 3 of 3)

Research Objective	System or Technology	Functional Requirement	
	Atmosphere supply/thermal control/water management/	Solid and liquid retention	
7X 1 - 7	waste management	Solid transport by gas drag	
93 ⁸ a	k old pije in inkersk dik to i i		
	Thermal control	Atmosphere circulation	
		Heat transfer	
	Thermal control/water management	Condensation	
	Thermal control/water management/waste management/crew protection	Liquid transport by gas drag	
	Water management	Liquid and liquid mixing	
	Crew protection	Fire prevention	
	All life support	Solid and liquid retention	
		Solid transport by Crew	
	· ·	Liquid and solid recovery and retention	

3.2 BIOTECHNOLOGY LABORATORY PROGRAM RECOMMENDATIONS

The life sciences research program consists of closely related activities in four research areas: biomedical, behavioral, bioscience, and LS/PS. Each of these areas is concerned with the reactions of living organisms, including man, and with life support for the latter. These activities have little or nothing in common with research activities in other scientific disciplines other than the improvement of man's ability to perform research in those areas as a result of the information gained from life sciences research. Thus, equipment required in a life sciences research laboratory is specialized and, generally, will have little application to research in other disciplines.

Therefore, because of the specialized nature of the research and the facilities required to support it, it is recommended that the biotechnology laboratory portion of a space station program be managed as an integrated project through all its phases. This approach will provide maximum opportunity to ensure that the life sciences research for the missions retains the position of prominence in which it has been placed by recent NASA center studies of space laboratory research programs and configurations. It will further ensure that the physical arrangement of the biotechnology laboratory facility will promote the most efficient application of the equipment which it contains.

The alternate to this integrated project approach is to have the integration of the life sciences research experiments managed as just another part of the total experiment program for the mission(s). Although this approach would no doubt result in a life sciences research facility, the integrated project approach would provide a more effective one for the reasons cited above.

With the integrated-project approach, all aspects of the laboratory program would be managed by a single office. This would include items such as space-station program interfaces, equipment development, laboratory layout and mockups, and laboratory assembly and test. This approach will require an "integrated" management team composed of representatives of at least the three offices (OMSF, OART, and OSSA) concerned with life sciences research. That such a management team will work has been shown by the effectiveness of the NASA team from those three offices which coordinated this study.

The concept of a biotechnology laboratory as a major "slice" of a space station is illustrated in Figure 3-1. A gross schedule of activities associated with the laboratory is also shown. (This study provided the information on experiment equipment indicated under 1968 in the figure.) Details of this schedule are discussed in the following paragraphs.

3.3 ACTIVITIES AND SCHEDULES

The implementation of the biotechnology laboratory as an integrated project within the planned space-station program requires the accomplishment of

BIOTECH LAB DEVELOPMENT APPROACH

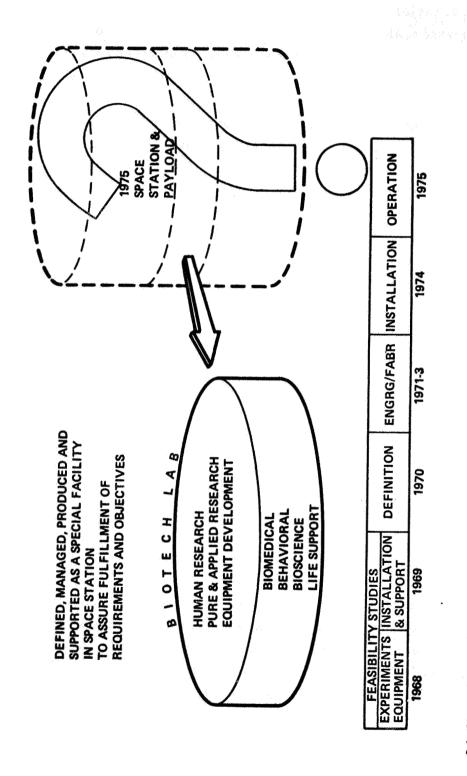


Figure 3-1. Biotech Lab Development Approach

several major tasks. Because of the foresight of several NASA offices associated with the life sciences research to be conducted in such a laboratory, several of the tasks have been accomplished or are in process. The timeliness of these tasks is illustrated in Figure 3-2, which relates the biotechnology laboratory project to the space station programs. The space station program schedule is based on the contractor's understanding of current NASA plans for the Space Station Phase B study and launch date goal.

The space station Phase C effort is the key activity affecting the timing of the biotechnology laboratory project, because the detail design of the space station will begin in this phase. To obtain maximum responsiveness to the biotechnology laboratory requirements in the space station design, these requirements should be imposed in advance, or at least concurrently with the Phase C effort. It is apparent from the schedule that a concurrent approach is practical. The information available in this document, together with information from other programs such as the Integrated Medical Behavioral Laboratory Measurement System (IMBLMS), will permit an indepth study of the requirements of a specific life sciences research program. The results of the study would provide the input to a biotechnology laboratory Phase C effort, beginning in late 1969 at about the same time that the space station Phase C will begin.

Many of the SRT programs for the biotechnology laboratory are already under way and will continue in support of all future missions. A suggested schedule for other activities associated with the laboratory and their relation to the space station program are also shown.

The most important single activity associated with this program is the formulation of the specific life sciences research objectives for the next major space research mission. These objectives must be defined to permit the imposition of specific requirements upon the space station design. Sufficient information is at hand to permit this to be done within the next year.

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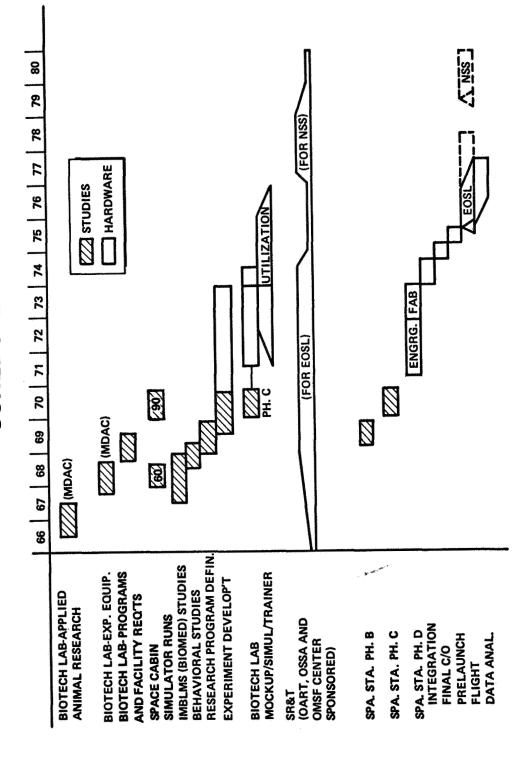


Figure 3-2. Biotech Lab Study and Development Schedule

Section 4 LIFE SCIENCES RESEARCH REQUIREMENTS

This section describes in detail the research programs which were used as a basis for identifying equipment requirements. The Programs are a concensus and thus are a valid base for future space research planning. Measurements and equipment requirements are identified and their common application is presented in matrix charts in Appendix A. The four areas of biomedical and human research, behavioral research, bioscience research, and life support and protective systems research, are covered in separate sections to permit selection of research from each area when "packages" to fulfill specific mission objectives are required.

4.1 BIOMEDICINE AND HUMAN RESEARCH

The program described in this section is directed toward a better understanding of human physiology in long-duration, null-gravity, space flight. This is achieved partly through studies that seek to qualify man for space flight by making measurements on man himself in that environment. However, it is recognized that the long-term qualification of man in space can be achieved only through a program which goes beyond the study of normal space physiology by including research in abnormal physiology, pathogenesis, and therapeutics. The accompanying program of human and animal experiments includes these subjects and transcends any current research plan of this nature.

4.1.1 Research Objectives

The representative biomedical experiments on man and animals in the program described here were selected by responding to the following objectives:

1. Determination of the effects of the space environment on human and other mammalian physiology.

- 2. Understanding the mechanisms of any alterations to physiology which may occur after prolonged exposure to the space environment, especially weightlessness.
- 3. Evaluation of preventive or remedial techniques designed to preserve or restore physiologic functions normal to the Earth environment.
- 4. Prediction of the onset, rate, and degree of physiological alterations which may occur in the weightless environment.
- 5. Study of induced pathological processes and exploration of the therapeutic means for their correction or elimination.

As these objectives indicate, the program seeks to explore not only normal physiology in the space environment, but also pathophysiological problems that will inevitably occur. Information of this nature clearly cannot be obtained with human subjects except as the natural need to apply therapeutic techniques arises. It seems prudent, therefore, to prepare for this eventuality by making investigations with animals in the experimental areas of pathophysiology and pharmacology before the need to apply the therapies to human subjects arises.

4.1.2 Experiment Selection

In biomedicine-human research there has been a definite evolution of ideas concerning the research objectives identified in Section 4.1.1. This is apparent in a progression of reports by various contractors, scientific evaluation groups, and NASA/government working groups. These and other reports were reviewed to identify specific information needs, which were then grouped under 10 separate subject headings. The divisions which were established to provide a structured approach toward the objectives agreed upon were:

- 1. Neurological
- 2. Cardiovascular
- 3. Pulmonary
- 4. Gastrointestinal
- 5. Renal
- 6. Nutrition and metabolism.
- 7. Musculoskeletal
- 8. Endocrine
- 9. Hematologic
- 10. Microbiologic and immunologic.

After these specific information needs were established, they were further subdivided into research categories, each of which represents a significant separate but related area of required investigation. Finally, experiment titles taken from all of the basic reference sources, as well as from the other documents, were grouped within the appropriate research categories. This compilation served as the baseline experiment list from which the final program was derived. During the review of individual experiments in the baseline list, it became evident that there was considerable redundancy, even among experiments whose titles differed widely. Such redundancy was eliminated by discarding certain experiments and by consolidating others which overlapped significantly. Comparison of the consolidated list with the information needs previously identified showed that certain critical areas were inadequately covered. In such cases, experiments were designed to produce a complete program. The final group of 55 experiments is directly responsive to all specified information needs and is sufficiently representative of current scientific thought and technology to provide a broad research capability extending considerably beyond the specific program objectives. The final program is shown in Table 4-1.

Table 4-1 (page 1 of 4)
BIOMEDICAL RESEARCH PROGRAM

Information Need	Research Category	Repr	esentative Experiments*
Neurological function	Vestibular function	1 - 1	Head Movement Effects
		1-2	Otolith and Semicircular Canal Sensitivity
		1-4	Vestibular Electrical Activity**
((3875) (1375) ((3875) (1375) (1375) (1375) (1375)	Circadian rhythms	1-3	Day-Night Cycle Alterations**
Cardiovascular function	Cardiovascular Deconditioning	1-5	Circulatory Response to Exercise
		1-6	Volume Effects on Arterial Pressure Control System

^{*}Numbers preceding experiment titles correspond to those in Volume II.
**Experiment uses animal subjects.

Table 4-1 (page 2 of 4)

Information Need	Research Category	Repr	esentative Experiments*
		1-7	Peripheral Venous Compliance
		1-8	Cardiac Dynamics
		1-9	Intraocular Arterial Blood Pressure
	Deconditioning Countermeasures	1-10	Lower Body Negative Pressure Device Evaluation
		1-11	On-board Centrifuge Evaluation
		1-12	Occlusive Cuffs Evaluation
	Homeostatic mechanisms	1-13	Carotid Sinus Sensitivity
		1-14	Peripheral Arteriolar Reactivity
		1-15	Blood Volume and Distribution
		1-16	Carotid Baroreceptor Electrical Activity
	Verification of measurement techniques	1-17	Direct Cardiac Output Measurement**
	Vascular collapse	1-18	Response to Shock Therapy**
Pulmonary function	Ventilatory mechanics	1-19	Pulmonary Mechanics
		1-20	Respiratory Control
	Pulmonary efficiency	1-21	Blood Gas Exchange
		1-22	Lung Cleansing**
	Pulmonary disease	1-23	Induced Pulmonary Infections**
	4444 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 120 - 1 120 - 120	1-24	Recovery from Non- Infectious Lung Trauma*
Gastrointestinal	Gastrointestinal	1-25	Intestinal Absorption
function	function and motility	1-26	Motility and pH

^{*}Numbers preceding experiment titles correspond to those in Volume II.
**Experiment uses animal subjects.

Table 4-1 (page 3 of 4)

Information Need	Research Category	Repre	esentative Experiments*
Renal function	Renal physiology	1-27	Indexes of Renal Function
	Renal pathology	1-28	Calculus Formation**
		1-29	Renal Infection**
Nutrition	General metabolism	1-30	Energy Metabolism
metabolism	Specific metabolites	1-31	Carbohydrate and Fat Metabolism
		1-32	Protein Metabolism
		1-33	Body Fluid Composition
		1-34	Mineral Metabolism
Musculoskeletal function	Skeletal decalcifica- tion	1-35	Bone Density
	Work capacity	1-36	Muscle Status
		1-37	Electromyographic Evaluation
	Bone pathology	1-38	Fracture Healing**
		1-39	Induction of Pressure Atrophy**
Endocrine function	Thermal regulation	1-41	Thermal Regulation
	Stress effects	1-40	Endocrine Assays
		1-42	Adrenal and Parathyroid Histopathology and Function**
		1-43	Gonad Histopathological Evaluation**
Hematologic function	Cytogenetics	1-44	Leukocyte Replication
	Cell dynamics	1-45	Erythrocyte Dynamics
		1-46	Leukocyte Dynamics
		1-47	Platelet Dynamics
		1-49	Leukocyte Mobilization after Chemical Challeng
		1-50	Maximum Rate of Erythrocyte Production

^{*}Numbers preceding experiment titles correspond to those in Volume II.
**Experiment uses animal subjects.

Table 4-1 (page 4 of 4)

Information Need	Research Category	Repr	esentative Experiments*
	Coagulation	1-48	Hemostasis
	Tissue pathology	1-51	Wound Healing**
Microbiology and immunology	Microbiology	1-52	Microbial Evaluation of Environment
		1-53	Microbial Evaluation of Crew Members
		1-54	Air Sampling
	Immunology	1-55	Immunological Evaluation of Crew Members

^{*}Numbers preceding experiment titles correspond to those in Volume II.
**Experiment uses animal subjects.

4.1.3 Program Implementation

Because of research and development requirements in certain areas and because of expected facilities constraints, it is not anticipated that the entire biomedical research program will be implemented at one time. The following implementation plan will maximize the efficiency of program development by identifying those areas deserving of the greatest and earliest emphasis.

In developing this plan, all the experiments in this program were evaluated with respect to the urgency of the information needs which they satisfied. It became evident that in all the information need areas there were specific requirements of equal importance. The experiments could not, therefore, be ranked according to strict priority because the information needs for the various functional systems contained equally important information gaps. Thus the total program was divided into three separate phases which might be considered as sequential implementation phases (See Table 4-2). Within each phase, experiments were considered to be equally important. Were the entire laboratory program to be implemented in separate phases, the summary detailed in Table 4-2 would represent an efficient way of answering the most important questions first. Answers to highly desirable but less pressing information needs would be deferred until activation of later program stages.

Table 4-2 (page 1 of 3)
BIOMEDICAL AND HUMAN RESEARCH PROGRAM IMPLEMENTATION PLAN

Information Need	Research Category	Repr	esentative Experiments*
	Phase I Experiments	(19)	
Neurological function	Vestibular function	1-1	Head Movement Effects
		1-2	Otolith Semicircular Canal Sensitivity
Cardiovascular function	Cardiovascular deconditioning	1-5	Circulatory Response to Exercise
		1 -6	Volume Effects on Arterial Pressure Control System
		1-8	Cardiac Dynamics
	Cardiovascular deconditioning	1-10	Lower-Body Negative- Pressure Device
	countermeasures	1 - 1 1	On-board Centrifuge Evaluation
		1-12	Occulusive Cuffs Evaluation
	Verification of measurement techniques	1-17	Direct Cardiac Output Measurement**
	Va sc ular collapse	1-18	Response to Shock-therapy**
Pulmonary function	Ventilatory mechanics	1-19	Pulmonary Mechanics
	Pulmonary disease	1-23	Induced Pulmonary Infections**
Nutrition-metabolism	General metabolism	1-30	Energy Metabolism
	Specific metabolities	1-34	Mineral Metabolism
	Skeletal decalcification	1-35	Bone Density
Musculoskeletal function	Bone pathology	1-38	Fracture Healing**

^{*}Numbers preceding experiment titles correspond to those in Volume II.

^{**}Experiment uses animal subjects.

Table 4-2 (page 2 of 3)

Information Need	Research Category	Repr	esentative Experiments*
	Phase I Experiments	<u>(19)</u>	
Hematologic function	Cell dynamics	1-50	Maximum Rate of Erythrocyte Production**
	Tissue pathology	1-51	Wound Healing**
Microbiology- immunology	Ecology	1-53	Microbial Evaluation of Crew Members
	Phase II Experiments	(16)	
Cardiovascular function	Cardiovascular deconditioning	1-14	Peripheral Arteriolar Reactivity
	Homeostatic mechanisms	1-16	Carotid Baroreceptor Electrical Activity**
Pulmonary function	Pulmonary efficiency	1-21	Blood Gas Exchange
	Pulmonary disease	1-22	Lung Cleansing**
		1-24	Recovery from Non- infectious Lung Trauma*
Renal function	General renal function	1-27	Indexes of Renal Function
Nutrition-metabolism	General metabolism	1-33	Body Fluid Composition
Musculoskeletal function	Work capacity	1-36	Muscle Status
*	Bone pathology	1-39	Induction of Pressure Atrophy**
Endocrine function	Stress effects	1-41	Thermal Regulation
		1-43	Gonad Histopathological Evaluation**
Hematologic function	Cell dynamics	1-45	Erythrocyte Dynamics
		1-46	Leukocyte Dynamics
		1-47	Platelet Dynamics
Microbiology-	Microbiology	1-54	Air Sampling
immunology	Immunology	1-55	Immunological Evalua- tion of Crew Members

^{*}Numbers preceding experiment titles correspond to those in Volume II.

**Experiment uses animal subjects.

Table 4-2 (page 3 of 3)

Information Need	Research Category	Repr	esentative Experiments*
	Phase III Experiments	s (20)	
Neurological function	Circadian rhythm	1-3	Day Night Cycle Alterations**
	Vestibular function	1-4	Vestibular Electrical Activity**
Cardiovascular function	Cardiovascular deconditioning	1-7	Peripheral Venous Compliance
		1-9	Intraocular Arterial Blood Pressure
	Homeostatic	1-13	Carotid Sinus sensitivity
	mechanisms	1-15	Blood Volume and Distribution
Pulmonary function	Ventilatory mechanics	1-20	Respiratory Control
Gastrointestinal	Gastrointestinal	1-25	Intestinal Absorption
function	function and efficiency	1-26	Motility and pH
Renal function	Renal pathology	1-28	Calculus Formation**
		1-29	Calculus Formation** Renal Infection**
Nutrition-metabolism	Specific metabolities	1-31	Carbohydrate and Fat Metabolism
		1-32	Protein Metabolism
Musculoskeletal function	Deconditioning indexes	1-37	Electromyographic Evaluation
Endocrine function	Stress effects	1-40	Endocrine Assays
		1-42	Adrenal and Parathyroid Histopathology and Function**
	Cell dynamics	1-49	Leukocyte Mobilization after Chemical Challenge
Hematologic function	Cytogenetics	144	Leukocyte Replication
	Coagulation system	1-40	Endocrine Assays
Microbiology- immunology	Ecology	1-52	Microbial Evaluation of Environment
	a little il Elicabili.		

^{*}Numbers preceding experiment titles correspond to those in Volume II.

^{**}Experiment uses animal subjects.

Assignment of individual experiments to the various phases was based on information in the references reviewed and on general familiarity with current opinion among participants in organized aerospace medicine research. Certainly, with the passage of time, changes in specific phase assignment would occur either because of technological developments or because answers to certain information needs might become more critical as experience in space increases.

4.1.4 Measurement Requirements

Measurement requirements for this research were developed through two approaches. The information needs and research categories were analyzed for measurements which could be anticipated on the basis of current scientific technology, regardless of the constraints imposed by current technology. Also, individual experiments were analyzed separately to determine the measurements specified in each of the descriptions. The two lists were then consolidated to produce a list which was used to analyze overall program commonalities.

During the review of experiments, it became evident that not all experiments were sufficiently detailed to specify measurements. When such deficiencies were identified, experiment plans which met the stated objectives were originated. The list of required measurements is shown in Table 4-3.

The measurements extracted from analysis or construction of individual experiment descriptions are included in the Experiments Requirements Summaries which are presented in Volume II. The results are also presented in the form of matrix charts (Appendix A) in which individual experiments are compared with the complete list of measurements required for the total biomedical program.

The experiments and measurements embodied in the Integrated Medical and Behavioral Laboratory Measurements System (IMBLMS) concept meet many of the information needs identified in this report. IMBLMS has therefore been a central reference point for this study. It would seem that the final similarities between the IMBLMS program and that outlined in this report, which was derived independently, tend to validate the requirements set forth in both program descriptions.

Table 4-3 (page 1 of 2) BIOMEDICAL AND HUMAN RESEARCH MEASUREMENTS

General, Environmental

Airflow

Ambient air pO2, pCO2 Ambient temperature Background radiation Cine photography Fluid temperature

Humidity

Light intensity

Photography, still (Polaroid, 33 mm, color)

Photomicrography, still (Polaroid, 33 mm, color)

Track contaminant levels

Neurological

Egocentric visual localization of the horizon (EVLH)

Electroencephalography

Eye muscle balance

Nystagmus

Ocular counter rolling

Oculogravic illusions

Oculogyric illusions

Task performance

Physiological

Arterial blood pressure

Arterial filling time

Ballistocardiography

Body mass

Body temperature, core and skin

Body volume Bone density

Breath-holding time

Cardiac output

Circulation time Electrocardiography

Electromyography

Biochemical -- Metabolic

Blood ACTH

Blood Blood bicarbonate

Blood free fatty acids

Blood glucose

Blood lactic acid

Blood PBI

Blood pH Blood pO₂, pCO₂

Blood TSH

Blood urea nitrogen

Gastric contraction force

Gastric volume

Gastrointestinal motility

Gastrointestinal pH

Heart rate

Muscle size

Muscle strength

Phonocardiography

Plethysmography

Spirometry

Venuous filling time

Serum alkaline phosphatase

Serum calcium Serum chloride

Serum creatinine Serum osmolality

Serum phosphorus

Serum potassium

Serum proteins (total)

Serum sodium

Table 4-3 (page 2 of 2)

Urine aldosterone

Urine acidity

Urine calcium

Urine catecholamines

Urine creatine

Urine creatinine

Urine glucose

Urine nitrogen

Urine osmolality

Urine pH

Urine phosphorus

Urine steroids

Urine volume

Hematological--Microscopic

Bacterial morphology

Bleeding time

Cell morphology

Clot retraction

Clotting time

Cytogenetic evaluation

Erythrocyte count

Erythrocyte size

Hematocrit

Hemoglobin

Radioisotopic Tracer Studies

Blood volume

Plasma volume

RBC mass

RBC survival time

Specific absorption rates

Total body water

Fecal mass
Fecal nitrogen
Fecal calcium
Fecal phosphorus

Leukocyte count (total and differential)
Phagocyte index
Platelet count
RBC fragility
Reticulocyte count
Sperm count
Sperm motility evaluation
Tissue fixation
Urinary sediment evaluation

4.1.5 Equipment Requirements

The equipment required to perform the biomedical research program is presented here in two forms. Table 4-4 presents a consolidated equipment list derived from the experiment requirements summaries in much the same manner as were the individual measurements. A matrix that relates equipment with experiments may be found in Appendix A. Some of the equipment represented in these tables needs no additional description, while other items required at least some specification, even when no accurate description was possible. These items are further detailed in the equipment description lists

Table 4-4 cas described a series

BIOMEDICINE-HUMAN RESEARCH PROGRAM--EQUIPMENT LIST

Aerosol particle analyzer*
Arterial pressure recorder*
Autoclave*
Automatic cell counter*

Bacterial culture equipment Ballistocardiograph* Bone densitometer*

Camera, cine
Camera, still (Polaroid)
Camera, still (35 mm)
Cardiotachometer*
Carotid cuff
Centrifuge, clinical*
Centrifuge, human, short-radius
Chemical analyzers*

Electrocardiograph*
Electroencephalograph*
Electromyograph*
Electrooculograph*
Endoradiosonde*
Erogmeter*

Filtration apparatus Frozen storage unit

Gas chromatrograph*
Gas flowmeter
Goggles, plethysmographic*

Hemotocrit tubes, capillary Hemocytometer Hemoglobinometer* Humidity recorder*

Incubator, 37°C

Lower-body negative-pressure device*

Manometer, recording*
Mass-measurement device, large*
Mass-measurement device, small*
Microscope*
Muscle dynamometer*

Nebulizer chamber* Nitrogen analyzer* Nystagmographic goggles*

Ophtalmoscope*
Oscilloscope*
Osmometer*
Otoscope*
Oximeter, ear*

markettings Nit

pH meter*
Phonocardiograph*
Phoropter*
Plethysmograph, capacitance*
Pressure, cuffs limb*
Pressure transducers

Scintillation counter/scaler*
Slide-staining apparatus
Specific ion electrodes*
Spectrophotometer*
Spirometer*
Stethoscope
Stomach tube and balloon
Surgical instruments
Syringes

Temperature recorder*
Thermistors, skin*
Thermistors, rectal*
Thermistor, tympannic*
Tilting motion chair*
Timers

Vision test targets Volumeter, whole body

^{*}Details in equipment description section.

in Section 7. In many cases, the equipment described is currently under development by various contractors to NASA. When possible, information was obtained which accurately reflects the current status of the individual items described. In other cases, specifications set forth by NASA in proposal requests were used as description and specification guidelines. In still other cases, where a requirement for the described equipment had not been specified elsewhere, the specifications were approximated in accordance with knowledge of the current technology.

It is anticipated that a major portion of the equipment outlined will be provided by the final result of IMBLMS development because a great deal of similarity exists between the IMBLMS program and the biomedical program outlined in this report. Nevertheless, the potential breadth of the research to be performed in the biotechnology laboratory conceived in this study exceeds that of IMBLMS; thus, a few items of equipment will require development if the program is to be accomplished in its entirety.

4.2 BEHAVIORAL RESEARCH

The behavioral research program defined in this study concerns man and the activities he performs in relation to his environment. The program is responsive to the objectives and information needs stated in numerous NASA and aerospace industry reports. The program was developed in the following steps:

- 1. Statement of the broad objectives of the behavioral research objectives.
- 2. Identification of the information-need categories related to the objectives.
- 3. Selection of the research categories and experiments responsive to the information needs.
- 4. Development of an implementation program plan.

4.2.1 Research Objectives

The objectives formulated for the behavioral research program are a synthesis of those stated throughout the various reference documents. Since these tend to concentrate on specific areas of research rather than on general objectives, it was necessary to restate them in broader terms. Thus, the following

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objectives were established to provide a basis for the selection of experiments and the definition of an extensive research program:

- 1. Determination of man's capability to perform meaningful tasks for long-duration missions.
- 2. Determination of the system design factors that affect morale over extended periods of time in orbit.
- 3. Determination of the effects of prolonged exposure to the orbital environment on the normal ranges of sensory, perceptual, and psychomotor functions.

These objectives are directed towards optimizing man's performance and providing information which will help to ensure his safety and well-being throughout the mission. Results will be used to develop design criteria for future spacecraft and to aid in formulating operational task requirements.

4.2.2 Experiment Selection

Experiment selection was accomplished by first compiling a catalog of 534 potentially useful experiments, then evaluating their responsiveness to the specific research categories. The catalog was the result of a review of numerous reference documents and MDAC experiment data. From the catalog, a concise listing of 80 experiments that met the following criteria was made:

- 1. Those that require human capabilities for accomplishment.
- 2. Those that investigate human capabilities directly or indirectly.
- 3. Those that are capable of investigating human performance through minor modification of observations.
- 4. Those that investigate traditional behavioral areas.

The source documents were then reviewed for information needs. Again, no single-source document or proposed program included all the information needs required. Specific information needs with a common purpose were grouped under three major categories: (1) man's effect on his environment, (2) effects of the environment on man, and (3) effects of man-hardware-environment interation. To provide a structured approach to the final selection of experiments, the three information-need categories were subdivided into research categories denoting the distinct areas of investigation required to fulfill each information-need.

The 80 candidate experiments were analyzed for relevancy to the various research categories. Some were responsive to several research categories, others were directed towards only one or two. Degree of responsiveness varied greatly, so that some categories were complete and others were not. There was not enough detailed information to determine the relative importance of a particular experiment to a specific research category.

To avoid redesigning experiments already documented in the literature, a new approach was adopted which consolidated the common objectives, measurements and techniques of the 80 experiments into 26 individual experiments responsive to each of the research categories. Thus, for each experiment area, all the measurements and techniques which might be employed to investigate the total behavioral gamut which could conceivably be included within that classification were identified. Where details regarding measurements, equipment, or time-constraints did not exist, approximations were made. A summary of the total program is shown in Table 4-5.

Evaluation of extravehicular activities (EVA) is not included as a separate experiment, since it overlaps considerably with evaluation of life support and protective systems (LS/PS). Thus, it seemed unreasonable to restate the requirements for pressure suits, life support systems, consumables, and other resources for the behavioral research program. Further the measurement parameters required for the mission activities and basic behavioral integrity experiments are applicable (with few exceptions) to both intravehicular and extravehicular performance. For EVA evaluation, the major functions are egress/ingress, translation, and work performance. All can be effectively measured during pressure-suit evaluation. In fact, the latter requires task performance in order to place appropriate loads on the hardware systems; such performance may be evaluated as part of the behavioral research program.

4.2.3 Program Implementation

Three approaches could be used to implement a behavioral research program. The first, and most desirable, would be to include all 26 experiments on the initial biotechnology laboratory mission.

Table 4-5 BEHAVIORAL RESEARCH PROGRAM SUMMARY

Information Needs	Research Category	Experiments
Man's capability to change his environment	Mission Activities	2-1 Accessibility 2-7 Communication and Recording 2-5 Mass Translation 2-8 Monitoring and Observation 2-6 Orientation, Stability, and Restraint 2-4 Personnel Translation 2-3 Restraint and Gross Force Generation 2-2 Restraint and Fine Force Generation
Effects of the environment on man	Basic behavioral integrity	2-11 Auditory Function 2-14 Chemical Sense Function 2-9 Higher Mental Function 2-13 Orientation Senses 2-15 Psychomotor Function 2-12 Somesthetic Function 2-10 Visual Function
Effects of man - hardware - environment interactions	Habitability	2-21 Clothing 2-16 Food Management 2-19 Hygiene System 2-22 Interior Design 2-24 Intrapersonal Factors 2-25 Interpersonal Factors 2-25 Work-rest-sleep Cycles 2-23 Recreation 2-20 Volume and Layout 2-18 Waste Management 2-17 Water Management

This program would be the most flexible. Any information need or research category could be emphasized at any time. The program would be limited, however, by the availability of hardware and techniques. Crew time or other station resources could also constrain the program.

Another approach would be to package the experiments according to each of the three research categories shown in Table 4-5. Packaging related experiments has the advantage of minimizing hardware and facilities required on board. However, there are two problems. First, the only basis for beginning priorities to the three groups would be judgmental. Second, it would be more desirable to have representative experiments from each of the three categories, because one type of research might substantiate (or contradict) the results of another.

The recommended approach is to establish priority ratings for individual experiments within each research category, based on the frequency with which they appear in the literature. From these ratings, sequential experiment programs can be defined. This approach is based on the premise that a frequently recommended experiment represents an urgent information need and, furthermore, appears to be the consensus of the scientific community.

Three candidate experiment packages based on this approach are shown in Table 4-6. They result from a detailed analysis of the measurements and observations required of 80 representative experiments selected from the literature. Experiments within each research category are listed according to decreasing frequency of appearance in the literature, with the most frequent shown at the top of each major column.

Each experiment package is identified by reading across the three columns. The division lines are admittedly judgmental rather than statistical. This concept is not completely arbitrary, however. The three categories of experiments included in each package tend to augment and complement each other. Furthermore, the three packages tend to proceed from the general to the specific. This is desirable because the program should provide the most significant data in the shortest time. In other words, overall task performance degradation would be expected early in the mission. Should it occur, specific causes could be investigated later in the mission. Finally,

Table 4-6 PROGRAM IMPLEMENTATION SUMMARY

		Dans Categories	
Mission		Mescartin caregories	Basic Behavioral Integrity
Phases	Mission Activities	нариарииу	
I	2-5 Mass Translation	2-24 Intrapersonal Function	2-15 Psychomotor Functions
	2-4 Personnel Translation	2-25 Interpersonal Function	2-13 Orientation Senses
	2-3 Restraint and Gross Force Generation	2-20 Volume and Layout	
	2-2 Restraint and Fine Force Generation		
	2-6 Orientation, Stability, and Restraint		
	7 1 A	2-16 Food Management	2-9 Higher Mental Functions
1	2-8 Monitoring and Observation	2-18 Waste Management	2-10 Visual Function
		2-19 Hygiene System	2-11 Auditory Function
		2-21 Clothing	
111	2-7 Communication and	2-17 Water Management	2-14 Chemical Sense Function
1	Recording	2-22 Interior Design	2-12 Somesthetic Function
		2-26 Work-rest-sleep Cycles	
		2-23 Recreation	

the program is consistent with the three kinds of information needed, and it is provided by the data collected. The desired results can be classified as follows:

- 1. Those which provide design requirements for future space vehicles.
- 2. Those which will permit a realistic assignment of integrated crew tasks for future missions.
- 3. Those which provide insight and understanding into man's adaptive processes in the orbital environment.

4.2.4 Measurement Requirements

The measurement requirements for each of the consolidated experiments were based initially upon those stated in the literature for the original 80 experiments. Additional measurements were added as necessary to make the consolidated experiments completely responsive. A listing of measurements is provided in Table 4-7.

Several of the measurements are used in most of the experiments. Others are specialized and are used in only one research category. Many of the sensory, perceptual, and neurophysiological measurements can be taken with the subject positioned in front of a test console or performing tasks in a limited area. Auditory detection, critical flicker fusion, reaction time, information processing, problem solving, and judgment, for example, will employ relatively standard psychological tests to compare responses in orbit with those made on the ground. Differences, if any, would be attributable to the effects of orbital stressors. These measurements can be readily performed in a minimum of time.

Other measurements deal with complex behavior, e.g., body position/ locomotion and mass translation. These measurements will require larger areas and more complex recording equipment. Metabolic measurements are required in conjunction with task performance evaluations in order to assess the impact of energy costs on spacecraft life support systems.

A third type of measurement is needed to evaluate psychological adjustment to the orbital environment. Included are the use of standardized paper and pen tests and of specially developed questionnaires to assess mood, opinion, and intrapersonal and interpersonal dynamics.

Table 4-7

BEHAVIORAL RESEARCH MEASUREMENTS

Sensory, Perceptual, and Psychomotor

Attention Judgment

Auditory detection/discrimination Light and dark adaptation

Body acceleration Locomotion
Body position Manipulation

Brightness detection/discrimination Memory

Color detection/discrimination Noise sensitivity

Complex sequence Problem solving

Critical flicker fusion Reaction time

Depth perception Tracking

Dexterity Veridicality

Eye movement Vigilance
Force production Visual acuity

Form discrimination

Physiological, Neurophysiological, and Metabolic

Electroencephalography Accommodations utilization

Energy expenditure Control and display dynamics

Food consumption Emotional assessment

Galvanic skin response Event record

Skin sensitivity Mass motion/acceleration

Smell sensitivity Mood assessment
Taste sensitivity Subjective opinion

Water consumption Time

Miscellaneous Voice record

In general, the measurements in this program are required to help explain, understand, and predict behavior. They are similar to those used in ground-based laboratories in that they are primarily concerned with recording responses (or events) which can be analyzed on the ground. The on-orbit subjects serve as their own controls, having performed the same experiments in a 1-g environment prior to and subsequent to the mission. A matrix of measurements versus experiment titles is shown in Appendix A.

4.2.5 Equipment Requirements

Equipment required for the behavioral research program is tabulated in Table 4-8. Items were derived from an analysis of the measurements and observations essential to conducting the 26 consolidated experiments. In some cases, equipment requirements were directly translatable from those stated in the literature. More frequently, however, they were developed specifically for this study.

A correlation of equipment items and experiment titles is shown in Appendix A. Some items are functionally compatible and could be combined to reduce constraints on the laboratory. For example, 16 of the 37 equipment items lend themselves to groupings which result in 5 multiple-item combinations:

(1) standard dexterity apparatus, CRT and other displays, display-associated

Table 4-8
BEHAVIORAL RESEARCH EQUIPMENT AND SUPPLIES

Measurement Equipment

Accelerometers and recorders Analog recorder Audio recorder Cameras CFF source Color plates CRT and other displays Display associated controls EEG sensors and recorder Electro oculograph and recorder Ergometer GSR sensors and recorders Gustatory solutions Light source Metabolic monitor Noise and vibration source Olfactory aerosols Orthorater Pure tone source Somesthetic stimuli Standard dexterity apparatus Timer Torque force apparatus Tuning fork Veridicality tester

Support Equipment and Supplies

Attachment devices (for restraints) Background grid Clothing Dionized sterile water Life support equipment/expendables Log book Microsyringes Maneuvering aids Masses Paper and pen tests Photographic film and magnetic tapes (audio and video) Questionnaires Recreational devices/supplies Restraints Storage volume Tape measure Tethers Writing area and surface Writing instruments

controls, and an analog recorder; (2) depth-perception tester, orthorater, light source, CFF source, and veridicality tester; (3) pure tone source and noise and vibration source; (4) metabolic monitor, EMG sensors and recorder, and GSR and EEG sensors and recorder; and (5) torque-force apparatus and ergometer.

An integrated measurement system such as indicated in the IMBLMS Request for Proposal would be extremely valuable in the final definition of equipment and techniques for the laboratory. In another recent NASA-funded study, The Utilization of MOL for Early NASA Objectives, it was concluded that an integrated biomedical/behavioral measurement system would significantly reduce not only weight, power, volume, and data-management requirements, but crew time as well. Two of the integrated-measurement devices recommended in that study were the MDAC-designed Computer-Performance and Reference Evaluator (COMPARE) and Visual/Auditory Tester (VAT). Together these two devices can measure all of the basic sensory, psychomotor, and perceptual functions necessary in this research program.

The performance characteristics of the equipment are summarized in Section 7. None of these data were available in the literature; instead, they were developed from Company experience in the laboratory. Experience in underwater and Keplerian - flight studies was extremely valuable. Weight, power, and volume estimates are gross because much of the equipment has not yet been designed for space use.

4.3 BIOSCIENCE RESEARCH

The ability to explore biological organisms or systems in the unique environment of an Earth-orbiting laboratory makes available an entirely new investigative approach to the biosciences. The importance and versatility of this advanced has challenged the ingenuity of a great many bioscientists and has resulted in the proposal of a wide variety of excellent and interesting experiments. The basic aim of the final bioscience research program is to assemble from these proposed experiments a group so representative of current scientific thought and technology that, ideally, any desired experiment could be performed in the laboratory. The following program represents a practical approach to this goal, with a realistic balance

between those experiment designs immediately available and those potentially attainable within the scope of the projected orbiting laboratory program.

4.3.1 Research Objectives

Bioscience, the broadest of scientific disciplines, has a variety of potential goals. Among the stated aims of the NASA Bioscience Program as set forth in the April 1968 Bioscience Working Group report is studying "the effects of the space environment on living organisms..."

This goal was taken as the basic charter for the experiments compiled in this section. Such other aims as the search for extraterrestrial life were considered but were discarded as inappropriate to a manned Earth-orbiting laboratory. Similarly, "reconnaissance and study of terrestrial ecology and development of remote sensing life detection techniques," stated in the AIBS report of December 1967, was felt to be generally inapplicable, largely because of the limitations imposed by the current technology. Finally, "development, test, and refinement of bioregenerative life support systems," also stated in the AIBS report, was excluded from the following program, at least as a primary goal, because of its applied and specialized nature. Thus, the research outlined herein conforms rather closely to the definition of a pure or basic science program.

4.3.2 Experiment Selection

Although both the NASA Bioscience Working Group and the AIBS have stated that research on man is a significant objective of the NASA bioscience program, such experiments were excluded from the following program and were treated separately in the section on biomedicine and human research. It should be pointed out, though, that certain areas of bioscience research, while not pursued in this program for the purpose of qualifying man in space, may well provide information of value in that regard. These experiments have been included to provide the breadth and versatility desired in the fully assembled laboratory.

The process of selecting representative experiments was difficult because of the great number of excellent and appropriate experiments proposed in recent years. Selection was simplified, however, when both the NASA coordinators and the MDAC study team agreed to use the AIBS 1967 final report to NASA as a guideline. This report considered basic bioscience objectives and the most efficient means of attaining them. Since the document so clearly represented the consensus of a large group of prominent scientists, its guidance proved invaluable to the selection process. The broad biological research categories outline in the AIBS report include geosensitivity, morphogenesis, metabolism, behavior, parasitism, genetics, and biosynthesis.

To these have been added reproduction and hemodynamics, both of which are aspects of physiology likely to be affected by weightlessness and/or radiation. Although hemodynamics is extensively investigated by noninvasive methods in the biomedicine and human research program, the scope of the experiment outlined in this section is sufficiently large to warrant its inclusion in the program, thereby ensuring that the necessary equipment will be available for associated biological measurements, should they be desired. Similarly, the tissue-culture/cell-reproduction experiment will not only add to the technological capability of the program but should provide much interesting and significant data on a basic process vital to all biosystems. In the overall program outline presented in Table 4-9, the bioscience section was organized in accordance with the AIBS objectives and with the same format as the other segments of the total laboratory program. One difference, however, is the use of the term "research objectives" rather "information needs, " since it was agreed that the former was more representative of the intent of any basic science program.

4.3.3 Program Implementation

In this comprehensive bioscience program, the variety of experimental subjects required presents a special problem of implementation. Because of the diverse life-support requirements, the logistics of providing for all desired test organisms becomes complex. Furthermore, it is unrealistic to assume that the technology for supporting these organisms will be uniformly available at the start of the program.

It is difficult to use priority of research objectives as a basis for separating the program into implementation stages because the very nature of this research precludes the assignment of priority values. Therefore, the

Table 4-9 (page 1 of 3). BIOSCIENCE RESEARCH PROGRAM

Behavior Vertebrate behavior 3-1 Primate Behavior 3-2 Mouse Growf 3-2 Mouse Growf 3-3 Teleost Behavior 3-4 Daphnia Behavior 3-5 Cockroach B 3-5 Cockroach B 3-7 Ant Behavior 3-7 Ant Behavior 3-7 Ant Behavior 3-10 Mouse Biorhythms 3-11 Cockroach Biorhythms 3-12 Drosophila P Plant biorhythms 3-13 Cockroach Bi 3-12 Avena Biorhythms 3-13 Gapsicum Biorhythms 3-13 Gapsicum Biorhythms 3-15 Genetics of Maptation 3-16 Drosophila Genetics Microbial genetics 3-18 E. coli Genet Ayegeny and 3-19 Lygsogeny and 3-19 L	Research Category Rep	Representative Experiments
3-2 3-3 Invertebrate behavior 3-4 3-5 3-6 3-7 3-8 Vertebrate biorhythms 3-12 Plant biorhythms 3-13 Yertebrate genetics 3-14 Vertebrate genetics 3-16 3-16 3-16 3-17 Microbial genetics 3-18	3-1	Primate Behavior
3-3 Invertebrate behavior 3-4 3-5 3-6 3-7 3-8 Wertebrate biorhythms 3-10 Invertebrate biorhythms 3-13 Plant biorhythms 3-13 Wertebrate genetics 3-15 Wicrobial genetics 3-16 3-16 3-17 Wicrobial genetics 3-19		Mouse Growth and Behavior
Invertebrate behavior 3-4 3-5 3-6 3-7 3-8 Vertebrate biorhythms 3-10 Invertebrate biorhythms 3-11 Plant biorhythms 3-14 Vertebrate genetics 3-16 3-16 3-17 Microbial genetics 3-19		Teleost Behavior
3-5 3-6 3-7 3-8 Wertebrate biorhythms 3-10 Invertebrate biorhythms 3-12 Plant biorhythms 3-14 Vertebrate genetics 3-16 3-16 3-17 Microbial genetics 3-19	3-4	Daphnia Behavior Patterns
3-6 3-7 3-8 Vertebrate biorhythms 3-10 Invertebrate biorhythms 3-11 Plant biorhythms 3-14 Vertebrate genetics 3-16 3-16 3-17 Microbial genetics 3-19		Cockroach Behavior Patterns
ms Vertebrate biorhythms 3-9 Invertebrate biorhythms 3-11 Plant biorhythms 3-13 Vertebrate genetics 3-16 Microbial genetics 3-18 Microbial genetics 3-18		Drosophila Behavior
Nertebrate biorhythms 3-9 Invertebrate biorhythms 3-11 Plant biorhythms 3-13 Vertebrate genetics 3-16 3-16 Microbial genetics 3-18 3-19		nt Behavior
Wertebrate biorhythms 3-10 Invertebrate biorhythms 3-12 Plant biorhythms 3-13 Vertebrate genetics 3-16 Microbial genetics 3-18 Microbial genetics 3-18		Dinoflagellate Behavior
Invertebrate biorhythms 3-11 3-12 Plant biorhythms 3-14 Vertebrate genetics 3-16 3-16 3-17 Microbial genetics 3-19	3-9	Primate Biorhythm
Invertebrate biorhythms 3-12 Plant biorhythms 3-13 3-14 Vertebrate genetics 3-15 Microbial genetics 3-16 3-17 Microbial genetics 3-19		Mouse Biorhythms
3-12 Plant biorhythms 3-13 3-14 Vertebrate genetics 3-16 3-16 3-16 3-17 Microbial genetics 3-18 3-19	3-11	Cockroach Biorhythms
Plant biorhythms 3-13 3-14 Vertebrate genetics 3-15 Microbial genetics 3-18 3-19		Drosophila Pupal Eclosion
3-14 Vertebrate genetics 3-15 3-16 3-16 3-17 Microbial genetics 3-19	3-13	Capsicum Biorhythms
Vertebrate genetics3-153-163-17Microbial genetics3-183-19		Avena Biorhythms
3-16 3-17 3-18 3-19	3-15	Genetics of Morphologic Adaptation - Teleost
3-17 3-18 3-19		Drosophila Genetics
3-18 3-19		Amoeba Genetics
	3-18	E. coli Genetics
Frenomens		Lysogeny and Genetic Phenomena
3-20 Neurospora		Neurospora Genetics

Table 4-9 (page 2 of 3)

Research Objective	Research Category	Representative Experiments
Geosensitivity	Vertebrate geosensitivity	3-21 Body Orientation in Fish
	Invertebrate geosensitivity	3-22 Locomotor Responses in Drosophila
	Plant geosensitivity	3-23 Tropic Responses of Capsicum
		3-24 Root and Shoot Tropism - Avena
	Cell geosensitivity	3-25 Human Cell Sensitivity
Hemodynamics	Primate vascular dynamics	3-26 Circulatory Dynamics of Primates
Metabolism	Vertebrate metabolism	3-27 Primate Metabolic Balances
		3-28 Metabolic Rates in the Mouse
	Invertebrate metabolism	3-29 Toxic Effects in Daphnia
		3-30 Drosophila Enzyme Activity
		3-33 Dinoflagellate Luminescence and Photosynthesis
	Plant metabolism	3-31 Crown Gall Metabolism
		3-32 Seed and Seedling Metabolism - Avena
	Microbial metabolism	3-34 Intermediary Metabolism in Neurospora
Morphogenesis	Vertebrate morphogenesis	3-35 Frog Embryogensis in Weightlessness
		3-36 Teleost Development
	Invertebrate morphogenesis	3-37 Head Shape in Daphnia
		3-38 Sexual and Asexual Development in the Flatworm
		3-39 Life Cycle Completion in Drosophila
		3-43 Amoeba Morphogenesis

Table 4-9 (page 3 of 3)

Research Objective	Research Category	Representative Experiments
Membranein	Plant morphogenesis	3-40 Crown Gall Growth in Weightlessness
Morphogenesis (continued)	0	3-41 Avena Life-cycle Completion
		3-42 Amaranthus Germination Potential
		3-44 Bromeliad Streamer Formation
Parasitism	Microbial parasitism	3-45 Altered Mitolic Division Rates in Crown Gall Development
		3-46 Induced and Spontaneous Mouse
	Invertebrate parasitism	3-47 Endamoeba Life Histories
Renroduction	Vertebrate reproduction	3-48 Human Cell Aging in Tissue Culture
	Invertebrate reproduction	3-50 Planarian Regeneration
Radiation	Vertebrate radiation effects	3-49 Radiation Effects in the Primate

approach to program implementation was based on the more practical criterion, frequency of utilization of the test organisms. To arrive at an appropriate weighting for the 18 species used, each was scored for the number of times it was utilized in the program. The distribution of this utilization was then determined by listing the number of high, medium-, and low-priority usages which were represented. From these distributions, it was then possible to rank the individual organisms according to their importance as weighted by both high utilization and high experiment priority. The rating and distribution are shown in Table 4-10. For convenience, the table has been arbitarily broken into three separate parts which could represent three phases of program implementation. The first phase (26 experiments) would be composed of those most important in terms of efficient subject utilization and high experiment priority. The second phase (14 additional experiments) would add six new organisms, four of which were used in high-priority experiments. The third phase (10 additional experiments) would add seven new test organisms, one of which was used for a high-priority experiment, and the rest for medium- or low-priority investigations. The experiments associated with each of the three phases are listed in Table 4-11.

The suggested ranking will ultimately be modified on the basis of availability of both experiment and life support hardware. The list is of value, however, for planning the relative emphasis which should be placed on the individual items which require development to support this program.

4.3.4 Measurement Requirements

The measurements required for this research program have been separately identified, since these demonstrate the versatility of the facility to perform research other than that specifically programmed. To evaluate this versatility, the selected experiments were analyzed for required measurements. As in biomedicine, not all experiments selected had sufficiently detailed protocols to permit direct analysis. In those cases, protocols were developed to achieve the stated experiment objectives, and the individual measurements were then extracted. Results of these individual protocol analyses appear individually in Volume II in the form of experiment requirements summaries.

Table 4-10

RESEARCH PROGRAM IMPLEMENTATION PLAN	Experiment Frequency in AIBS* Priorities	* wo *	2 0 0 0 0 1	00000	1000107	
		*muib9M	0 7 7 0 0	777777	00	
		*4giH	ក្រុកស្នេកប	00	0-10000	1961
PLEMENTA		Reproduction	×	×	×	ber 19
		maitiaa1s4	××	×	×	December
M IM	res	Morphogenesis	× × ××	××	** * *	1 1
PROGRAN	Research Objectives	Metabolism	××××	××	×	NASr-132,
		Hem od ynamics	×			1 1
ARCH	searc	Geosensitivity	× × ×	×		Report to NASA,
ESE/	Re	Genetics	× ×	××	×	rt to
BIOSCIENCE R		Biorhythms	××××	××		
		Behavior	×× ×	××	××	AIBS
		Test	Drosophila Primate Avena Mouse Crown Gall Fish	Daphnia E. coli Neurospora Cockroach Capsicum Planarian	Amoeba Frog Dinoflagellate Ant Bromeliad Cell culture Amaranthus	*AIBS Priority Rating;
		Рһаѕе	H	п	目	*AIBS

Table 4-11 (page 1 of 3)
BIOSCIENCE RESEARCH IMPLEMENTATION PLAN

Research Objectives	Research Category	Repr	esentative Experiments
	Phase I Experiments		
Behavior	Vertebrate Behavior	3-1 3-2 3-3	and marginations, and an arranged to the state of
	Invertebrate Behavior	3-6	Drosophila Behavior
Biorhythms	Vertebrate Biorhythms	3-9 3-10	Primate Biorhythms Mouse Biorhythms
	Invertebrate Biorhythms	3-12	Drosophila Pupal Eclosion
	Plant Biorhythms	3-14	Avena Biorhythms
Genetics	Vertebrate Genetics	3-15	Genetics of Morphologic Adaptation - Teleost
	Invertebrate Genetics	3-16	Drosophila Genetics
Geosensitivity	Vertebrate Geosensitivity	3-21	Body Orientation in Fish
	Invertebrate Geosensitivity	3-22	Locomotor Responses in Drosophila
	Plant Geosensitivity	3-24	Root and Shoot Tropism - Avena
Hemodynamics	Primate vascular Dynamics	3-26	Circulatory Dynamics of Primates
Metabolism	Vertebrate Metabolism		Primate Metabolic Balances Metabolic Rates in the mouse
	Invertebrate Metabolism	3-30	
 8	Plant Metabolism	3-31 3-32	Crown Gall Metabolism Seed and Seedling Metabolism - Avena
Morphogenesis	Vertebrate Morphogenesis	3-36	Teleost Development
	Invertebrate Morphogenesis	3-39	Life Cycle Completion in Drosophila

Research Objectives	Research Category	Repr	Representative Experiments		
	Plant Morphogenesis	3-40 3-41	Crown Gall Growth in Weightlessness Avena Life-Cycle Completion		
Parasitism	Microbial Parasitism		Altered Mitotic Division Rates in Crown Gall Development Induced and Spontaneous Mouse Leukemia		
Reproduction	Vertebrate Reproduction	3-48	Human Cell Aging in Tissue Culture		
	Vertebrate Radiation Effects	3-49	Radiation Effects in the Primate		
	Phase II Experimen	nts			
Behavior	Invertebrate Behavior	3-4	Daphnia Behavior Patterns		
		3-5	Cockroach Behavior Patterns		
Biorhythms	Invertebrate Biorhythms	3-11	Cockroach Biorhythms		
	Plant Biorhythms	3-13	Capsicum Biorhythms		
Genetics	Microbial Genetics		Coli Genetics Lysogency and Genetic Phenomena		
Geosensitivity	Plant Geosensitivity		Neurospora Genetics Tropic Responses of Capsicum		
Morphogenesis	Invertebrate Morphogenesis		Head Shape in Daphnia Sexual and Asexual Development in the Flatworm		
Reproduction	Invertebrate Reproduction	3-50	Planerian Regeneration		
	Phase III Experime	nts	Maria de la Carta de		
Behavior	Invertebrate Behavior	3-7 3-8	Ant Behavior Dinoflagellate Behavior		
Genetics	Invertebrate Genetics	3-17	Amoeba Genetics		

Table 4-11 (page 3 of 3)

Research Objectives	Research Category	Repr	Representative Experiments	
Geosensitivity	Cell Geosensitivity	3-25	Human Cell Sensitivity	
Metabolism	Invertebrate Metabolism	3-33	Dinoflagellate Luminescence and Photosynthesis	
Morphogenesis	Vertebrate Morphogenesis	3-35	Frog Embryogenesis in Weightlessness	
	Invertebrate	3-43	Amoeba Morphogenesis	
	Plant Morphogenesis	3-42	Amaranthus Germination Potential	
		3-44	Bromeliad Streamer Formation	
Parasitism	Invertebrate Parasitism	3-47	Endamoeba Life Histories	

For convenience, a consolidated list of measurements is shown in Table 4-12. This permits the reader to grasp the overall research potential of the total program. They are also presented in matrix form in Appendix A.

In general, measurements required for a comprehensive bioscience laboratory are quite similar to those detailed for biomedicine. Measurement equipment must often be designed specifically for use in the individual biomodules which contain the environmental control and life support units for the subject organisms. There is less commonality of equipment than might be expected, however, either because health considerations for the crew dictate isolation of equipment or because the circumstances of the bioscience experiments. The possibility of animal-borne diseases transmissable to man, especially in the unknown environment of a spacecraft laboratory, suggests that a relatively complete separation of human and animal research facilities is the most prudent and conservative design approach.

Certain equipment items which make measurements on body fluids, exudates, or excretions which clearly can be utilized for all programs. Similarly there are remote measurements such as electrocardiograms, and electroencephalograms which could utilize common equipment, provided separate body sensors were employed, but even in these cases it is expected that

Table 4-12 BIOSCIENCE RESEARCH PROGRAM MEASUREMENTS LIST

General, Environmental

Air particle count Ambient air pO2, pCO2, pN2 Ambient temperature Background radiation levels Body mass Chemical trace contaminants Cine photography Fluid intake (volume) Food intake (mass volume) Light intensity Light microscopy Photomicrography Relative humidity Specimen mass Still photography Time lapse photography

Animal physiology

Arterial blood pressure Blood flow Body fat Body temperature Bone density Brain temperature Capillary blood flow Cardiac output Electroencephalography Electromyography Electrooculography Ergometry Expired air pO2, pCO2 Galvanic skin response Heart rate Intracranial pressure Lean body mass Metabolic rate Radiation dose Respiratory rate Skin temperature Total body water Venous pressure

Biochemical, Metabolic Plant Physiology

Blood bicarbonate Blood hematocrit Blood hemoglobin Blood plasma volume Blood RBC mass Blood total nitrogen Blood urea nitrogen Blood volume Plasma albumin Plasma globulins Serum calcium Serum chloride Serum creatinine Serum phosphorus Serum potassium Serum sodium Urine acidity Urine calcium Urine creatine Urine creatinine Urine microscopic examination Urine osmolality Urine pH Urine phosphorus Urine protein Urine steroids Urine sugar Urine total nitrogen Urine volume

Animal/Plant Behavior

Behavior performance Biological rhythms Body orientation Development Gross body activity Reproductive behavior

Embryo/plant development Germination rate Luminescence Photosynthesis rate Tropisms

General Biological

Chromosome analysis Growth rate Metabolism rate RNA/DNA kinetics

Biochemical, Metabolic

Fecal calcium Fecal mass Fecal nitrogen schedules and crew availability may require a certain redundancy. A more detailed analysis of these commonalities is presented in another section of this report.

4.3.5 Equipment Requirements,

Bioscience equipment requirements fall into two broad categories. The first of these is equipment for the housing and care of the biological organisms:
"Organism Environmental Control/Life Support Modules." Included in this category are the systems and subsystems required for the modular organism EC/LS housing configurations. The second equipment category encompasses a broad range and variety of analytical instruments, the need for which depends on the degree of sophistication of the measurements and observations to be performed. The need for these analytical instruments will also depend upon the results of engineering tradeoff studies concerning biological sampling rates, in-flight analyses of samples, in-flight storage of samples, and return of samples for terrestrial laboratory analyses. Equipment in the two categories is listed in Table 4-13. Brief descriptions of each item are provided in Section 7.

In the EC/LS equipment category, the 50 plant and animal experiments could be housed in 20 modular housing enclosures ranging from special primate modules through fish aquariums, insect chambers, and cell-culture containers, to plant growth chambers and seed-culture tanks. Not included within the context of experiment housing are such general animal-holding facilities as cages for backup test animals. All this modular equipment requires supporting research and development technology. Some items are now in the prototype and test stage. Most such equipment has been conceived or designed originally as single-flight experiment packages, generally adapted to unmanned, near-Earth orbiting satellites. However, with minor modifications, these equipment modules can be adapted to a manned orbiting biotechnology laboratory, maintaining most of those features which would tend to minimize requirements for constant or overly frequent, direct human participation in support of the experiment. Such minimum-attention features are highly desirable because of the tremendous overall demands expected on the time of each scientist-astronaut.

Table 4-13

BIOSCIENCE RESEARCH EQUIPMENT LIST

General Equipment

Autoclave Automatic cell counter Automatic plate scanner/counter Automatic physiological gas monitor Bone densitometer Camera, cine Camera, plate film Camera, roll film Camera, polaroid Camera, video Centrifuge, organism Centrifuge; refrigerated, high speed Clinostat Colorimeter Dosimeters Dry sterilizer Drying oven Electroanalytical apparatus Electrophoresis apparatus Ergometer, primate Ergometer, rodent Freezer Gas bottles Gas chromatograph Histology kit Incubator Liquid gasses Lyophilizer Mass measurement device (macro) Mass measurement device (micro) Mass spectrometer Micromanipulator Microscope, light, compound Microscope, electron Microscope, dissecting Microtome Mixer/shaker Polarographic gas sensors

Recorder, oscillographic

Recorder, voice

Recorder, potentiometric, linear

Refrigerator
Spectrophotometer, IR
Spectrophotometer, UV-Visible
Ultrasonic cleaner
Vacuum desiccator

Organism EC/LS modules

Amoeba culture chamber Ant growth chambers Bacteria chemostat Bacteriophage flight package Daphnia containers Drosophila biorhythm module Fish aquarium Frog-egg flight package Insect chamber Mouse calorimetric module Mouse reproduction module Planaria container Plant growth chambers Pocket-mouse ECS module Primate calorimetry module Primate godesic module Seed germination tank Seedling culture tank Tissue culture module

The various systems planned and under development for the EC/LS housing modules involve at least 50 subsystem equipment items, most of which are clustered or configured into automated packages. Thus, measuring equipment for monitoring animal bioelectrical signals such as ECG, EEG, EOG, EMG, ZPG, GSR, temperatures, and pressures can be condensed into biotelemetry packages. The primate automated hemodynamic subsystem is comprised of several components, including optical densitometers, a refractometer, a mechanical drive pump, and a infusion/withdrawal syringe. A matrix chart (Appendix A) shows the type of subsystem equipment required by each of the organism EC/LS modules. Although the commonality of a type of subsystem is apparent, this does not imply the commonality of a specific subsystem design. Clearly, the nutrient supply and dispenser for a mouse reproduction module will bear no resemblance to the same subsystem for a flatworm container. On the other hand, a temperature-control subsystem will have broad application.

Finally, the general analytical and analytical support equipment consists of approximately 50 items, including several sophisticated spectral-analysis instruments requiring further development for space-flight rating. The equipment associated with each of the experiments is shown in Appendix A. Various combinations and condensations of these instruments, based on certain shared principles and/or components, have been suggested. It may not be feasible to include certain of the suggested instruments for the earlier orbiting laboratory flights. The chief instrument in this questionable category is the electron microscope, which has large voltage requirements. If electron microscopy analyses are not included as a part of early orbital-flight experiment performances, the requirements placed on such equipment items as the microtome and histology kit will be considerably less, because ultrathin tissue sections will not be needed.

Keen visual observations and a laboratory notebook can, of course, replace a large category of equipment items, such as TV, movie, and Polaroid cameras. But the time requirements placed on the astronaut-scientist as observer will increase enormously without such image recorders.

4.4 LIFE SUPPORT AND PROTECTIVE SYSTEMS RESEARCH

The basic goal of this research, both in Earth and manned Earth-orbiting laboratories, is to advance the technology that supports man during long exposures to the space environment. This study has concerned itself with that research which, when performed in the unique environment of an orbiting laboratory, will contribute to reaching this goal. Identification of requirements was accomplished in the same manner as previously discussed. However, whereas research goals, objectives, and categories had been suggested in the literature in the other areas, in life support and protective systems, no similar organized approach was found. The rationale which guides the study team in the preparation of the research program is described in the following paragraphs.

4.4.1 Research Objectives

A plan of required research for life support and protective systems was structured by first identifying the objectives, information needs, and research categories. It became apparent that, because this research was extensively concerned with hardware development, the concept of information needs and research categories did not provide a natural subdivision of research activities. Research was therefore categorized by "systems" and the associated "functional requirements."

From the review of the literature and current research and development programs, three clear objectives were aimed at "advancing the technology of the support of man":

- 1. Flight qualification and verification of integrated life support systems.
- 2. Flight qualification and verification of subsystems and components.
- 3. Research in basic technology to advance Items 1 and 2.

Emphasis was placed on providing closed integrated life support systems for manned space vehicles. The degree of closure was determined by the capabilities of subsystems and components to regenerate such consummables as oxygen and water. The need for this equipment is understood when the logistics problems presented by resupply of consummables on long Earth-orbiting missions is considered. Other important portions of the system were also investigated.

The many functional requirements of the systems which are part of an integrated system may be satisfied in a variety of ways. For example, in a water condenser - separator associated with humidity control, at least eight methods have been reported in the literature as being actively investigated or recommended for research. Whichever approaches are finally considered to be most promising as a result of Earth laboratory research will require further testing in the orbiting laboratory. The need for this type of subsystem and component testing in the operational environment will make up a large portion of the research program in this area.

Extended mission time calls for increased reliability of equipment. This is particularly true of the life support and protective system equipment, since the success of the mission depends on its proper functioning. Increased functional reliability may be achieved through design and proper maintenance. Effectiveness of maintenance may be tested only in space, since only there may the man-machine interface be studied in an operational environment.

Research in basic technology may be conveniently categorized and subdivided by identifying it with the life support and protective systems and functional requirements which it supports. These subdivisions therefore, provided a natural basis for categorizing this research. As in the area of subsystems and components, several basic technology problems are associated with a functional requirement. The majority of research is concerned with physical phenomena, basic to the operation of life support and protective systems, about which little is known or understood in the weightless environment.

Results of structuring the life support and protective systems research requirements are shown in Tables 4-14 through 4-16. The experiments and hardware or test status are discussed in following paragraphs.

4.4.2 Experiment Selection

Once the research required in this area was identified and categorized, the next step was to select representative experiments which could be analyzed for laboratory and support equipment requirements. The ground rule followed was that the experiment needed the space environment to provide

BIOTECHNOLOGY LABORATORY LIFE SUPPORT INTEGRATED SYSTEM EXPERIMENTS Table 4-14

			Hardware	
System	Functional Requirement	Experiment	Status**	2
Advanced integrated EC/LS	Life support	4-1Advanced Integrated Life Support System I 4-2Advanced Integrated Life Support System II	×	
Integrated EC/LS and power systems	Life support and power	4-3Integration of Radioisotope Power and Environmental Control/Life Support	×	
Animal EC/LS	Life support	4-4Advanced Integrated Life Support Systems for Animals	×	
AH EC/LS systems	Maintenance	4-5Maintenance and Repair in Zero G	×,	

*1. Basic research and development stage.
2. A working prototype subsystem.
3. Prototypes have been integrated and tested in a simulator.
4. Prototypes have been integrated and tested successfully in a manned simulator.
5. Flight-tested in Mercury, Gemini, and/or Apollo.

BIOTECHNOLOGY LABORATORY LIFE SUPPORT SUBSYSTEM OR COMPONENT EXPERIMENTS Table 4-15 (Page 1 of 5)

Hardwar Status* Farnced Two-gas Atmosphere-Supply and -Control system vanced Fluid Management and Gaging Subsystem vanced Fluid Management and Gaging Subsystem subcritical Storage and Supply Supercritical Storage and Supply Gaseous Storage and Supply Chemical Storage and Supply Gaseous Storage and Supply Chemical Storage and Supply Chemical Storage and Supply Refrigeration/Reliquefaction Subsystem x vanced Atmosphere Supply Subsystem ctrolysis Methods and Components G. E. Doubie-Membrane Electrolysis Unit ROH Liquid Electrolysis Unit ROH Liquid Electrolysis Unit Rotating Hydrogen Diffusion Cell Restribere Former Acetylene Former Acetylene Former Acetylene Former Acetylene Former Methane Cracker Hydrogen Stripper Sabatier Acetylene Stripper Social Electrolyte X Notice Coracker X Social Electrolyte Social Electrolyte Social Electrolyte Social Electrolyte X Social Electrolyte Social Electrolyte Social Electrolyte X Social Electrolyte Social Electrolyte					
Two-gas control Subsystem Gaging 4.7Advanced Two-gas Atmosphere-Supply and -Control Subsystem Mitrogen and oxygen 4.8Atmosphere-Supply Methods and Components supply 3. Gaseous Storage and Supply 3. Gaseous Storage and Supply 4. Chemical Storage and Supply 5. Refrigeration/Reliquefaction Subsystem A+9Advanced Atmosphere Supply Subsystem A+9Advanced Atmosphere Supply Subsystem Coxygen supply by 4-10Electrolysis Methods and Components electrolysis 1. G.E. Double-Membrane Electrolysis Unit A-11Water Electrolysis Subsystem Coxygen recovery 4-11Water Electrolysis Subsystem A-11Water Electrolysis Subsystem Coxygen recovery 4-11Water Electrolysis Subsystem A-11Water Electrolysis Subsystem Coxygen recovery 4-11Water Electrolysis Subsystem Coxygen Recovery A-11Water Electrolysis Subsystem Coxygen Recovery	Constitution	Functional	There is a constant	Hardwar Status*	
Gaging 4-6Advanced Two-gas Atmosphere-Supply and -Control Subsystem Gaging 4-7Advanced Fluid Management and Gaging Subsystem Nitrogen and oxygen 4-8Atmosphere-Supply Methods and Components 1. Subcritical Storage and Supply 2. Supercritical Storage and Supply 3. Gaseous Storage and Supply 4. Chemical Storage and Supply 5. Refrigeration/Reliquefaction Subsystem A+9Advanced Atmosphere Supply Subsystem Coxygen supply by 4-10Electrolysis Methods and Components electrolysis 1. G.E. Double-Membrane Electrolysis Unit 2. Water-Vagor Cell 3. Water-Vagor Cell 4-11Water Electrolysis Subsystem Coxygen recovery 4-11Water Electrolysis Subsystem Coxygen recovery 4-12Oxygen-Recovery Methods and Components 5. Acetylene Former 7. Babatier 7. Acetylene Former 7. Acetylene Former 7. Acetylene Former 7. Babatier 7. Acetylene Former 7. Babatier 7. Acetylene Former 7. Acetylene Former 7. Babatier 7. Acetylene Former 7. Babatier 7. Acetylene Former 7. Babatier 7. Babatier 7. Acetylene Former 7. Babatier 7. Babatier 7. Babatier 7. Babatier 8. Babatier 8. Babatier 9. Acetylene Former 9. Methods and Components 9. Methods and Components 9. Babatier 9. Acetylene Former 9. Methods Cacker 9. Babatier	System	reduit ement	rxperiment	5 3	
4-7Advanced Fluid Management and Gaging Subsystem 4-8Atmosphere-Supply Methods and Components 1. Subcritical Storage and Supply 2. Supercritical Storage and Supply 3. Gaseous Storage and Supply 4. Chemical Storage and Supply 5. Refrigeration/Reliquefaction Subsystem 7. Gaseous Storage and Supply 7. Chemical Storage and Supply 8. Chemical Storage and Supply 9. Refrigeration/Reliquefaction Subsystem 1. G. E. Double-Membrane Electrolysis Unit 2. Water-Vapor Cell 3. KOH Liquid Electrolysis Unit 4. TRW Porous Electrolysis Unit 5. Rotating Hydrogen Diffusion Cell 7. Sabatier 7. Acetylene Former 7. Methane Cracker 7. Methane Cracker 7. Hydrogen Stripper 7. Solid Electrolyte 7. Mothane Cracker 8. Mothane Cracker 9. Mothane Cracker 10. Mothane Cracker 11. Mothane Cracker 12. Mothane Cracker 13. Mothane Cracker 14. Mothane Cracker 15. Mothane Cracker 16. Mothane Cracker 17. Mothane Cracker 18. Mothane Cracker 19. Mothane Cracker 10. Mo	Atmosphere supply and pressurization	Two-gas control	4-6Advanced Two-gas Atmosphere-Supply and -Control Subsystem	X	×
1 Subcritical Storage and Supply 2. Supercritical Storage and Supply 3. Gaseous Storage and Supply 4. Chemical Storage and Supply 5. Refrigeration/Reliquefaction Subsystem 4-9Advanced Atmosphere Supply Subsystem 4-9Advanced Atmosphere Supply Subsystem 1. G. E. Double-Membrane Electrolysis Unit 2. Water-Vapor Cell 3. KOH Liquid Electrolysis Unit 4. TRW Porous Electrode Unit 5. Rotating Hydrogen Diffusion Cell 7. Rotating Hydrogen Diffusion Cell 8. Rotating Hydrogen Diffusion Cell 7. Actylene Former 7. Acetylene Former 7. Acetylene Former 7. Acetylene Cacker 7. Model Stripper 7. Subosch 7. Model Cacker 7. Model Electrolyte 8. Mode		Gaging	4-7Advanced Fluid Management and Gaging Subsystem	×	
1. Subcritical Storage and Supply 2. Supercritical Storage and Supply 3. Gaseous Storage and Supply 4. Chemical Storage and Supply 5. Refrigeration/Reliquefaction Subsystem 4-9Advanced Atmosphere Supply Subsystem 4-10Electrolysis Methods and Components 2. Water-Vapor Cell 3. KOH Liquid Electrolysis Unit 4. TRW Porous Electrolysis Unit 5. Rotating Hydrogen Diffusion Cell 7. Rotating Hydrogen Diffusion Cell 7. ActilWater Electrolysis Subsystem 8. Methone Cracker 9. Meth		Nitrogen and oxygen			
ly by 4-9Advanced Atmosphere Supply Subsystem 1. G.E. Double-Membrane Electrolysis Unit 2. Water-Vapor Cell 3. KOH Liquid Electrolysis Unit 4. TRW Porous Electrode Unit 5. Rotating Hydrogen Diffusion Cell X 4-11Water Electrolysis Subsystem 4-12Oxygen-Recovery Methods and Components 1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Mothan Cracker 6. Solid Electrolyte 7. Mothan Cracker 7. Mothan Cracker 7. Mothan Cracker 8. Mothan Cracker 9. Mothan C		h 1ddns		×	**
ly by 4-10Electrolysis Methods and Components 1. G.E. Double-Membrane Electrolysis Unit 2. Water-Vapor Cell 3. KOH Liquid Electrode Unit 4. TRW Porous Electrode Unit 5. Rotating Hydrogen Diffusion Cell 4-11Water Electrolysis Subsystem 4-12Oxygen-Recovery Methods and Components 1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Methane Cracker 7. Methane Cracker 8. Methane Cracker 9. Solid Electrolyte 7. Methane Cracker 7. Methane Cracker 8. Methane Cracker 9. Methane Cracker 10. Methane Cracker 11. Methane Cracker 12. Methane Cracker 13. Methane Cracker 14. Hydrogen Stripper 15. Methane Cracker 16. Methane Cracker 17. Methane Cracker 18. Methane Cracker 19. Methane Cracker 20. Methane Cra			4-9Advanced Atmosphere Supply Subsystem	×	
1. G. E. Double-Membrane Electrolysis Unit 2. Water-Vapor Cell 3. KOH Liquid Electrolysis Unit 4. TRW Porous Electrode Unit 5. Rotating Hydrogen Diffusion Cell X 4-11Water Electrolysis Subsystem 4-12Oxygen-Recovery Methods and Components 1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Mollar Contracts 8. Mollar Contracts 9. Mollar Contracts 1. Mollar Contracts 1. Mollar Contracts 2. Mollar Contracts 3. Methane Cracker 4. Hydrogen Stripper 6. Solid Electrolyte 7. Mollar Contracts 8. Mollar Contracts 8. Mollar Contracts 8. Mollar Contracts 8. Mollar Contracts 9. Mollar Contracts 9		Oxygen supply by	4-10Electrolysis Methods and Components		
4-11Water Electrolysis Subsystem recovery 4-12Oxygen-Recovery Methods and Components 1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Methane Cracker 8. Kritery Kripper 9. Methane Cracker 1. Acetylene Former 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Methane Cracker 8. Kritery Kripper 8. Kritery Kripper 9. Kr					
recovery 4-12Oxygen-Recovery Methods and Components 1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Methane Controlyte 7. Methane Controlyte 8. Solid Electrolyte 9. Solid Electroly			4-11Water Electrolysis Subsystem	×	
Sabatier Acetylene Former Methane Cracker Hydrogen Stripper Bosch Solid Electrolyte X X X X			4-12Oxygen-Recovery Methods and Components		
			1. Sabatier 2. Acetylene Former 3. Methane Cracker 4. Hydrogen Stripper 5. Bosch 6. Solid Electrolyte 7. Molten Carbonate	×	¥

^{9 6 4} G

A working prototype subsystem.
Prototypes have been integrated and tested in a simulator.
Prototypes have been integrated and tested successfully in a manned simulator.
Flight-tested in Mercury, Gemini, and/or Apollo.

Table 4-15 (Page 2 of 5)

			Har	Hardware	
	Functional Requirement	Experiment	Sta	Status*	5
		4-13Integrated Oxygen-Recovery Subsystem 1. Waste By-Products Not Used by Attitude Control 2. Waste By-Products Used by Attitude Control	×		×
Atmosphere purification and control	Carbon-dioxide control	4-14Carbon-Dioxide Collection Methods and Components 1. LiOH Expendable Subsystem 2. Regenerative Molecular Sieve with Vacuum			×
		Desorption 3. Regenerative Molecular Sieve with O ₂ Recovery 4. Carbonation Cell 5. Magnesium Oxide 6. Solid Amine 7. Electrodialysis	× ×		v
		4-15Advanced Integrated Atmosphere-Purification and Thermal-Control Subsystem			×
	Trace-contaminant control	4-16Integrated Trace-Contaminant Control andMonitoring Subsystem			
		 Mass Spectrometer/Gas Chromatograph Catalytic Burner Isotope Toxin Burner Charcoal Absorption, Partidulate Filters, and 	××		× :
	Microbial control and	10 2			×
	monitoring	1. Filters Regenerated by Killing Microbes by Heat 2. Silver Ion Generator 3. Ultraviolet Light 4. Viable Sampling by Membrane Filtration 5. Optical and Resistance Measurement	×	×	× ×

^{*1.} Basic research and development stage.
2. A working prototype subsystem.
3. Prototypes have been integrated and tested in a simulator.
4. Prototypes have been integrated and tested successfully in a manned simulator.
5. Flight-tested in Mercury, Gemini, and/or Apollo.

Table 4-15 (Page 3 of 5)

Circ b now	Functional			var s*	1 1	1
oy sterm	Trailie Illa	Experiment	7 1	η 1	4	ړ
Thermal control	Humidity control	4-18Water Condenser-Separator Methods and Components				
		1. Condenser with Liquid Gas Separation by Porous				
				×		
		2. Condenser with Liquid Gas Separation by Wick			,	5
		Reat Exchanger 3 Condenser with Liquid Gas Senaration by			•	×
				×		
		4. Condenser with Liquid Gas Separation by				
		•	×			
		5. Condenser with Liquid Gas Separation by			,	
		Hydrophobic/ Hydrophilic		•	×	
			*			
		7. Condenser with Liquid Gas Separation by	4			
		Membrane	×			
		8. Vapor Electrolysis	×			
	Process cooling	4-19Advanced Cooling System Methods and Components				
					n.	×
			ļ		~	5d
		5. Vapor Cycle 4. Absorbtion Cycle for Zero g	× ×			
		Cryogenic Cooling System	×			
	Process Heating	4-20Integrated Thermal Control System Utilizing Waste Heat and Electrical Energy	×			
Water management	Potable water and wash	4-21Water-Recovery Methods and Components				
	water	 Open- and Closed-Loop Air Evaporation System Vapor Pyrolysis System Vacuum Distillation Unit 	××	n	×	
		4. Membrane Diffusion/Permeation 5. Vapor Compression Unit	××			
			×			
				×		
		6. Reverse Osmosis	××			
		•	∢			

^{*1.} Basic research and development stage.
2. A working prototype subsystem.
3. Prototypes have been integrated and tested in a simulator.
4. Prototypes have been integrated and tested successfully in a manned simulator.
5. Flight-tested in Mercury, Gemini, and/or Apollo.

Table 4-15 (Page 4 of 5)

			Hardware
Svetem	Functional Requirement	Experiment	Status* 1 2 3 4 5
		Management Subsystem	×
		Service of the control of the contro	×
	Potability verification	4-23"Flight-Type" Potability Monitoring System	1
	To co o o o o o	4-24Waste-Management Methods and Components	!
Waste management	collection	1. Sleeve Attachment	×
			4
		_	×
		Collection and Removal	×
	Feces Processing		×
		5. Chemical Treatment System	×
			×
			: ×
			ł
		-,-	×
		Process	
		4-25Complete Waste Management Subsystem	*
	in leaves to a cr	4-26 Food Storage, Preparation, and Feeding Methods	
Food management	r coa sappri		×
		I. Freeze-aried room	×
			×
		3. Algae 4. Hydrogenomonas	×
	;	1	
Crew protection	Clothing space suit	4-2/Frotective Coming and Assemblies	
		1:::0	×
		2. Ames Suit	
			4 >
			∢.
		4. Umbilical Lanes 5. Advanced Cooled Suit Undergarments	×
			×
	Back pack	4-28EVA Suit and Blopack	

^{*1.} Basic research and development stage.

2. A working prototype subsystem.

3. Prototypes have been integrated and tested in a simulator.

4. Prototypes have been integrated and tested successfully in a manned simulator.

5. Flight-tested in Mercury, Gemini, and/or Apollo.

Table 4-15 (Page 5 of 5)

System	Functional Requirement	Equipment	Hardware Status* 1 2 3 4 5
	Gravity compensation	4-29Cardiovascular Conditioning and Maintenance	
			××
		 Gentrifugation Oscillation Advance Exercising System 	×××
	Personal hygiene	4-30Equipment and Procedures for Personal Hygiene	
		 Shower with Airflow-Directed Droplets Sponge Cleaner Mechanical Vacuum Shaver 	× × ×
	Fire protection	4-31Fire-Prevention and -Sensing in Zero G or Reduced Gravity	
	-	 Fire-Detection Systems Fire Extinguishers 	××
	Leakage	4-32Leak Detection	, ×
	Airlock	4-33Flexible Airlock	×
		4-34Airlock Gas Conservation	×

*1. Basic research and development stage.
2. A working prototype subsystem.
3. Prototypes have been integrated and tested in a simulator.
4. Prototypes have been integrated and tested successfully in a manned simulator.
5. Flight-tested in Mercury, Gemini, and/or Apollo.

Table 4-16 (Page 1 of 2)

BIOTECHNOLOGY		LABORATORY LIFE SUPPORT BASIC TECHNOLOGY EXPERIMENTS	ENTS
System	1	Experiment	Test Category* 1 2
Atmosphere emmly	Hast transfor	4-36 Density Drafiles of Linnide at and Near the	
Addre orondronner		Critical Region	×
Atmosphere supply/thermal	Liquid gas separation	4-36Capillary Studies	×
control/water management		4-37Kinetics and Dynamics of Gas Bubbles	×
		4-38Absorption of Gases by Liquids in Zero G	×
		4-39Gas-Free Liquid Maintenance	×
		4-40Static and Motion Tests of Interface Phenomena	×
Atmosphere supply/thermal	Solid and liquid retention	4-41Vapor Purge of Liquid Systems in Zero G	×
<pre>control/water management/ waste management</pre>	Solid transport by gas drag	4-42Transport of Solids by Gas Drag	×
Thermal control	Atmosphere circulation	4-43Solid-to-Gas Heat Transfer in Cabin Air Heating	×
		4-44Gas-to-Solid Heat Transfer in Cabin Air Cooling	×
		4-45Cabin Air Distribution and Control	×
Thermal control	Heat transfer	4-46Effectiveness of Thermal Insulation and Surface Coatings	×
		4-47Convective Heat Transfer at Zero G	×
		4-48Measurement of Solar Absorptivity and Thermal Emissivity of Various Materials by Spectrometry	×
Thermal control/water management	Heat transfer	4-49Pool Boiling in Long-term Zero G	×
Thermal control/protective system		4-50Effect of Wall Temperature, Ventilation Rate, Cabin Pressure, Gas Composition, and Grew Clothing on Comfort Level	×
Thermal control/water management	Condensation	4-51Condensing Heat Transfer and Condensation Rate in Heat Exchangers	×
Thermal control/water management/waste management/crew protection	Liquid transport by gas drag	4-52Transport of Liquids by Gas Drag	×

^{*1.} Man-machine interface and/or life support technological problem that requires the space environment to validate design concepts.
2. Basic data required in null gravity to substantiate analytical concepts and short-duration low-gravity ground-test results.

Table 4-16 (Page 2 of 2)

System	Functional Requirement	Experiment*	Test Category** 1 2
Water management Crew protection	Liquid and liquid mixing Fire prevention	4-53Water-Recovery System Pretreatment Mixing 4-54Composition Mixing and Heat Transfer 4-55Solids and Fluids Combustion	
All life support	Solid and liquid retention	4-56Retention Techniques for Liquids and Solids During Equipment Servicing, Repair, and Maintenance	*
	Solid transport by crew Liquid and solid recovery and retention	4-57Manual Transport of Solids 4-58Spillage Recovery and/or Cleanup	* *

*1. Man-machine interface and/or life support technological problem that requires the space environment to validate design concepts.

2. Basic data required in null gravity to substantiate analytical concepts and short-duration low-gravity ground-test results.

information in basic technology for LS/PS development or to provide flight qualification and/or verification of LS/PS hardware, either as an integrated system or at the subsystem and component level.

Review of current and proposed research resulted in identifying 58 representative experiments. Relatively few specific descriptions were found in the literature. Most of representative experiments described in this study were conceived by the study team to reflect the research experiments previously identified and the suggestions of the many scientists in this field. The representative experiments associated with each objective are shown in Tables 4-14 through 4-16. Brief descriptive information for each experiment is provided in Volume II. The experiment descriptions were prepared only to the level that permitted definition of major measurement and support equipment items. A summary of the total power, weight, volume, crew, and mission requirements is given in Section 5.

The several design approaches to a functional requirement mentioned earlier are recognized in the experiment list. For instance, the single representative experiment associated with trace-contaminant control covers four approaches. The laboratory requirements described for this experiment will permit research using any of those four methods. It is not implied that all four methods be investigated simultaneously, although this would of course provide distinct advantages. The particular method investigated in any specific mission program will depend on the extent to which it has been carried on in Earth laboratories.

A spinoff result of the study of experiment activities for the life support and protective systems research program was the production of information on the development or test status of equipment in each category. Inasmuch as this information will be useful to program planners, it is included in the research program tables. It was found that design concepts and hardware development could be described by five status levels:

- 1. Basic research and development.
- 2. Working prototypes
- 3. Prototypes integrated and tested in a simulator.
- 4. Prototypes integrated and tested successfully in a manned simulator.
- 5. Prototype flight-tested in Mercury, Gemini, and/or Apollo.

Much of the hardware now at Levels 1 and 2 may eventually be tested in a manned space-cabin simulator. Thus, new hardware could potentially reach Level 4. If this occurs, many of the experiments which have been described could be designed using the more advanced concepts. Typical life support subsystems that could fall into this category are those different functional methods for water recovery, oxygen recovery, waste management, and electrolysis that are listed as potential candidates in the experiment list. For example, water recovery from urine by vapor pyrolysis, vapor compression, or diffusion still could replace the air evaporation until experiment that currently is the only method that reached Level 4.

4.4.3 Program Implementation

The implementation of the LS/PS research program is directly dependent on the development of each item of LS/PS hardware and support equipment. The information and hardware development need is principally dependent on duration, crew size, and vehicle launch dates of future missions.

Tables 4-17 through 4-19 present a suggested phased research implementation program. The experiments proposed for Phase I are the highest priority experiments that are necessary to flight-verify regenerative LS/PS's that make possible the missions beyond Apollo and involve equipment furthest advanced in development status. For example, Experiments 4-14 and 4-15 can be used to flight-verify a regenerative molecular sieve for CO2 removal needed to replace an expendable LiOH unit for missions longer than 14 days. Other high-priority LS/PS experiments in Phase I include those that permit the flight-verification or qualification of a toxin burner (Experiment 4-16), a water-recovery system (4-21), an electrolysis unit for O2 supply (Experiment 4-10), and a commode (Experiment 4-24 and 4-25). These can be qualified for the space laboratory as separate subsystem experiments or as a part of an integrated LS/PS experiment. For example, Experiment 4-1, Advanced Integrated Life Support System - I with Priority 1 can be used to complete research on one or all of the previously mentioned subsystems. Apollo nonregenerative life support would be used as the primary system while the new equipment was being qualified. The basic technology experiments listed will aid in the future development of LS/PS's. The functional LS/PS methods for each process are many, as noted in the experiment list in Volume II.

Table 4-17 (page 1 of 3)

PROPOSED LIFE SUPPORT AND PROTECTIVE SYSTEM RESEARCH IMPLEMENTATION PROGRAM - PHASE I

Research Objective	System or Technology	Experiment*
Integrated system flight	Life support	4-1 - Advanced Integrated Life Support System I
verification and/or qualification	Maintenance	4-5 - Maintenance and Repair in Zero G
Subsystem flight verification and/or qualifica-	Two gas control	4-6 - Advanced Two-Gas Atmosphere-Supply and Control Subsystem
tion**	Nitrogen and oxygen supply	4-8 - Atmosphere Supply Methods and Components
	Oxygen supply by electrolysis	4-10 - Electrolysis Methods and Components
	Carbon dioxide control	4-14 - Carbon-Dioxide Collection Methods and Components
		4-15 - Advanced Integrated Atmosphere-Purification and Thermal-Control Subsystem
	Trace Contaminant control	4-16 - Integrated Trace-Contaminant-Control and Monitoring Subsystem
	Humidity control	4-18 - Water Condenser-Separator Methods and Components
	Process cooling	4-19 - Advanced Cooling System Methods and Components
	Potable water and wash water	4-21 - Water-Recovery Methods and Components 4-22 - Regenerative Water-Management Subsystem
	Feces and urine collection	4-24 - Waste-Management Methods and Components 4-25 - Complete Waste-Management Subsystem
	Food supply	4-26 - Food Storage, Preparation, and Feeding Methods

Table 4-17 (page 2 of 3)

Research Objective	System or Technology	${ m Experiment}^*$
	Clothingspace suit	4-27 - Protective Clothing and Advanced Space Suit Assembly
	Back pack	4-28 - EVA Suit and Biopack
	Fire protection	4-31 - Fire Prevention and Sensing in a Zero or Reduced Gravity
	Leakage	4-32 - Leak Detection
Basic technology***	Liquid gas separation	4-39 - Gas-Free Liquid Maintenance
	Solid liquid retention	4-41 - Vapor Purge of Liquid Systems in Zero G
	Solid transport by gas drag	4-42 - Transport of Solids by Gas Drag
	Atmosphere circulation	4-43 - Solid-to-Gas Heat Transfer in Cabin Air Heating
		4-44 - Gas-to-Solid Heat Transfer in Cabin Air Cooling
	Atmosphere circulation	4-45 - Cabin Air Distribution and Control
	Heat transfer	4-46 - Effectiveness of Thermal Insulation and Surface Coatings
		4-47 - Convective Heat Transfer at Zero G
		4-48 - Measurement of Solar Absorptivity and Thermal Emissivity of Various Materials by Spectrometry
	Condensation	4-51 - Condensing Heat Transfer and Condensation Rate in Heat Exchangers
	Liquid transport by gas drag	4-52 - Transport of Liquids by Gas Drag

Table 4-17 (page 3 of 3)

Research Objective	System or Technology	${ m Experiment}*$
	Solid and liquid retention	4-56 - Retention Techniques for Liquids and Solids During Equipment Servicing Repair, and Maintenance
	Solid transport by crew	4-57 - Manual Transport of Solids
	Liquid solid recovery and retention	4-58 - Spillage Recovery and/or Cleanup

*System or subsystem that has reached a hardware status of 4 or 5 and a Priority 1 rating; also basic technology Priotiry 1 experiments.

**Subsystem can be included in an integrated experiment.

***Basic technology experiment can be included with a subsystem or integrated system experiment.

Table 4-18 (page 1 of 2)

PROPOSED LIFE SUPPORT AND PROTECTIVE SYSTEM RESEARCH IMPLEMENTATION PROGRAM -- PHASE II

Research Objective	System or Technology	Experiment*
Integrated system flight verification and/ or qualification	Life support	4-2 Advanced Integrated Life Support System II4-4 Advanced Integrated Life Support Systems for Animals
Subsystem flight verification and/or	Gaging	4-7 Advanced Fluid Management and Gaging Subsystem
qualification**	Nitrogen and oxygen supply	4-8 Atmosphere Supply Methods and Components 4-9 Advanced Atmosphere Supply Subsystem
	Oxygen supply by electrolysis	4-10 Electrolysis Methods and Components 4-11 Water Electrolysis Subsystem
	Oxygen recovery	4-12 Oxygen-Recovery Methods and Components
	Carbon-dioxide control	4-14 Carbon-Dioxide Collection Methods and Components
	Trace contaminant control	4-16 Integrated Trace-Contaminant Control and Monitoring Subsystem
	Microbial control and monitoring	4-17 Biological Control and Monitoring of Life Support Subsystems
	Humidity control	4-18 Water Condenser-Separator Methods and Components
	Process cooling	4-19 Advanced Cooling System Methods and Components
	Process heating	4-20 Integrated Thermal Control System Utilizing Waste Heat and Electrical Energy
	Potable water and wash water	4-21 Water-Recovery Methods and Components

Table 4-18 (page 2 of 2)

Research Objective	System or Technology	Experiment*
	Potability verification	4-23 "Flight-Type" Potability Monitoring System
	Feces and urine collection	4-24 Waste-Management Methods and Components
	Clothing-space suit	4-27 Protective Clothing and Advanced Space Suit Assembly
	Gravity compensation	4-29 Cardiovascular Conditioning and Maintenance Method
	Personal hygiene	4-30 Equipment and Procedures for Personal Hygiene
	Airlock	4-33 Flexible Airlock
Basic technology***	Liquid gas separation	4-36 Capillary Studies
		4-37 Kinetics and Dynamics of Gas Bubbles
		4-40 Static and Motion Tests of Interface Phenomena
	Heat transfer	4-49 Pool Boiling in Long-Term Zero G
		4-50 Effect of Wall Temperature, Veltilation Rate, Cabin Pressure, Gas Composition, and Crew Clothing on Comfort Level
	Liquid and liquid mixture	4-53 Water-Recovery System Pretreatment Mixing
	Fire prevention	4-54 Composition Mixing and Heat Transfer
		4-55 Solids and Fluids Combustion

^{*}Any pertinent system or subsystem experiment that is now in a hardware status of 1, 2, or 3 that attains a 4 rating at least 2 years before experiment must be made available; also basic technology Priority 2 experiments.

^{**}Subsystem can be included in an integrated experiment.

^{***}Basic technology experiment can be included with a subsystem or integrated system experiment.

Table 4-19

PROPOSED LIFE SUPPORT AND PROTECTIVE SYSTEM RESEARCH IMPLEMENTATION PROGRAM -- PHASE III

Research Objective	System or Technology	Experiment*
Integrated system flight verification and/or qualification	Life support	4-3 Integration of Radioisotope Power and Environmental Control/Life Support
Subsystem flight verification and/orqualification	Oxygen supply by electrolysis	4-10 Electrolysis Methods and Components
	Oxygen recovery	4-12 Oxygen-Recovery Methods and Components
		4-13 Integrated Oxygen-Recovery Subsystem
	Carbon dioxide control	4-14 Carbon-Dioxide Collection Methods and Components
	Microbial control and monitoring	4-17 Biological Control and Monitoring of Life Support Subsystems
	Potable water and wash water	4-21 Water-Recovery Methods and Components
	Feces and urine collection	4-24 Waste-Management Methods and Components
	Food supply	4-26 Food Storage, Preparation, and Feeding Methods
Basic technology	Heat transfer	4-35 Density Profiles of Liquids at and Near the Critical Region
		4-38 Absorption of Gases by Liquids in Zero G

^{*}Other system or subsystem experiments that are now in a hardware status of 1, 2, or 3 that attain a 4 rating at least 2 years before experiment must be made available. Also, basic technology Priority 2 experiments not listed in Phase I and II.

The Phase II experiments given in Table 4-18 list more advanced LS/PS's that are not as far along in development as those in Phase I. Additionally, Priority 2 basic technology experiments that are needed to support future development are given. Under normal funding conditions, the time required to bring a new LS/PS subsystem from the working prototype stage to the level of having been proven feasible as an integrated unit in a manned space-laboratory simulator would be from 3 to 5 years. Several more years could be needed to provide a flight-qualified unit developed from prototype equipment test data that have been evaluated in the manned simulators.

Experiment 4-1, Advanced Integrated Life Support System - I, must be flight-verified or qualified before Experiment 4-2 can be changed from Priority 2 to Priority 1. Additionally, before an LS/PS experiment can become a reality, it should at least reach hardware status Level 4 before considering building a flight-type, flight-weight system to be used as an experiment. The reasoning, of course, is that the system should have proven itself at least as a prototype on the ground before the cost of building a flight unit and testing it as an experiment on board a spacecraft is expended. Once Experiment 4-1 is qualified, then it can become the biotechnology laboratory baseline system for future tests of more advanced life support system experiments. The experimental systems, when proven, can be used for future spacecraft. Once other LS/PS's have reached Level 4 they could also be developed into flight-type equipment and flown as "piggyback" experiments while the baseline EC/LS is used for primary support until the more advanced system has been qualified. The basic technology experiments given in Volume II may be completed as part of (1) an integrated life support system test, (2) a subsystem test, or (3) as a separate piggyback experiment.

Table 4-19 lists many experiments that require a longer research and development lead time as well as others that could be conducted as a Phase III.

Experiments not performed as part of Phases I and II could be added into a Phase III program. The implementation program previously discussed will require reconfiguration and refinement once a firmer LS/PS experimental program has been developed and approved by the scientific and engineering communities.

It is interesting to note the 28 of the 29 subsystems and component-level experiments could be conducted piggyback in a 1-year period with a power level of less than 600 W. The 29th, airlock pumpdown experiment, requires up to lkW. The total weight of the experiment test specimens is approximately 1,600 lb. Only one crew member at a time is needed to support 26 of the 29 experiments for 16 hours a day of testing. Two crewmen at a time are required for the other three experiments. The testing period for the three experiments would be 4 hours/day for a 5-month period. The only pacing item that is required to complete the above is the experiment and support equipment availability as a function of vehicle availability and launch date. Those LS/PS experiment requirements that would vary with crew size were designed for a one-man crew to obtain the above power and weight values. Only one type of LS/PS at a time was assumed to be used for each functional method. The power levels and weight values for an LS/PS scaled for larger crews will also provide a practical experimental package. The weight, power, and volume values for these larger crews may be approximated by using the information given in Volume II.

4.4.4 Measurement Requirements

Knowledge of the measurement capability of a research facility is an asset in planning research which digresses from that upon which the facility design is based. Changes in anticipated activities may occur for a variety of reasons. A list of the measurements required by the life support and protective systems research described in this document is given in Table 4-20. Two categories of measurements are listed: those which apply directly to the performance of equipment under test and those which are general in nature.

Additional information is provided by the matrix charts in Appendix A, which show the commonality of the measurements. The most important and common measurements include pressure, temperature, power requirement, humidity, dew point, flow rates, time, gas composition, quantities of materials, water potability, leakage, metabolic output, velocity, and several other items. The data presented in the tables indicate the commonality of measurement type that exists for the presented LS/PS experiments as well as many specialized measurement requirements. These measurement requirements are representative of those for the type of LS/PS experiments that could be

Table 4-20

LIFE SUPPORT AND PROTECTIVE SYSTEMS RESEARCH MEASUREMENTS

Equipment Performance Measurements

Absorbtivity/emissivity

Absorption rate Boiloff rate

CO₂ removal rate Condensation rate Contaminant level

Demsity
Dew point
Diffusion rate

Efficiency or effectiveness Electrical conductivity

Flow rate
Friction/drag
Gas composition
Gas flow distribution
Gas generation rate
Heat requirement

Heat transfer rate/heat balance

Humidity

Leakage

Liquid composition Performance data pH and oxygen demand

Power level Pressure

Reflective index

Reliability and maintainability data

Surface tension Surface finish Temperature

Time

Use or delivery rate Vacuum requirement

Viscosity

Water conductivity
Water - removal rate

Wetting angle

General Measurements

Acceleration

Amps

Biomedical monitoring

Burning rate Centrifugal force

Chemical analysis of liquids

Chemical contaminants

Comfort criteria

Damping

Debris generation

Diet

Food storage and preparation

Food and water use rate

Ignition time
Ionized radiation

Metabolic rate

Microbial contaminants

Mobility

Motion characteristics

Motion picture Orientation

Palatability of food Surface shapes Total energy

Urine/blood/biological samples

Velocity/velocity profile

Volts

Waste generation rate Water generation rate

Table 4-21

LIFE SUPPORT AND PROTECTIVE SYSTEMS RESEARCH EQUIPMENT REQUIREMENTS

Performance Measurements Equipment

Bubble chamber
CO₂ sensor
Dew point meter
Diffusion columns
Flowmeter
Friction measuring device
Gas chromatograph

Humidity sensor Infrared spectophotometer Integrated reflectometer

Liquid gaging unit

Mass spectrometer
Metabolic measuring device
Optical density sensor
Optical pyrometer
Oscillograph
Pressure sensor
Reference background grid
Shadowgraph

Temperature sensor

Timer

General Support Equipment

Accumulator/holding tanks Accelerometer/velocity meter

Animal housing

Biomedical equipment

Biomedical monitoring equipment

Carbon bag

Chemical/microbial laboratory

Comfort simulator

Crew hygiene protection system

Crew special restraints

Cryogenic supply

Debris disposal/container Dielectrophorese device

Dosimeter

EC/LS back pack

Emergency backup EC/LS

Ergometer Fire detector

Fire extinguishing agents

Gas compressor

Gas liquid separator Hand pump

Heating element Ignition device Illumination device

Isotope heater Leak detector

Liquid container/tank

Log book

Microscope Movie camera Orientation device

Photo cell

Plastic squeeze container

Plumbing purge unit

Porous plate water separator

Pressure controller Pretreatment unit

Pumps, fans, and blowers

Pyrex flask

Radiation laboratory Radioisotope shielding

Refrigerator

Solid/gas separator

Special animal sampling equipment

Special clothing/space suits

Special space radiator

Still camera

Stiring unit

Test container with bladder Test food/water dispensers

Tether/umbilicals

Tools for maintenance and repair

Transport container Vacuum cleaner Watt/volt/amp meter

Work bench

Zero gravity scale

performed in a space laboratory. The most prominent measurements requirements are those needed to evaluate operational performance and reliability, as would be expected.

4.4.5 Equipment Requirements

Only a few of the 71 support equipment items necessary for the measurements required for the 58 LS/PS experiments have been fully developed for space flight application. The list of equipment is given in Table 4-21. Two categories of equipment are shown: performance measurement equipment and general support equipment. Matrixes of representative support equipment for each of the 58 LS/PS experiments that could be performed in a space laboratory are presented in Appendix A. The data presented in the tables indicate support equipment commonality that exists between LS/PS experiments as well as specialized equipment requirements. The equipment accuracies, weight, power, and volume requirements given were estimated using existing commercial technology as a guideline, except where information on flight-type equipment was available. The equipment description summaries indicate that a minimum of flight-type experiment-support equipment exists. This stresses the importance for the future development of flight-type support equipment for approved high-priority experiments.

The support equipment for measuring many of the more simple LS/PS operational performance characteristics appears to be available or under development. These support equipment items include pressure sensors, temperature sensors, CO₂ sensors, flowmeters, and watt meters. However, items such as gas analysis devices, water-potability analysis equipment, leak-detection/location devices, dew pointers, integrating reflectometers, optical density sensors, optical pyrometers, and other similar equipment included in the 71 support equipment items listed in Section 7 are not available and must be developed in conjunction with the experiments they support. It is estimated that 90% of the different types of support equipment required to complete the experiments must be developed. A description and status summary is given in Section 7 for most of the support equipment items required to make measurements.

Section 5

BIOTECHNOLOGY LABORATORY SUPPORT REQUIREMENTS

The biotechnology laboratory portion of a manned Earth-orbiting space station consists of the facilities and areas in the space vehicle in which research in the life sciences is conducted. The extent of those facilities, and thus the scope of the research which may be conducted in a specific space vehicle configuration, will be constrained by the mission objectives and the accommodation capabilities of the vehicle. The general nature of the requirements which the biotechnology laboratory will impose on any mission are discussed in this section.

As a further aid to planning activities, a summary of the requirements of each experiment is provided. Tables 5-1 through 5-4 cover the experiments in biomedicine, behavior, bioscience, and life support and protective systems, respectively. They are based on the equipment listed for each experiment on the Experiment Requirement Summaries which appear in Volume II.

Three points must be remembered when referring to these tables: (1) the power, weight, and volume information is based on estimates because, in many cases, detailed information concerning individual equipment items is not available; (2) the number of crew members involved and the duration of an experiment will depend on the protocol defined by the investigator; and (3) the requirements for each experiment do not take into account the commonalities which will exist in specific mission experiment programs.

The electrical power shown is the total power required by all the equipment associated with the experiment. In reality, it would never be as large as shown and would depend on the experiment protocol. The weight and volume requirements would be reduced by taking advantage of commonalities.

BIOMEDICAL EXPERIMENTS SUPPORT REQUIREMENTS SUMMARY Table 5-1 (Page 1 of 2)

1-1 Head Movement Effects 1-2 Otolith and Semicircular Sensitivity 1-3 Day-Night Cycle Alterations 1-4 Vestibular Electrical Actively 1-5 Circulatory Response to Exercise 1-6 Volume Effects on Arterial Pressur- Systems 1-7 Peripheral Venous Compliance 1-8 Cardiac Dynamics 1-9 Intraocular Arterial Blood Pressuri 1-10 Lower-Body Negative-Pressure Dev Evaluation 1-12 On-Board Centrifuge Evaluation 1-12 Sensitivity Carotid Sinus 1-14 Peripheral Arteriolar Reactivity 1-15 Blood Volume and Distribution 1-16 Carotid Baroreceptor Electrical Act 1-17 Direct Cardiac Output Measurement	Head Movement Effects Otolith and Semicircular Sensitivity Day-Night Cycle Alterations Vestibular Electrical Actively Circulatory Response to Exercise Volume Effects on Arterial Pressure Control Systems Peripheral Venous Compliance Cardiac Dynamics Intraocular Arterial Blood Pressuring Lower-Body Negative-Pressure Device Evaluation	340 70 5	190	265	•	40
	ar Sensitivity ations Actively to Exercise erial Pressure Control mpliance slood Pressuring	70 5 95	•	2	70	>
	ations Actively to Exercise terial Pressure Control mpliance llood Pressuring	95	33	e	.2	9
	Actively to Exercise cerial Pressure Control mpliance slood Pressuring	95	80	11	2	30
	to Exercise terial Pressure Control mpliance slood Pressuring		110	230	60	30
	ierial Pressure Control mpliance Nood Pressuring Pressure Device	23	92	٣	en	35
Peripheral Venous Cocardiac Dynamics Intraocular Arterial B Lower-Body Negative-Evaluation On-Board Centrifuge I Occlusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	mpliance slood Pressuring Pressure Device	20	40	=	m	90
Cardiac Dynamics Intraocular Arterial B Lower-Body Negative- Evaluation On-Board Centrifuge I Occlusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	llood Pressuring -Pressure Device	25	7	0.25	т	30
Intraocular Arterial B Lower-Body Negative- Evaluation On-Board Centrifuge I Occlusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	llood Pressuring -Pressure Device	12	0.25	1	7	30
Lower-Body Negative-Evaluation On-Board Centrifuge J Occlusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	-Pressure Device	ı	0.15	;	N	10
On-Board Centrifuge Doclusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output		35	35	8 0	က	30
Occlusive Cuffs Evalu Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	Evaluation	9	25	1.5	ю	20
Sensitivity Carotid Sin Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	ation	9	25	1.5	m	20
Peripheral Arteriolar Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	sni	2.1	27	1.5	4	20
Blood Volume and Dis Carotid Baroreceptor Direct Cardiac Output	Reactivity	12	.7	0.04	. 73	15
Carotid Baroreceptor Direct Cardiac Output	tribution	110	20	0.7	e	9
	Electrical Activity	40	20	5.5	7	92
	Measurement	82	125	236	7	20
1-18 Response to Shock Therapy	erapy	110	140	236	7	;
1-19 Pulmonary Mechanics		7	12	:	2	7
1-20 Respiratory Central		*	17	0.25	m	92
1-21 Blood Gas Exchange		14	17	28	7	9
1-22 Lung Cleansing		ν'n	180	11		09
1-23 Induced Pulmonary Infections	fections	14	200	16	1	01
1-24 Recovery from Non-Infectious Lung Trauma	fectious Lung Trauma	ĸņ	200	15		09
1-25 Intestinal Absorption		15	52	4.2	-	9
1-26 Motility and pH		25	40	2	,eò	20
1-27 Indexes of Renal Function	tion	210	09	7	72	120
1-28 Calculus Formation		230	140	17	-	10

Table 5-1 (Page 2 of 2)

Experiment Number*	Experiment Title	Power (W)	Weight (1b)	Volume (cu ft)	Crew Involved	Duration (min.)
1-29	Renal Infection	LC)	195	14	1	30
1-30	Energy Metabolism	2	45	6.5	2	15
1-31	Carbohydrate and Fat Metabolism	3.7	33	9	2	09
1-32	Protein Metabolism	38	13	2	2	09
1-33	Body Fluid Composition	280	170	14	က	09
1-34	Mineral Metabolism	10	40	7	2	;
1-35	Bone Density	25	45	1	2	30
1-36	Muscle Status	1	39	9	2	40
1-37	Electromyographic Evaluation	10	09	8.9	2	20
1-38	Fracture Healing	15	40	9	2	20
1-39	Induction of Pressure Atrophy	15	40	.9	-	120
1-40	Endocrine Assays	220	105	10	2	09
1-41	Thermal Regulation	12	48	7	2	99
1-42	Adrenal and Parathyroid Histopathology and Function	20	40	5. š		45
1-43	Gonad Histopathological Evaluation	i t	15	4		30
1-44	Leukocyte Replication	1	15	4	2	40
1-45	Erythrocytes Dynamics	15	25	4	2	99
1-46	Leukocytes Dynamics	ι'n	26	4	2	30
1-47	Platelet Dynamics	ú	. 92	4	2	30
1-48	Hemostasis	1 1	15	4	2	30.
1-49	Leukocyte Mobilization after Chemical Challenge	ĸ	195	4	1	30
1-50	Maximum Rate of Erethrocyte Production	5	195	4	æ	09
1-51	Wound Healing	190	106	234	ш	30
1-52	Microbial Evaluation of Environment	35	31	0.75	2	15
1-53	Microbial Evaluation of Crew Members	15	30	0.7	2	40
1-54	Air Sampling	15	30	0.7	2	30
33.7	Transmission of Charles Acres Mountains	č	r.	0.7	6	9.2

*Numbers correspond to those in Volume II.

BEHAVIOR EXPERIMENTS SUPPORT REQUIREMENTS SUMMARY Table 5-2

BEHAVIOR EXPERIMENTS SUPPORT REQUIREMENTS SUMMARY

Duration (min.) 9 25 25 30 9 2 2 20 20 9 40 40 Crew Volume (cu ft) 3.8 Weight (1b) 20 95 125 65 9 91 16 16 20 95 15 Power (W) 200 200 165 165 195 220 220 225 225 100 400 165 225 165 165 165 155 Restraint and Gross Force Generation Restraint and Fine Force Generation Orientation, Stability, and Restraint Communication and Recording Monitoring and Observation Experiment Title Chemical Sense Function Higher Mental Functions Work/Rest/Sleep Cycles Personnel Translation Psychomotor Function Intrapersonal Factors Intrapersonal Factors Somesthetic Function Water Management Volume and Layout Orientation Senses Waste Management Food Management Auditory Function Mass Translation Hygiene System Visual Function Interior Design Accessibility Recreation Clothing Experiment Number* 2-10 2-13 2-15 2-16 2-17 2-18 2-19 2-20 2-11 2-12 2-14 2-21 2-22 2-23 2-24 2-25 2-26 5-9 2-7 2-8 2-2 2-3 2-5 2-4 2-6

*Numbers correspond to those in Volume II.

BIOSCIENCE EXPERIMENTS SUPPORT REQUIREMENTS SUMMARY Table 5-3 (page 1 of 3)

Experiment No. *	Experiment Title	Power (W)	Weight (1b)	Volume (cu ft)	Duration (days)
3-1	Primate Behavior	1,800	575	200	100
3-2⊹	Mouse Growth and Behavior	225	272	15	360
3-3	Teleost Behavior	175	100	4	0.9
3-4	Daphnia Behavior Patterns	175	85	4	30
3-5	Cockroach Behavior Patterns	170	80	٣	30
3-6	Drosphila Behavior	200	110	3.5	14
3-7	Ant Behavior	200	110	4.5	09
3-8	Dinoflagellate Behavior	225	100	4	120
3-9	Primate Biorhythms	1,800	1,000	200	120
3-10	Mouse Biorhythms	30	190	12	06
3-11	Cockroach Biorhythms	235	140	70	06
3-12	Drosphila Pupal Eclusion	30	32	Ŕ	Ŕ
3-13	Capsicum Biorhythms	315	120	6	06
3-14	Avena Biorhythms	210	275	80	06
3-15	Genetics of Morphologic AdaptationTeleost	180	100	4	120
3-16	Drosophila Genetics	800	009	160	14
3-17	Amoeba Genetics	280	500	8.5	40
3-18	E. coli Genetics	2, 200	625	19	2
3-19	Lysogency and Genetic Phenomena	009	400	9.5	14
3-20	Neurospora Genetics	320	800	5.5	06

Table 5-3 (page 2 of 3)

Experiment		Power	Weight	Volume	Duration
. No. *	Experiment Title	(W)	(1P)	(ca ft)	(days)
3-21	Body Orientation in Fish	205	80	4	14
3-22	Locomotor Responses in Drosophila	200	110	48.5	14
3-23	Tropic Responses of Capsicum	135	390	œ	06
3-24	Root and Shoot TropismAvena	22	118	3, 5	06
3-25	Human Cell Sensitivity	80	18	4.0	14
3-26	Circulatory Dynamics of Primate	1,485	700	23	06
3-27	Primate Metabolic Balances	1,450	200	21	09
3-28	Metabolic Rates in the Mouse	2, 500	500	35	09
3-29	Toxic Effects in Daphnia	75	270	ĸ	30
3-30	Drosophila Enzyme Activity	9	50	3.5	14
3-31	Crown Gall Metabolism	100	155	ហ	06
3-32	Seed and Seedling Metabolism Avena	na 210	350	7	06
3-33	Dinoflagellate Luminescence and Photosynthesis	380	300	6	120
3-34	Intermediary Metabolism in Neurospora	1, 300	525	16	06
3-35	Frog Embryo Genesis in Weight- lessness	15	40	1.6	~
3-36	Teleost Development	195	115	5.5	120
3-37	Head Shape in Daphnia	100	275	4	30
3-38	Sexual and Asexual Development in the Flatworm	105	06	ហ	06
3-39	Life Cycle Completion in Drosophila	. 225	140	9	14
3-40	Crown Gall Growth in Weightlessness	s 150	300	7.5	06
*Numbers cor	*Numbers correspond to those in Appendix A.				

Table 5-3 (page 3 of 3)

Experiment No. *	Experiment Title	Power (W)	Weight (1b)	Volume (cu ft)	Duration (days)
3-41	Avena Life Cycle Completion	150	285	7	06
3-42	Amaranthus Germination Potential	100	250	4.5	360
3-43	Amoeba Morphogenesis	375	400	12	40
3-44	Bromeliad Streamer Formation	20	100	2	06
3-45	Altered Metotic Division Rates in Crown Gall Development	350	525	13	06
3-46	Induced and Spontaneous Mouse Leukemia	3, 500	1,000	44	360
3-47	Endamoeba Life Histories	200	100	6	40
3-48	Human Cell Aging in Tissue Culture	300	100	2.5	. 06 - 35 - 37
3-49	Radiation Effects in the Primate	1, 200	009	260	360
3-50	Planaria Regeneration	165	300	8.5	06

*Numbers correspond to those in Appendix A.

Table 5-4 (Page 1 of 3)
LIFE SUPPORT AND PROTECTIVE SYSTEM EXPERIMENTS
SUPPORT REQUIREMENTS SUMMARY

Experiment Number*	Experiment Title	Fower (W)	Weight (1b)	Volume (cu ft)	Crew Involved	Duration (days)
4-1	Advanced Integrated Life Support System I	400**	300**	15**	NA	Mission
4-2	Advanced Integrated Life Support System II	400**	300**	15**	NA	Mission
4-3	Integration of Radioisotope Power and EC/LS	:	**009	**59	2	30
4-4	Advanced Integrated Life-Support Systems for Animals.	200	430	400	2	180
4-5	Maintenance and Repair in Zero G	100	75	10	Ą	8
				ł	required	required
4-6	Advanced Two-Gas Atmosphere-Supply and Control Subsystem	ĸ	, c c	ď	, AZ	, O.
4-7	Advanced Fluid-Management and Gaging Subsystem	15	440(max)	6	- 7	40 hr/
	· Very a					Specimen
4-8	Atmosphere-Supply Methods and Components	15**	280**	4.5**	2	06
4-9	Advanced Atmosphere-Supply Subsystem	15**	**08	1.5**	,-,	30
4-10	Electrolysis Methods and Components	300**	25**	**	1	06
4-11	Water Electrolysis Subsystem	300**	25**	* * T	As	06
					required	
4-12	Oxygen-Recovery Methods and Components	20**	5**	0.5**	2	06
4-13	Integrated Oxygen-Recovery Subsystem	325**	30**	* *	2	· . 06
4-14	Carbon-Dioxide Collection Methods and Components	75**	25**	3**	2	06
4-15	Advanced Integrated Atmosphere-Purification and Thermal-Control Subsystem	150	Ç.		. 0	ું. • પ ર
4-16	Integrated Trace-Contaminant Control and Monitoring Subsystem	30**	** ** **	0. 0. %	· -) C
4-17	Biological Control and Monitoring of Life-Support Subsystems	300	'n	Ŕ		As
4-18	Water Condenser/Separator Methods and Components	15	10	~	a	required
4-19	Advanced Cooling System Methods and Components	30	50	7	, Q	40 hr/
4-20	Integrated Thermal-Control System Utilizing Waste Heat and Electrical Energy	See 4-3	300**	**01		30
4.21			4 4	; ;	. '	

^{*}Numbers correspond to those in Volume II. **Requirements per man.

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Table 5-4 (Page 2 of 3)

Experiment Number*	Experiment Title	Fower (W)	(1b)	(cu ft)	Involved	(days)
4-22	Regenerative Water-Management Subsystem	100**	**05	***	As required	Mission
4-23	Flight-Type Potability Monitoring System	35	15	1.5	1	30
4-24	Waste-Management Methods and Components	50	25	4	4	30
4-25	Complete Waste-Management Subsystem	100	50	6	2	30
4-26	Food-Storage, Preparation, and Feeding Methods	100**	**09		As required	30
4-27	Protective Clothing and Advanced Space Suit Assemblies	100	43	4	2	20 hr
4-28	EVA Suit and Biopack	100	43	₹	2	20 hr
4-29	Cardiovascular Conditioning and Maintenance	100	100	16	Crew	Mission
4-30	Equipment and Procedures for Personal Hygiene	45	15	30	Crew	40 hr
4-31	Fire-Prevention and Sensing in Zero G or Reduced-Gravity	150	0.8	۲-	2	4 hr/ Experiment
ું જ	Total Defeation	01	ų́Λ	0.5	73	7
4-56	Flexible Airlock	150	20	6	2	4 hr
4.34	Airlock Gas Conservation	1,000	100	150	8	* !
4-35	Density Profiles of Liquid At and Near the Critical Region	250	66	2.6	ı	40 hr
4-36	Capillary Studies	20	100	.60	=	30
4-37	Kinetics and Dynamics of Gas Bubbles	0.5	100	9	1	30
4-38	Absorption of Gases by Liquids in Zero G	50	109	7	1 ,	30
4-39	Gas-Free Liquid Maintenance	10	10	7		
4-40	Static and Motion Tests of Interface Phenomena	50	213	12	7	30
4-41	Vapor Purge of Liquid Systems in Zero G	100	100	80	ન ં(4 pr
4-42	Transport of Solids by Gas Drag	100	100	&		40 hr
4-43	Solid-to-Gas Heat Transfer in Cabin Air Heating	40	10	0.5	-	40 hr
4-44	Gas-to-Solid Heat Transfer in Cabin Air Cooling	40	10	0.5		40 hr
4-45	Cabin Air Distribution and Control	40	20	٣	.	30
4-46	Effectiveness of Thermal Insulation and Surface Coatings	35	170	50	2	Mission
*Numbers c	correspond to those in Volume II.				*	
wantening ner man						

Table 5-4 (Page 3 of 3)

24 hr/ test 4 hr/ Experiment 4 hr/ Duration (days) 80 hr 200 hr 20 hr Experiment 40 hr 5 hr Mission Mission 30 30 required Crew AB ~ ~ 2.25 0.2 Volume (cu ft) 0.8 0.5 2 00 180 œ Weight (Ib) 175 0 15 เก 100 100 36 300 15 15 2 5 Table 5-4 (Page 3 of 3) Power (W) 15 30 01 300 100 100 100 2 10 Effect of Wall Temperature, Ventilation Rate, Cabin Pressure, Gas Composition, and Crew Clothing on Condensing Heat Transfer and Condensation Rate in Retention Techniques for Liquids and Solids during Equipment Servicing, Repair, and Maintenance Measurement of Solar Absorptivity and Thermal Emissivity of Various Materials by Spectrometry Water Recovery System Pretreatment Mixing Composition Mixing and Heat Transfer Convective Heat Transfer at Zero G Pool Boiling in Long-Term Zero G Transport of Liquids by Gas Drag Experiment Title Spillage Recovery and/or Cleanup Solids and Fluids Combustion Manual Transport of Solids Heat Exchangers Comfort Level Experiment Number* 4-47 4-50 4-55 4-48 4-49 4-53 4-56 4-58 4-52 4-54 4-57 4-51

*Numbers correspond to those in Volume II.

The crew involved in the conduct of the experiments is realistic. However, in the bioscience experiments, because crew involvement will depend on the degree of automation provided and on the final experiment protocol, no attempt was made to estimate the requirement.

The "duration" for biomedicine and behavior is the time required on each occasion to make the observations. The experiments are expected to be performed throughout the mission. This time might be shortened as a result of the observations. The duration of the bioscience and life-support system experiments is the length of mission time required to obtain meaningful results.

5.1 LABORATORY EQUIPMENT COMMONALITIES

The description of common requirements for items of equipment was prepared for two reasons: (1) to facilitate the planning of research programs to maximize the utilization of the laboratory facilities (2) to guide research and development planners to the equipment which should receive earliest attention. The matrix charts in Appendix A describe the equipment commonalities within each research area. The commonalities across the four research areas are shown in Table 5-5. In a more detailed analysis of common equipment applications such as will be possible with a defined set of research objectives for a space mission, it will be necessary to consider the measurement ranges required by the various experiments which use a particular type of equipment. Typical of this problem are the spectrophotometers required by the biomedical and the bioscience research programs. Both areas require a unit which makes optical density measurements in the wavelength range of 200 to 700 mm. However, bioscience has a requirement for a unit which makes measurements in the range of 2.5 to 3.5 mm.

One common requirement not shown in the charts is that of animal life support systems for the bioscience and biomedical research programs. Whether or not the same animal subject may be used for more than one experiment will of course depend upon the nature of the research. Their life support systems impose large demands on the space-station resources; therefore, research in both areas should be conducted if an animal facility is to be installed.

Table 5-5
BIOTHECNOLOGY LABORATORY COMMON REQUIREMENTS

		Research A	rea	w
Equipment	Bioscience	Biomedicine	Behavior	LS/PS
Torque-force apparatus	,	x	x	
Human centrifuge		X	x	
Ergometer		X	X	X
Electroenecephalograph		X	x	X
Electrocardiograph		X	x	
Electromyograph		X	x	
Calorimeter, respiratory		X	X	
Animal housing and support	X	X		x
Voice recorder	X	X	X	1
Still camera	X	X	X	X
Motion picture camera	X	X	X	X
Film-processing equipment	X	X	X	x
Slide-straining apparatus	X	X		x
Bacterial culture equipment	X	X	:	X
Automatic cell counter	X	X		
Frozen storage unit	X	X		
Autoclave	X	X		
Incubator	X	X		x
Dry heat sterilizer	X	X		
Lyophilizer	X	X		
Radioactive tracers	X	X		
Gas chromatograph	X	X		X
Mass spectrometer	X			x
Mass-measurement device (macro)	×	x		x
Mass-measurement device	-	-		
(micro)	x	x		x
IR spectrophotometer	x	x		\ddot{x}
UV-Visible spectrophotometer	l x	x		^*
pH Meter	x	x		x
Fluid-handling equipment	x	x		x
Centrifuge, clinical	x	x	Ì	
Radiation dosimeter	x	x		

5.2 ELECTRICAL POWER REQUIREMENTS

The review of the electrical power requirements of the experiment or laboratory equipment has indicated that, in general, nominal power loads not exceeding 0.5 kVA for a single experiment may be expected. Furthermore, this load will only occur if all the measurement equipment associated with an experiment is operated simultaneously. In most cases, equipment

operations will be programmed to minimize instantaneous loads. The major exception to this low power demand of intermittently operated equipment is the operation of the human centrifuge. During periods of positive acceleration, this device may be expected to impose a load of approximately 5 kVA. Even this load will be acceptable with appropriate management of the electrical power system.

The operation of life support modules for experiment subjects required by biomedical and bioscience research will impose continuous loads. In the case of primates, these loads are in the order of 2.5 kVA. Loads of this magnitude must be identified and imposed as a requirement upon the design of the electrical power system for a space station.

5.3 DATA MANAGEMENT

If it is assumed that experiments are timelined to permit reasonable schedules and to provide programs for contingency rescheduling, the process of biological measurements cannot be considered a technological problem. Neither does the imposition of such practical constraints as operation with existing space-station equipment and within present range limitations, tend to alter these assertions to any marked degree.

This fortunate condition occurs because of two major factors. First, biological functions occur or change at a slow rate compared to most electrical or mechanical functions, and it is the rate of change or response time of a measurement that is a limiting quantity in determining whether a measurement may be accommodated on a given transmission link. The second factor, in part resulting from the first, is that the amount of data which must be processed and relayed to the ground is not particularly large nor is the arrival of data urgent.

5.3.1 Data Requirements

A review of the experiments and equipment recommended in the present study provided the source data for Table 5-6. It was determined that most parameters, consisting of position, pressure, temperature, rate, and event measurements, fall into the frequency range of less than 2-cps. Although

Table 5-6

DATA HANDLING

Frequency Response (cps)	Range 1 $(0 \ge f > 2 \text{ cps})$	Range 2 $(2 \ge f > 100 \text{ cps})$	Range 3 $(100 \ge f > 1000 \text{ cps})$	Range 4 (f > 1000 cps)
Potential number of		T.		s. De phys P
measurements	007	or more	or more	s or more
Preferred modulation	PCM/PM	PCM/FM	FM/FM	AM
technique	or PCM/PM	$_{ m PCM/PM}$	or PCM/PM	FM
		or FM/FM		PCM
Degree of handling				
difficulty		2	m _.	4
Source	Component/	Cardiograms Blood pressure	Cardiograms Radiosonde	Voice Television
			Myograms	Scan conversion

many were placed in this range because of inherent low-response characteristics, a number of the measurements in their raw form had considerably wider bandwidths. However, the nature of the desired outputs were of a form which would allow use of such signal-conditioning and conversion devices as log counters which would reduce bandwidth requirements.

Table 5-7, generated during a previous study, illustrates the case in handling data of this nature. Although test times for all technologies that were run on a random day were comparable, the number of bits resulting from the behavioral science experiments was less by several orders of magnitude than the number deriving from experiments in other technologies.

5.3.2 Data Handling

The analog parameters in Range 1 are currently being sampled at a rate between a theoretically possible 2 to a maximum of 10 samples/cycle before being quantized. The number of bits per sample range from 4 to 10 or greater, depending on the desired accuracy of the measurement involved. The encoded data are then gated by the data-acquisition subsystem programmer, together with words obtained in digital form, to produce a serial bit stream. Rates for this serial data, if compatibility is to be maintained with the Manned Space Flight Network (MSFN), are 1.6 and 51.2 kbits/sec.

These rate limitations account for the "degree of difficulty" figure attached to parameters falling into Range 3. The rating of 3 results not from any real difficulty in processing the data but from the impact on system capability if many such measurements are made. For example, 5 measurements at 1,000 cps, encoded to 8 bits/sec produce a serial rate of 40 kbits/sec absorbing 80 percent of the system capacity. If the number of channels to be handled is such that this percentage cannot be allocated, a new system utilizing subcarrier oscillators for the high-response measurements must be carried.

New PCM equipment, now coming into general use, has overcome these channel and rate limitations by incorporating flexible formatting which, simply stated, permits only those measurements to be acquired that are

Table 5-7

EXPERIMENT DATA SUMMARY--ONE DAY

Technology: Physical science Running time: 180 min.

Technology: Behavioral science (COMPARE test battery) Running time: 120 min.

Experiment		No. of	Bits Per		
No.	Data Type	Words	Words	Bit Rate	Total Bits
98	Digital	261	24	Not annicable	6 26 103
				on applicable	0. 20 % 10

Communications and navigation Technology: Communica Running time: 376 min.

Experiment No.	Duration (sec)	Data Type	No. of Channels	Sample Rate	Bits Per Sample	Bit Rate	Total Bits
1030	$4.8 \times 10^{3} $ 4.8×10^{3}	Analog Digital	190 2	10	100	1.52 x 10 ⁴ 200	7.3×10^{7} 9.6 × 10 ⁵

Total run time = 10 hrTotal bits = 9.4×10^7 Transmission time (at 1×10^6 bits/sec) = 1.5 min.

needed during the experiment in progress. Therefore, although both Ranges 2 and 3 indicate FM (subcarriers) as a potential technique, use of FM will gradually decrease because of these new PCM methods.

Data in Range 4 are those required for obtaining a verbal and visual record of the manner in which each experiment is performed and also may constitute all or part of the experiment information itself. The bandwidth necessary for acceptable voice is normally specified to be from 300 to 3,000 cps in its analog form. Digital voice specifications vary from 4,800 to 9,600 bits/sec for a vocoder. It is also possible to sample at slightly greater than twice the highest voice frequency and employ straight analog-to-digital conversion at 6 bits/sample (as planned on the Manned Orbiting Laboratory) which results in a considerably higher rate. The signal, whatever its form, may then amplitude- or frequency-modulate a carrier or subcarrier. To reduce the amount of data accummulated, space-station recorders should be voice-operated. They should be of the cassette type and provide 60 min. of storage.

Visual data may be in the form of still photographs, movies, or television. If photographic data are retained in their original form and returned at the end of the mission or during the mission with data recovery capsules (indicated as satisfactory by investigators), the question of transmission does not arise. However, should transmission of film frames be required because of some unforeseen circumstance, electronic scanning with use of cathode ray tubes and photodiodes becomes necessary and rates may become exceedingly high. For example, if a resolution requirement of 100 lines/mm, 35 mm² film, and an analog-to-digital conversion ratio of 5 bits/sample are assumed, 6.2 x 10⁷ bits are generated. It becomes readily apparent that if very many frames must be sent and the data are not operated upon with use of data-compression algorithms, a somewhat impossible communications problem may develop.

The return of television signals presents a somewhat analogous situation in that the amount of information generated is a function of frame rate and resolution. Typical video bandwidths range from 70 kHz used for telescope

pointing, to 500 kHz on Apollo, to 3 MHz for commercial TV, to 10 MHz for high-resolution industrial systems. Obviously, video greater than 500 kHz will present a serious problem on the MSFN.

5.3.3 Existing Data Facilities

As previously indicated, the MSFN unified S-band will accommodate PCM data at a maximum rate of 51.2 kbits/sec. Seven subcarriers normally used for EVA biomedical monitoring with center frequencies of from 4 to 12 kHz spaced 1.4 kHz apart may be calculated to handle signals with response requirements of from 70 to 175 kHz each, depending on the degree of error that can be tolerated. Three additional subcarriers with center frequencies of 65, 95, and 125 kHz provide data bandwidths of 1.95 kHz, 2.85 kHz, and 3.75 kHz, respectively. Finally, there is the 500-kHz bandwidth capability achieved by directly modulating the FM transmitter carrier and the 300-kHz to 2.3-kHz voice channel which modulates the 1.25-MHz voice subcarrier. These facilities, when compared with requirements, appear to be satisfactory.

5.4 ATTITUDE CONTROL REQUIREMENTS

Only in two research areas have research activities been found which will affect the space station attitude control system. Many of the bioscience experiments have indicated the desirability or necessity of gravitational fields in the order of 10^{-4} g, if meaningful data are to be obtained. The operation of a human centrifuge will perturb the stability of a space station, thereby requiring attitude-control compensation. The remainder of the research may be conducted in the normal operational status of the space station stability.

It appears impractical to ensure the low gravitational field requirements of the bioscience experiments on a manned space station. Although the desired low levels may be experienced briefly, many random disturbances will occur. Crew movement alone will produce changes. Operational requirements of the mission for attitude changes and attitude and pointing requirements of other research such as astronomy or Earth resources, will similarly affect the g-levels. In recognition of these facts, two solutions

have been proposed if the requirement cannot be relaxed in a manned space station. The first is a bioscience module within the space station which is isolated from the perturbations described above. The second is a separate, independent module operated in conjunction with a space station and visited periodically by the crew. Both of these solutions require large development programs and disturbing forces would still be exerted during crew observations.

Unless compensated for, centrifuge operation can significantly affect the spacecraft attitude hold, rate stabilization, and orientation. The most efficient means of eliminating the centrifuge effects is by use of such momentum storage devices as control moment gyros (CMG's). CMG's are primary control actuator candidates for intermediate, long-term space station configurations. These actuators have the desirable characteristics of linear control for attitude performance and momentum storage for the cyclic disturbances. To provide control for centrifuge operation, the CMG's provide an angular momentum vector which negates that of the centrifuge. Hence, the CMG size and configuration must include the angular momentum of the centrifuge and that defined for the mission requirements. For high-g centrifuge operation, such as a re-entry profile, the CMG size may become, for practical design, too large. Control for this case would be provided by the reaction jets.

5. 5 ENVIRONMENTAL CONTROL REQUIREMENTS

The life sciences research program will not impose special requirements on the space-station environmental control system. The separate life support modules for experiment subjects, while in general requiring special atmospheres, will be closed systems. Isolation of the area in which these facilities are located may be required to prevent contamination of the cabin atmosphere. In this situation, special provisions would be required to supply cabin atmosphere to the area. Similarly, the area in which fluids are handled may require isolation. However, as mentioned above, this isolation is not required by the research, and it is possible that the design of the life support modules and fluid handling facility will remove the need for isolation.

5.6 CREW SKILLS REQUIREMENTS

The contributive role to be played by scientist-crew members who are specialists in the various space-station experiment protocols has been described in several documents. (See Bibliography) The necessary or desirable crew skills associated with the research activities considered in this study reinforces those recommendations.

Results of this study and of others (e.g., NASA MOL and MORL) indicate that the primary skill classifications required would include the following: physician, microbiologist, biochemist, and physiologist to conduct, analyze, and interpret experiments, and an electromechanical engineer to maintain equipment and to conduct the LS & PS research.

An on-board physician would greatly enhance experiment capability throughout the mission. A major contribution would be to evaluate and interpret data as it is gathered and to recommend changes in experiment content and protocol. Further, he could supervise preliminary biochemical analyses on-board and thereby eliminate samples which do not show significant change. Samples which do indicate change would be stored on-board and analyzed in more detail on the ground. Detailed on-board analyses are not considered feasible for the 1971 time period because neither appropriate procedures nor flight-rated equipment will be completely developed. Furthermore, detailed analysis is costly in terms of crew time and other station resources. For the NASA MOL, approximately 3 hours/man/day were required for a 6-man crew to perform a comprehensive biomedical-behavioral experiment program. This estimate was based upon having a physician available to supervise and evaluate the program.

The primary skills mentioned above are definable at three levels: technician, general scientist, and specialist, with the latter requiring the most training and experience. The technician's role would be primarily to operate and maintain equipment; the general scientist would analyze and interpret data; the specialist would be responsible for evaluating the overall progress of the specified portions of the experiment program, make decisions, and redirect the program as required.

It is suggested that the crewmen for the biotechnology laboratory be selected largely from the current astronaut pool. With the possible exception of the physician, physiologist, and biochemist, it should not be necessary to select astronauts with specific academic backgrounds; instead they could be trained in the procedures of conducting, observing, and serving as subjects. The extent of cross-training required of analytic and diagnostic skills cannot be defined until experiments are detailed and scheduled and crew size is established. It is anticipated that the physician would have the overall responsibility for the experiment program.

SECTION 6

SUPPORTING RESEARCH AND TECHNOLOGY REQUIREMENTS

The need for a large research and development program of laboratory equipment was well known before this study was undertaken. The specific items of equipment which are applicable to life sciences research and which require development have been identified in this study. The Equipment Description Summaries in Section 7, include brief comments concerning the development requirements of each item of equipment. However, to facilitate the evaluation of the SRT program requirements, the items of equipment have been categorized according to their development status. The development status categores are first, currently available equipment requiring little or no adaptation; second, equipment currently under development; and finally, equipment requiring development but for which no development program could be identified. The laboratory equipment requirements of each of the areas of biomedical and human research, behavioral research, bioscience research, and life support and protective systems research, are listed in Tables 6-1 through 6-4. It should be noted that the lists do not include life support modules for animal or other organism experiment subjects or items under test in life support and protective systems research, since the need for these will depend on the experiment programs selected, and they will be a part of the experiment development. This is discussed in a later section.

The breadth of application of equipment items will be an important consideration in the selection of items which are to receive emphasis in the research and development programs. The information concerning the commonality of equipment listed in the tables will be of assistance in the selection process.

Table 6-1 BIOMEDICAL RESEARCH EQUIPMENT

Available Equipment

Electrocardiograph Otoscope

Electroencephalograph
Erogmeter
Limb pressure cuffs
Gas flowmeter
Specific ion electrodes
Hemoglobinometer
Temperature recorder

Ophthalmoscope Therimistor

Under Development

Aerosol particle analyzer Tilting motion chair
Ballistocardiograph Human centrifuge
Arterial pressure recorder Endoradiosonde
Automatic cell counter Gas chromatograph

Bone densitometer Impedance cardiograph

Carotid cuff Lower-body negative-pressure device

Muscle dynamometer Large mass measurement
pH meter Small mass measurement
Plethysmograph Nystagmographic goggles

Scintillation counter Oscilloscope
Spectrophotometer Ear oximeter

Chemical analyzers

No Development Underway

Autoclave Light microscope
Cardiotachometer Nebulizer chamber
Clinical centrifuge Nitrogen analyzer

Plethysmographic goggles Osmometer Humidity recorder Phoropter

Manometer recording Whole body volumeter

Table 6-2 BEHAVIORAL RESEARCH EQUIPMENT

Available Equipment

Audio tape recorder

Movie camera

Still camera

Video camera

Accelerometers and recorders

Color plates

Pure tone source

Equipment Under Development

Metabolic monitor

Torque-force apparatus

Standard dexterity apparatus

depth perceptometer

CRT and other displays

Display-associated controls

Analog Recorder

Critical flicker fusion (CFF) source

EMG sensors and recorders

GSR and EEG sensors and recorder

Ergometer

Equipment Not Now in Development

Veridicality test

Orthorater

Noise and vibration source

The measurement range of an item of equipment may vary from experiment to experiment. It is important that equipment developed in response to a common requirement cover all the measurement ranges required. In this regard, it is recommended that measurement ranges required by potential future applications of the equipment also be provided for in the development.

It is recognized that there are several methods by which some measurements may be accomplished. This study did not examine the variety of approaches. The equipment shown for a specific measurement was based on current practice within Earth laboratories or on planned space research activities. It is recommended that the variety of techniques and equipments for making a measurement be examined in depth as part of a development program. Some of the factors to be considered are efficacy of the method, degree of difficulty to calibrate and operate in the weightless environment, reliability and maintainability, data output format, and resources demands.

Table 6-3 BIOSCIENCE RESEARCH EQUIPMENT

Available Equipment

Automatic cell counter Movie camera Plate film camera Clinostat Polaroid camera Dosimeters Radiation shielding Electroanalytical apparatus Radiation source Electrophoresis apparatus High-speed oscillographic recorder Roll film camera Linear potentiometric recorder Video camera Voice recorder Miscellaneous

Equipment Under Development

Colorimeter Mass-measurement device (micro)

Gas chromatograph Organism centrifuge

Mass spectrometer Polarographic CO₂ and O₂ sensors

Mass-measurement device (macro) Primate exerciser/ergometer

Equipment Not Now in Development

Autoclave Liquid nitrogen Dewar flask

Automatic plate scanning Lyophilizer

counter

Bone densitometer Microtome Microtome

Colorimeter Mixer/shaker

Compound microscope Refrigerated high-speed centrifuge

Dry heat sterilizer Refrigerator

Drying Oven Rodent exerciser/ergometer

Electroanalytical apparatus Storegapie dissecting

Freezer Stereoscopic dissecting microscope

100201

Histology kit/slide cabinet Ultrasonic cleaner

Incubator UV-visible spectrophotometer

IR spectrophotometer Vacuum desiccator

Isotope tracer equipment Veterinary medical kit

Table 6-4 LIFE SUPPORT AND PROTECTIVE SUBSYSTEMS RESEARCH EQUIPMENT

Available Equipment

Illumination device Flow meter

Aerosol particle analyzer Accelerometer

MSA indicator tubes and Accumulator

hand-pump Holding tanks

Work bench for maintenance and repair Cryogenic supply

Tether CO₂ sensor

Umbilicals Emergency backup EC/LS

EC/LS back-pack

Test container with bladder

Special clothing Pressure controller

Space suits Plastic squeeze container

Pressure sensor

Temperature sensor

Heating element
Orientation device

Equipment Under Development

Mass-measurement device Biomedical monitoring equipment

Total aerosol collector Metabolic measuring device

Water analysis by total organic Leak detector

carbon (TOC) Radioisotope shielding

Dispersive total hydrocarbon Isotope heater

analyzer

Fire detector

Fire-extinguishing agents

Pretreatment unit

Debris disposal

Optical pyrometer

Solid/gas separator

Ergometer Solid/gas s

Equipment Not Now in Development

Mass spectrometer (for trace Nondispersive IR spectrophotometer contaminants)

Calorimeter or Coleman Reflectometer

spectrometer (junior model) Optical density sensor

Turbiditimeter Photo cell

Since this study addressed itself to the research and equipment requirements of the biotechnology laboratory in the space station, the subject of supporting ground programs (other than equipment development) was not considered. However, one cannot study the research to be performed in space without realizing that a large program of research on Earth must precede it. The majority of the research in life sciences is aimed at discovering and understanding the effects of weightlessness on the living organism. However, in the space laboratory several environmental factors in addition to the gravitational field will be changed. To a degree, each of these variables, except weightlessness, may be simulated in a laboratory on Earth. Therefore, those experiments which would be affected by these variables should be conducted in a simulator so that a reference frame will be available for interpretation of the results of the experiment conducted in space. This, of course, would not avoid the synergism of weightlessness and the other factors. The problem of establishing ground controls for the space experiments is expected to impose large demands for both time and money in any space research program.

Section 7

LABORATORY EQUIPMENT DESCRIPTION SUMMARIES

This section contains a brief description of the major items of laboratory equipment which has been identified as a requirement for each research program described in Section 4 of this report. The information was obtained from immediately available sources. In instances where this type of information was not available, estimates were made. Tables 7-1 through 7-4 list the laboratory equipment required for each research program.

In several cases, an item of equipment with the same name and function may be described in each of the four sections, but generally its performance characteristics differ. It is recommended that a wide range of performance characteristics be considered when the requirements for equipment of this type are defined in a development program so that the resulting design will be broadly applicable.

Such current programs as the Integrated Medical Behavioral Laboratory Measurement System (IMBLMS) and the Automated Primate Research Laboratory (APRL) have been cited in cases where the equipment being developed in those programs is either directly applicable or applicable with modification. Commercial equipment now in laboratory use on Earth has been referenced to clarify the performance requirements. This type of equipment generally will require major modification to make it suitable for use in a manned Earth-orbiting laboratory.

BIOMEDICAL AND HUMAN RESEARCH PROGRAM LABORATORY EQUIPMENT DESCRIPTION Table 7-1 (Page 1 of 4)

Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
1. Aerosol particle analyzer	Count and size airborne particles in size range of 0.3 to 10 µ.	NASA-ERC Cambridge for Apollo	Not applicable	0,25	0.25	Available 1968
2. Arterial pressure recorder	Detects pressures in 0-to 250-mm Hg rangs; 20-mV output to provide digital readout and analog recording.	Spacelabs; Garrett for Gemini	12	8.	;	Under development for IMBLMS.
3. Autoclave	Provides steam pressure, electrically or isotopically, to provide 120°C (30 psia).	Numerous commercial none directly applicable	1,150	15	0.05	Requires engineering development.
4. Automatic cell counter	Coulter-type; detects particles in 1 to 10-µ size range.	Commerical model not applicable for space	10 to 12	-	;	Under development for IMBLMS
5. Ballistocardiograph	Miniature accelerometer; detects in range of 0, 1 to 35 Hz.	Donner - or Kissler- type U.S. Naval Aeromedical Instruction	Not applicable	0.3	0.02	Under development
6. Bone densitometer	Provides isotope source with calibrated aluminum wedges and associated electronics.	Spacelabs; Lockheed	15	52	0.22	Under development for IMBLMS
7. Cardiotachometer	Detects cardiac R wave with amplified signal; digital readout preferred,	Numerous clinical (Sanborn Harvard Instruments)	24	7.0	0.25	Development required
8. Carotid cuff	Moulded negative-pressure cuff for cervical spine; provides 0- to 5- V analog output, representative of -150 to ±150 mm Hg range.	Prototype available	.pel	2	o. o.	Under development for IMBLMS.
o Centrifuge, clinical	Rotates to 2,500 rpm and produces 0-to 20-g range.	Commercial types e.g., Beckman Instruments	100	8.0	0.25	Requires adaptation.
10. Centrifuge, human	Short-radius type; produces 0- to 6-g range acceleration.	Previous development MDC, U. S. Navy	275	155	502	Specific design for final spacecraft configuration.
11. Chemical analyzers	Automated analyzers; makes various chemical determinations. To utilize spectrophotometric techniques; accuracy ±5% maximum possible and desirable.	Technicon and	Not applicable	Not applicable	Not applicable	Some under development for IMBLMS. None directly applicable at present.
12. Electrocardiograph	Frequency response: 0.2 to 100 Hz ±0.25%. Output signal: 5.0 ± 0.2 Vdc; variable gain: 0.5 to 200.	Spacelabs for Gemini	40 mW	0.2	1	Available now.

Table 7-1 (Page 2 of 4)

Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
13. Electroencephalograph	Response: 0 to 50 Hz. Multiplex pulse-width modulated output. For oscilloscope display or analog tape recording.	Spacelabs/Bio Com	50 mW	4.0	:	Available now.
14. Electromyograph	Condition: 0- to 100-mV input signal for analog output to display and recorder. Performance similar to ECG conditioner.	E and M Instruments	50 mW	4.0	; ;	Modified Apollo-type signal conditioner under development.
15. Endoradiosonde	Detects and transmits pressure pH and temperature. Short-range telemeter to on-board rf converter/recorder.	Rockefeller University	Self- contained	;	i	Under development; available date unknown.
16. Ergometer	Provides calibrated load: 25 to 350 W ± 5% with pedal output 50-to 100-rpm range. Hand or foot operation.	USAF-AMD	Self.contained	15	30	Undergoing flight qualification.
17. Gas chromatograph	Analyzes expired air for O ₂ and CO ₂ content to be expressed as pO ₂ , pCO ₂ , 0- to 100-mm Hg ±5%. Digital readout with analog recording.	Melpar for Apollo	a	7.5	0,5	Available by 1969.
18. Gas flowmeter	Measures and indicates flow in 0.04 to 0.3 lb/hr to 0.3 to 3 lb/hr.	Apollo Series 5000	Not applicable	6.0	3	Available now.
 Goggles, plethysmo- graphic 	Sealed hydrostatic pressure unit with hand pump (bulb) pressure source and recorder; 0- to 200-mm Hg range.	Prototype developed and described,	Not applicable	0.5	0.2	Requires development; can be available by 1970.
20. Hemoglobinometer	Estimates Hgb content by light comparater method using color standards and range of 10- to 20-mg ±5%.	Various commercial types available	Not applicable	0.25	0.01	Available now,
21. Humidity recorder	Records H ₂ O vapor in 1 to 1,000 ppm by volume. Accuracy ±5% at full scale.	Commercial-type only	1 to 2	61	}	Development required.
22. Impedance cardiograph	Used for computation of cardiac output to provide average base impedance, change, and rate of change in range of 10 to 40 ±3%.	Spacelabs; for Apollo	70 mW	8.0	;	Available by 1969.
23. Lower-body negative- pressure device	Collapsible pressure vessel; provides 0- to 60-cm H ₂ O negative pressure; continuously variable over range; ventilated.	NASA-MSC for AAP Program: USAF-AMD	1	24	8.3	Available by 1970

Table 7-1 (Page 3 of 4)

	Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
. 72	24. Manometer recording	Pressure range: 0 to 500 mm Hg ±0.1%; analog output for digital readout and tape recording.	Various models available	0.25	1.0	0 5	Development required.
25.	25. Mass Measurement, large	Measures masses in 1- to 100-kg range with ±5% accuracy.	Lockheed	35	•	1.9	Under development.
26.	26. Mass measurement, small	Measures masses in 0- to 1,000-g range with $\pm 5\%$ accuracy.	USAF-AMRL) { !	2.0	0,3	Available by 1969.
27.	27. Microscope, light	Standard clinical light microscope: 10X to 1,000X magnification: photo adapter integral light source.	Various commercial	25	10	1.0	Can be redesigned for light weight.
28.	Muscle dynamometer	Measures muscle force in 0- to 300-lb range ±10% accuracy.	Lockheed	10	4	-	Under development for IMBLMS.
29.	Nebulizer chamber	Delivers aerosol particles in 0.5- to $10-\mu$ range, up to density of $10^5/cu$ ft.	De Vilbiss commercial	Vacuum	2.0	1.0	Requires development and adaptation; available.
30.	30. Nitrogen analyzer	Digests urine/fecal slurry and titrate ammonia salts by Kjeldahl method. Automated with readout of 0 to 500 mg ±5% N2.	None	6.0	m	0.03	Requires full development; no prototype available.
31.	31. Nystagmographic goggles	Monocular; measure two-dimensional visual space orientation. Cine photo-recording with fiber optics; eyeball rotation 0°to 10°±1; horizon tilt 0° to 90°±1°.	Prototype: U.S. Naval Aeromedical Institute.	Not applicable	9.6	;	Prototype developed requires adaptive integra- tion.
32.	Ophthalmoscope	Standard clinical type; miniature.	Welch-Allyn	Self- contained	0.1	{	Available now.
33.	Oscilloscope	Eight-channel display sweep; 2.5 to 100 mm/sec by manual selection.	Various	25	25	1.1	Will require specific design development for final system configuration.
2 .	34. Osmometer	Measures osmotic pressure in aqueous systems; 300- to 3, 000-mosmol range with 5% accuracy.	Hewlett-Packard 300 series	200	40	72	Development required for light flight-rated unit.
35,	Otoscope	Standard clinical type; miniature.	Welch-Allyn	Self- contained	0,1	÷	Available now.
36.	Oximeter, ear	Measures blood O ₂ saturation 50 to 100% range; accuracy ±5%, uses red and IR filtered photocell.	NASA-ARC	и	0.05	;	Under development; availability date unknown.
37.	37. pH meter	Measures hydrogen ion concentration in pH range of 0 to 11, with accuracy of ±0.1% in temperature range of 10° to 50°C.	Beckman; portable, battery- operated	Not applicable	1.5	0.03	Under development for IMBLMS.

Table 7-1 (Page 4 of 4)

Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
38. Phonocardiograph	Crystal pickup microphone sensitive to 5 to 4,000 Hz. Signal amplifiers to provide oscilloscope display and analog recording.	NASA-MSC for Gemini Spacelabs for Apollo	40 mW	0.5	1	Available now.
39. Phoropter	Test visual activity and eye muscle balance; phorias to be tested in range of 0 to 10 prism diopters at 0.25A intervals.	Keystone, B and L, Titmus	52	5.0	0.04	Flight-weight model required; could be available by 1970.
40. Plethysmograph	Capacitance technique; measures limb blood flow in 0- to 5-liter/min. range. 10% accuracy required. Output as analog signal, 0 to 5 V.	None	0.1	0.15°	;	Under development for IBLMS.
41. Pressure cuffs, limb	Pneumatic cuffshand-bulb- operated; provide 0- to 200-mm Hg sustainable pressure with gage readout.	Tycos	Not applicable	0.25	0, 01	Direct application of existing models possible.
42. Scintillation counter	Liquid and well types. Integration desirable to detect specific activity in ranges 10^{-1} to 10^{-4} $\mu c/ml$ with 0.5% reproducibility.	None	10	10	0.2	Under development for IMBLMS.
43. Specific ion electrodes	Measure concentrations of various inorganic ions-specific ranges variable, Desired accuracy, ±0.1%.	Beckman	Not applicable	0.1	;	Available and under additional development for null-gravity systems.
44. Spectrophotometer	Multipurpose; measures optical density in λ range 200 to 600 m μ or selected bands by filtration. Accuracy, 0.1%.	Beckman	10 to 25	ហ	Ħ.	Under development for IMBLMS.
45. Spirometer	Measures standard lung volumes and flow rates. Can measure mechanically with analog signal output for data recording.	Various commercial manufacturers	۲-	12	0.27	Requires development; no commercial model directly applicable,
46. Temperature recorder	Signal conditioner for thermistor output to analog recording.	Spacelabs, Inc. for Gemini/Apollo	Not applicable	0.2	1	Available now.
47. Thermistor	Registers body temperature, Range 35 to 40°C with accuracy of 0.1°C.	Yellow Spring Instruments	Not applicable	0.1	1	Available now.
48. Tilting motion chair	Graybiel Miller type for zero-g operation to calibrated linear and rotary acceleration.	USNAMI for AAP program Lockheed modified	40	25	3.0	Prototype available; modi- fication under development for IMBLMS.
49. Volumeter, whole body	Plastic enclosure for measurement of body volume by gas dilution technique. Desired accuracy, ±2%.	Unknown	Not applicable	<u>ن</u> ن	0.1	Used with other gas analyses apparatus for actual determination.

Table 7-2 (Page 1 of 4)

BEHAVIOR EXPERIMENT LABORATORY EQUIPMENT DESCRIPTIONS

Power Weight Volume (W) (1b) (cu ft) Remarks	5.0 1.0 Recommend noise-cancelling mike, voice-activated with minimum delay and 2-sec follow-up. If head phones used, provide sidetone from mike to indicate adequate volume.	3.0 0.2 All controls should be oberable in the pressurized	pressure garment assembly (PGA); light source must be included.	pressure garment assembly (PGA); light source must be included. 1.5 0.2 Include strobe-type flash and variety of film: color, black and white, high contrast, and high speed.	1.5 0.2	1.5 0.2 2.0 0.15 1.0 0.02	1.5 0.2 2.0 0.15 1.0 0.02 4.0 1.0
Pow Available Model (W	Ampex	Bolex	Hasselblad	NASA-MSFC	MDAG COMPARE	Biomedical apparatus (IMBLMS)	MDAC COMPARE 10 IMBLMS
Performance Characteristics	Records voice over range of 200 to 9,000 Hz, 10 to 110 dB. Mission time automatically put on tape when activated by voice or pushto-talk switch.	I to 64 fps; single-frame exposure; range finder; 16 mm or larger; zoom, wide-angle and telephoto lenses, cartridge load; automatic exposure select; pop-up external viewfinder; film remaining indicator.	35-mm or larger single lens reflex; I automatic exposure select; wideangle, telephoto and close-up lenses; one-hand operation.	Data-transmission capability and light source. Should require only one hand for operation. Standard sweep acceptable.	0 to 24 hr; 0.01-sec accuracy; sweep second hand; time out; portable; capable of hand or electronic pulse control.	Measures respiration rate and volume. Must be of a mask or mouthpiece type and register from 0 to 6.6 liter and 0 to 80 liter min. pO2 0 to 2.5 psi: CO2: 0 to 1.5 psi: temperature sensitivity core: 900 to 1010F: and skin: 50° to 1150F. EKG: 300 to 5,000 µV: bandwidth 0.5 to 50 Hz.	Hand strength in knob twisting up to 45 in. /lb; hand strength in full grip twisting up to 120 in. /lb; hand strength in squeeze up to 175 lb; arm strength at various elbow angles up to 180 lb; backstrength; (vertical pull) up to 650 lb; leg strength (vertical pull) up to 1,950 lb.
Name/Function	1. Audio tape recorder	2. Movie camera	3. Still camera	4. Video camera	5. Timer	6. Metabolic monitor	7. Torque-force apparatus

Table 7-2 (Page 2 of 4)

Name/Function	Performance Characteriatics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
8. Standard dexterity apparatus	Provides capability for determining aiming and manipulation factors separately.	(SRT) MDAC COMPARE	1	0.5	0, 01	These tests typically employ many loose items not acceptable in zero gravity; new norms must be developed for a modified task.
9. Ergometer	Must exercise major flexor and extensor muscle groups by repetition rather than through use of unusually large forces. Whole body push and pull appears appropriate with resistance forces adjustable from 25 to 150 lb.	(SRT)	Not applicable	20.0	2.0	Frictional devices often produce particles from abrasion which may "float" and be inadvertently inhaled or ingested. Some form of "pacer" may be used to time and count repetitions. Heart rate display should be visible to subject or monitor.
10. Accelerometers and recorders	Measurement of accelerations imparied to crewman by self or to transported masses. Sensitivity should range from 0.5 to 45.0 pfs.		32	0	8.0	Available model of required high sensitivity are high-cost items.
11. Depth perceptometer	Classical Howard-Dolman apparatus may be scaled down from 10 to 2 ft long to be used for measurements of angle of disparity using 0, 001-in. increments. Should allow measurement of 1 to 150 sec of visula arc.	MDAC Visual- Auditory Tester	2	3.0	9.0	Conceptual design complete.
12. CRT and other displays.	CRT may be used for complex tracking, vigilance, and pattern recognition. Minimum of 4 to 5 in, diam: capable of dual input for target and cursor: 6 x 6 grid projections; probability monitoring needle gages; vigilance lights; alpha-numeric display.	MDAC COMPARE	3.8	40.0%	2.0	CRT should be computer-driven and scored. Should included 6 x 6 grid reference and test patterns (about 1,000), capable of higher-order pursuit tracking random inputs.
 Display-associated controls 	Items related to Item 10 above; appropriate joystick, pushbuttons for vigilance and probability monitoring, numerical entry, pattern identification.	MDAC COMPARE	ហ	40.0#	2. 0*	Some controls may also serve as displays by transillumination.
14. Analog recorder	Must be capable of time-shared recordings of times, stimuli, responses, correct-error indication. Telemetry capability desirable. (Related to Item 10 above.)	MDAC COMPARE	30	40.0%	5.0*	Recorder flexibility must increase as does the randomness of the stimuli presented to the crewman. Primary difficulty will be tags necessary to identify stimuli and responses from several concurrent tasks.

Table 7-2 (Page 3 of 4)

Remarks	If vernier acuity not feasible, minimum separable distance next best. Equipment indicated may also be employed to investigate near-static depth perception.	If technology permits, more detailed observations could include eve move- ments, convergence, shape, and size constancy.	Should apply method of limits or method of adjustment with reference to a control source.	If integrated with CRT or command control, weight and volume do not apply.		Tones must be presented by a head set. Requires programming of random stimuli for psychophysical measurement to prevent sequence learning.
Volume (cu ft)	4.0		4.	0.4	0.1	9.0
Weight (1b)	2	ę. 0	٥ ٥	2.0	0.25	9.0
Power (W)	2	!	'n	1	.i. i.	25
Available Model	Bausch and Lomb- Orthorater, Visual- Auditory Tester	(SRT)	MDAC Visual- Auditory Tester	(SRT)	Ishihara	Hewlett-Packard, MDAC Visual-Auditory Tester
Performance Characteristics	Measure visual acuity down to 0.2 arc-sec, preferably by vernier acuity technique.	This equipment may be combined with Items 15 and 17. Used to examine: A. Figural after-effects. B. Figural masking. C. Dark adaptation; 0.00001 m1. minimum. D. Phi phenomenon: 100 to 200 msec; 2 to 12 in. E. Stereoscopic effects. F. Autokinesis. Two lights; separately controllable in frequency and intensity from 0 to 60 Hz and from 0 to 100 ft-c, with known chromaticity.	Apparatus must control (possibly vary): A. Intensity of the positive phase. B. Difference between the two phases. C. Time proportion of the two phases. D. Area of flickering field. E. Part of the retina stimulated. CFF may, depending on manipulation of above variables, range from 5 to 55 cps.	Illuminated bar rotable in field of vision on single plane in both directions from true vertical.	Pseudoisochromatic alphanumeric printed plates	Absolute threshold (0.0005 dynes/ cm ² at 2,000 cps) differs with frequency from -70 to +50 dB: just-noticeable pitch differences from 2 to 30 cps, depending on frequency and intensity. The differential limen for loudness also varies with frequency and
Name/Function	15. Orthorater	16. Light source	17. Critical flicker fusion (CFF) source	18. Veridicality test	19. Color plates	20. Pure tone source

Table 7-2 (Page 4 of 4)

Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
	intensitymost efficiency at 2,000 cps. Auditory test apparatus must consider these sensory capabilities. Requirements: 2 tone generators 0.0005 to 100 dynes/cm ² 25 to 15,000 cps					
21. Noise and vibration source	Provides vibration to evaluate somesthetic and cutaneous sensitivity. For cutaneous sensitivity: 200 to 300 Hz with 0.001-mm amplitude is generally considered to be the minimum detectable vibration, although 12,000 Hz can be detectable at higher amplitudes. For somesthetic sensitivity, sinusoidal rather than complex vibration recommended. Sensitivity and tolerance varies greatly in different body members and areas. Tolerance is least at 4 to 8 Hz.	(SRT)	20	0.0	6 0	Precise definition is required of body areas; frequency and amplitude characteristics to be investigated.
	Noise tolerance depends both on intensity and frequency characteristics, but with white noise, roughly corresponds to the Speech Interference Level (SIL), e.g., 80 dB at 1 ft. 65 dB at 10 ft.	Hewlett-Packard	1	1	1	
22. EMG sensors and recorders	Sensors: 100 to 10,000 μV: 0.5 to 5,000 Hz. Bandwidth: 2 to 15 msec.	Biomedical apparatus (IMBLMS)	7	3.0	9.0	
23. GSR and EEG sensors and recorder	GSR: 5, 000 to 100, 000Ω or more-variable sensitivity appropriate	Biomedical apparatus (MBLMS)	ú'n	5.0	9.0	
	EEG: 10 to 300 μV : 0.2 to 100.0 H_Z bandwidths					EEG sensors must be attached directly to the scalp and highly amplified. Recorders for EMG, GSR, EKG, if highly sophisticated, could be combined.

Table 7-3 (page 1 of 12) BIOSCIENCE LABORATORY EQUIPMENT DESCRIPTION

Automatic cell counter 0 to 7.5 x 10 ⁶ celts per cu mm Fisher Autocytometer 20 10 0.4 Autociave 30 to 40 psis, 100° to 135°C 20 25 2 Autociave 20 25 2 Automatic plate 20 25 2 Bone destionmeter Operating range: 20 to 50 kV Phillips Oralix 1,000 100 2 Bone destionmeter Operating range: 0.02 to 0.5 sec Super 50 Denial X-ray Unit 1,000 100 2 Movie Camera 55 mm film; frame rates between 0 0 0 0 Roll film camera 1 hr1 and 20 sec-1 0 2.75 0.037 Roll film camera 125 proy tones. Frame rates Frame rates Frame rates 0 2.75 0.037 Wideo camera 125 proy tones. Frame rates Frame rates Frame rates 5 0.1 Contritinge, organism 0.1 x g to 1	1	Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
Automatic plate and desironneter Coperating range: 20 to 50 kV Bone desironneter Coperating range: 20 to 50 kV Coperating range: 20 to 50 sec I hrl and 20 sec-1 Coperating range: 20 to 50 sec I hrl and 20 sec-1 Coperating range: 20 to 50 kV Coperating range: 20 to 50 rule range between Video camera Standard 3-1/4 in. x 4-1/4 in. Video camera Standard 3-1/4 in. x 4-1/4 in. Coperating range: 20 to 50 rule range rang	14	Automatic cell counter	× 106	Fisher Autocytometer (Blood Cells)	20	10	4.0	
Automatic plate Scanning counter Bone desitometer Contribuge, refrigerated Colorimeters Accommodates (20 to 50 kV philips Oralix property per per pen passer of the p	~i			:	1, 000	25	,	Possibly use small unit with less power (1, 000 w).
Docimeters Coperating range: 20 to 50 kV Phillips Oralix 1,000 100 2	m,	Automatic plate scanning counter	i	:	20	20	7	With automatic sample changer and paper printout.
Movie Camera 35 mm film; frame rates between 0 0.3 Plate film camera 1 hr1 and 20 sec-1 0 0.37 Polaroid Camera Standard 3-1/4 in. x 4-1/4 in. 0 2.75 0.037 Roll film camera 35 mm 0 5 0.1 Video camera 128 ery fonce; Frame rates between 4 and 128 ery fonce; Frame rates between 1 hr.1 and 30 sec-1. With the recorder and display. 0 5 0.1 Centrifuge, organism 0.1 x g to 1 x g. 50 25 1.5 Centrifuge, refrigerated Accommodates sight 50-ml tubes Lourdes Model A 2.000 200 18 Speed 40,000 xg. Temperature range 1.00 300 18 1.5 Clinostat Spectronic 20 10 0.4 Colorimeter Range: 340 to 950 mµ Spectronic 20 10 0.4 Resolution: 20-mµ band pass: Pilm badges:	4 :	Bone desitometer	Operating range: 20 to 50 kV Exposures: 0.02 to 0.5 sec	Phillips Oralix Super 50 Dental X-ray Unit	1,000	100	2	Can be used for human limbs (not skull or torso), besides nonhuman vertebrates. Includes lead shielding. Uses
Movie Camera 35 mm film; frame rates between 1 hr1 and 20 sec-1								slow film to avoid background fog.
Plate film camera Standard 3-1/4 in. x 4-1/4 in.	rç.	Movie Camera	35 mm film; frame rates between I hr1 and 20 sec-1	;	.0	0	0.3	Slow emulsion film to avoid background fogging. Possible
Standard 3-1/4 in. x 4-1/4 in. Contrad Design-NAA Contrad Design C	٠.	Plate film camera	1	,1 1 1	i	! ! !	;	Polaroid. Conceivably defete
Roll film camera 35 mm 0 5 0.1 Video camera 250,000 elements, between 4 and 128 grey tones. Frame rates between 1 hr-l and 30 sec-1. With recorder and display. 160 60 2 Centrifuge, organism 0.1 x g to 1 x g. 50 25 1.5 Centrifuge, refrigerated Speeds to 18,000 rpm; forces to high speed Accommodates eight 50-ml tubes fruge Lourdes Model A 2,000 200 18 Clinostat Speeds to 18,000 rpm; forces to 400°C. Beta-Fuge 2,000 200 18 Clinostat Colorimeter Range: 340 to 950 mp Bausch and Lomb 20 0 0 4 Dosimeters Resolution: 20-mg band pass. Spectronic 20 Spectronic 20 0 0 0 4 Dosimeters point changer. Film badges 13-mr to 405 2500 series 0	۲.	Polaroid Camera	Standard 3-1/4 in. x 4-1/4 in.	1	0	2.75	0.037	ent resolutions and scan rates)
Video camera 250,000 elements, between 4 and 128 eye chorses. Frame rates between 1 hr-1 and 30 sec-1. With recorder and display. 160 60 2 Centrifuge, organism 0.1 x g to 1 x g. 50 25 1.5 Centrifuge, refrigerated high speed Accommodates eight 50-ml tubes of 18,000 rpm; forces to 40,000 xg. Temperature range from -20°C to +40°C. Beta-Fuge of 40,000 xg. Temperature range from -20°C to +40°C. 18 1.2 Clinostat Spectronic 20 Spectronic 20 10 0.4 Colorimeters Resolution: 20-mµ band pass. Spectronic 20 10 0.4 Dosimeters Pen dosimeters (200-mr full scale) in Chamber and film badges tail badges (3-mr full scale). In Chamber and film badges 10 0.2	80	Roll film camera	35 mm	l	0	ιn	0.1	All cameras with close-up optics compatible with microscopes.
Centrifuge, organism 0.1 x g to 1 x g 50 0.15 1.5 Centrifuge, refrigerated Speeds to 18,000 rpm; forces to high speed 40,000 xg. Temperature range from -20°C to +40°C. Clinostat	6	Video camera	250,000 elements, between 4 and 128 grey tones. Frame rates between 1 hr-1 and 30 sec-1. With recorder and display.	į	160	09	74	
refrigerated Accommodates eight 50-ml tubes Lourdes Model A 2,000 200 18 40,000 xg. Temperature range from -20°C to +40°C. Enaberature range from -20°C to +40°C. 1.2 Conrad Design-NAA 25 8 1.2 Range: 340 to 950 mμ Resolution: 20-mμ band pass. Spectronic 20 10 0.4 Pen dosimeters (200-mr full scale) Nuclear Chicago 401 0 5 0.2 with changer. Film badges (3-mr full scale). Ion Chamber and film badges to 405 2500 series 5 0.2 radiation meter. radiation meter. 100 Chamber and film badges 100 Chamber and film badges 100 Chamber and film badges	10.			1	20	25	1.5	For subvertebrates through mice, counterrotating design.
Colorimeter Range: 340 to 950 mp Bausch and Lomb 20 10 0.4 Resolution: 20-mp band pass. Spectronic 20 Dosimeters Pen dosimeters (200-mr full scale) Nuclear Chicago 401 0 5 0.2 with changer. Film badges (3-mr to 405 2500 series to 600-mr full scale). Ion Chamber and film badges	:	Centrifuge, refrigerated high speed	Accommodates eight 50-ml tubes Speeds to 18, 000 rpm; forces to 40, 000 xg. Temperature range from -20°C to +40°C.	Lourdes Model A Beta-Fuge	2, 000	200	18	May need counterrotating design to null angular momentum. Also consider microcentrifuge 0.4-ml tubes, 15,000 rpm, 100 W, 10 lb, 0.2 cu ft, Beckman Model 152 Microfuge).
Colorimeter Range: 340 to 950 mµ Bausch and Lomb 20 10 0.4 Resolution: 20-mµ band pass. Spectronic 20 Dosimeters Pen dosimeters (200-mr full scale) Nuclear Chicago 401 0 5 0.2 with changer. Film badges (3-mr to 405 2500 series to 600-mr full scale). Ion Chamber and film badges radiation meter.	2	Clinostat	!	Conrad Design-NAA	25	œ	1.2	Centrifuge included.
Dosimeters Pen dosimeters (200-mr full scale) Nuclear Chicago 401 0 5 0.2 with changer. Film badges (3-mr to 405 2500 series to 600-mr full scale). Ion Chamber and film badges radiation meter.	ю́.	Colorimeter	Range: 340 to 950 mµ Resolution: 20-mµ band pass.	Bausch and Lomb Spectronic 20	20	10	4.0	Dial readout (no recorder). For body fluid clinical analysis
	4.		Pen dosimeters (200-mr full scale) with changer. Film badges (3-mr to 600-mr full scale). Ion Chamber radiation meter.		0	: LC \	0.2	Gamma radiation detectors.

Table 7-3 (page 2 of 12)

	Available Model	(m)	(1p)	(cu ft)	Remarks
150°C to 190°C range	` ;	20	ıΩ	0.3	For small instruments.
Ambient to 125°C range	;	009	40	73	Possibly combined with incubator as single unit.
Itage	Beckman Electroscan 30	10	20	ed .	For cyclic voltammetry, chronoamperometry, con-
ranges between ± 0.5 v and ± 5.0 V, with scan rates between 0.5 mV/sec and 500 mV/sec.					anodic stripping voltanme- try, etc. (polarographic analyses). Includes pH and Redox measurements.
Power supply provides either con- B	Beckman Model R	20	20	ä	Needs a recorder.
ienguis selected with interference filters (typically 500 mm).					
;	;	0	ທ	0.2	:
;	1 1 7	0	0.5	0.01	:
-20°C	:	25	S	-	1 4 1
2, 000-psia air, hydrogen, helium	;	0	150	ĸ	Air and hydrogen for flame photometer and hydrogen, flame detector of gas chromatograph. Helium for gas chromatograph. Safety hazard.
Thermal control range: 40°C to B	Beckman GC-2A	100	10	9.0	Hydrogen flame detector needs
gen flame detector has typical sensitivity of 50 parts per billion (full scale) for organic compounds (10 ⁴ times as sensitive as thermal conductivity detector.)					Needs helium carrier gas supply. Needs recorder.

Table 7-3 (Page 3 of 12)

	Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
4.	Histology kit/slide cabinet	4.9.1		0	ń	1	
25.	25. Incubator	5°C to 60°C	1	09	20	0.7	For both experiments and paraffin sample embedding.
1 2	Z6. Isotope tracer equipment						
	A. Isotope solutions and shielding	C-14 labeled acetate, amino acids, etc.	1 3 3	0	10	0.5	:
	B. Liquid scintillation counter	:	Nuclear Chicago 6850	200	200	ĸ	Automatic sample changer and paper printout. With shielding (more may be needed).
	C. Radioautography apparatus	Slow film to avoid background fogging.	1	0	7	0.2	
27.	Liquid nitrogen Dewar flask,	-100°C to -190°C depending on design.	1	0	100	69	Boiloff problem with cryogenic storage.
28.	Lyophilizer	Isopentane cooled with liquid nitrogen to -160°C.	i	0	10	8.0	Uses space vacuum. Alternate cooling electrically to -60°C for 200 W, 20 lb, 1 cu ft.
29.	Mass-measurement device (macro)	Measures 0, 1 kg to 30 kg.	1	0	ស	5.2	•
30.	Mass-measurement device (micro)	Measures 0.1 g to 100 g.	1	0	-	0.2	•
31.	31. Mass spectrometer	Detection to mass number 1,000 and 1 part per million. Unit resolution at mass 500.	EAI Quad 300 (quadruple type)	200	40	0	Uses space vacuum to eliminate pumps. Works with gas chromatograph. Needs oscillograph recorder and computer data processing.
32.	32. Micromanipulator	Ellis piezoelectric-type	;	10	ĸ	0.5	;
33.	33. Microscope, compound A. Trinocular tube B. Accessories for bright- field, dark-field, polarized light, phase contrast, interference contrast, and fluores- cence microscopy.	10X eyepiece, 25X, 10X, 40X and 100X objectives.	Zeiss Binocular Microscope, Model RA	2 2	0	9.5	1
34.	34. Microscope, electron	50-kV 100-Å resolution Magnification: 2000X to 4000X	Japan Electron Optics Laboratory (JEOL) JEM-50	200	150	·ίn	Uses space vacuum to eliminate pumps.

Table 7-3 (page 4 of 12

			Power	Weight	Volume	O C
Name/Function	Performance Characteristics	Available Model	(w)	(a)	(ca it)	Actitativo
35. Microscope, stereoscopic dissecting	2.5X to 100X magnification at 80-mm working distance. 5X to 200X at 28-mm distance.	Zeiss Stereo Microscope III (Zoom)	30	30	-	With trinocular tube and illuminator.
36. Microtome	Section thickness: 1 to 50. Ultra-thin sectioning adapter achieves 0.05-thickness.	AO Precision Rotary Microtome No. 821	0	10	0.3	Freezing attachment may need modification from CO ₂ . Electron microscope needs ultra-thin sections. Most applications need only 5- to 50-µ thickness.
;		1	20	10	ĸ	Total.
		: !	0	7	0.3	Preservative and reagents.
A. Chemicais B. Fluid-handling	: :	;	0	ស	0.5	Syringes, test tubes, stoppers, etc.
equipment C. Timers	3	ţ	0	1	0.1	:
	*	;	20		0, 1	;
E. Dissection/surgery/	}	;	0	ທ		: : : :
F. Accelerometer	10^{-7} g to 10^{-1} g at 10^4 cps or less. 10^{-1} g to 10^3 g at 10^4 cps or less	11	10	. v s	0,1	Continuous operation.
G. Rulers, protractors, etc.	1 1 2	;	0	8	0.5	1
H. Toxicity detectors	***	1,77	20	20	0.2	Continuous operation.
	1 1 1	;	0	20	0.3	Notebooks, recorder charts, etc.
38. Mixer/shaker	:	.† .†	10	ហ	0.3	}
39. Polarographic CO ₂ and O ₂ sensors	Partial pressure CO2: 1 to 20 mm Hg.	Beckman	0	0.3	0.1	Need amplifiers and dial readouts; probably part of electro-analytical apparatus.
	Partial pressure O_2 : 0 to 1,000 mm Hg.	140732 (CO ₂ sensor) 7841IV (O ₂ sensor)		1 1	11	
40. Radiation shielding	Lead	;	0	200	0.3	•
41. Radiation source	85-strontium gamma source with 180° cone of radiation. Dose levels to 2,000 R/day.	Biosatellite II (hardware)	0	ហ	0.2	Includes shielding and automatic open/close mechanism. Half-lite (65 days) may be marginal for long missions.

Table 7-3 (page 5 of 12)

Remarks	Serve as output units for the spectrophotometers, gas chromatograph, electrophoreses densitometer, electro-analytical apparatus, and mass spectrometer.	ļ	•	With shielded chamber for radioisotope solutions.	Has a heated flow-through cell for use with gas chromatograph. Needs a recorder (oscillographic when in fast-scan mode with gas chromatograph).	Does not include recorder. Flame photometer needs air and hydrogen supplies (detects alkali metals: typically sodium, potas- sium, and calcium).	With tanks, cleaner fluid, and other accessories	Reclaims water with cold trap. Uses space vacuum. Probably combined with lyophilizer.	Includes otoscope/ophthal- moscope, reflex hammer hemoglobinometer, tour- niquet etc.
Volume (cu ft)	•	1	0.3	m	N	-		7	
Weight (1b)	20	20	5	20	20	ın	20	10	ហ
Power (W)	40	40	20	50	250	01	30	0	0
Available Model	Texas Instruments Oscillo/riter	Beckman 100506	•	1	Beckman IR 20	Beckman DK-2A	1	:	;
Performance Characteristics	•	10-in. scale with disk integrator	1 1	7₀₹	Range: 2.5 to 3.5u. Resolution better than 0.1% wavelength. Scanning time 5 sec to 50 hr. Sample temperature to 250°C with heated cell. Double beam Nichrome source. Thermocouple detector.	Range: 185 to 700 in Resolution: better than 0.1% wavelength. Scanning time: 3 min. to 5 hr. Dougle beam, ratio recording. Tungsten and hydrogen lamp sources. Lead sulfide cell and photomultiplier tube detectors.	Operates at 20 kHz (and/or others) with piezoelectric transducers.	;	;
Name/Function	42. Recorder, high-speed oscillographic	43. Recorder, linear potentiometric	44. Recorder, voice	45. Refrigerator	46. Spectrophotometer, IR	47. Spectrophotometer, UV-visible A. Flame photometry attachment B. Reflectance photo-	48. Ultrasonic cleaner	49. Vacuum desiccator	50. Veterinary medical kit

Table 7-3 (page 6 of 12)

		Distraction among	()	(ar)	(ca rr)	Remarks
Primate calorimetry module (sealed adiabatic chamber)	Contains one instrumented, 20-lb male, adult, pig-tailed macaque monkey (Macaca nemistrina) and EC/LS and biologic fluid subsystems for a 90-day experiment.	Laboratory envelope, structure (McDonnell) for UC Berkeley APRL	70	06	230	Two such modules required. These sealed calorimetric primate capsules subserve the primate metabolism and
A. Oxygen supply	140 f/day for 90 days	; ;	1	40) 1	primate hemodynamics experiments. CO2
B. Nutrient supply	900 g/day for 90 days.	14 .1	}	180	i i	absorbed by LiOH (32 1b).
C. Nutrient dispenser	1 1	;	1	1	;	Food and water mixed in a slurry.
D. Urine-collection subsystem	1	;	1	TBD	;	
(1) Urethral catheters	1	1 1 1	!	!	1	
(2) Urine pump	1 ! !	1 1	;		;	
(3) pH meter	1		1	.1		
(4) Pressure sensor	1	!!!	1	1 1	1	
(5) Relief over- pressure	;) 	! !	; ; }) 1 7	
(6) Flow meter	:	;	1 1	1	1	
(7) Automated urine analyzer	10% of total urine volume analyzed JPL unit in flight	;	į	11	TBD	Analyzes calcuim creatine.
(8) Urine-storage freezer		}	1 1	:	:	
(9) Temperature sensor	:)) 1	! !	1	4	
E. Feces-collection subsystem	• ***	;	;	1	TBD	
(1) Flatus-gas remover		;	;	i	;	
(2) Solid-feces storage freezer	;	1 1 1	}	1 .1	;	
F. Automated hemodynamic subsystem	•	;	\$ \$ 1	1	TBD	
(1) Vascular catheters	!	: :	}	ŧ 1	3 3 1	
(2) Withdrawal and infusion syringes	t : t	ţ į		ł	ł	
G. Pressure transducer		1 7 7	!	:	1	
H. Optical densitometer	r Measure density at 805µ	1 6 7	!	1	į	
I. Optical densitometer	r Measure density at 660µ	;	1 1	4	1	
J. Refractometer	Measure refraction at 805µ		!	1	į	

Table 7-3 (page 7 of 12)

	Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
	K. Pump, mechanical drive			.\$ \$ 3	-	1	
52	Primate geodesic module	Contains one instrumented chimpanzee (25 to 40 lb) and EC/LS, waste-management, and neurophysiologic test equipment for a 60- to 120-day experiment.	Laboratory envelope structure (Adey-UCLA, Mechan-USC)	700	20	180	Two such geodesic. spheres required: one chimp restrained, unrestrained. The two modules subserve the primate behavior and the primate bioghythms experiments (Total PCM: 10 MHz.)
	A. Electroluminescent house lights B. Oxygen C. Nutrient supply D. Nutrient dispenser E. Waste-management subsystem F. Ergometric joystick manipulanda G. Speaker for tone generator	;	!	100-	0	1	6.5-sq ft total surface area. LiOH (32 lb) to absorb CO ₂ . Food and water mixed in a slurry. Air-flow and wiperblade action.
	H. Stereo-binocular microscope with rheostat-controlled incident illuminator	Low to high stereomagnification.	Zeiss Instruments	!	1	1	Can visualize microcirculation of blood in scleral conjunctival vessels of eye. (This apparatus may better subserve the macaque monkey hemodynamic experiment than the chimpanzee biorrhythms experiment.)
5.	Mouse reproduction module (with central nesting area) A. Oxyten supply B. Nutrient supply (1) Food pellets (2) Water C. Nutrient dispensers D. Water dispenser	!	1	5-8	180	10	Pressurized to 14. 7 psi; 20% O ₂ /80% N ₂ , CO ₂ absorbed by LiOH, Air flow rate 6 cu in. /min. Forms part of the 'hat'' of chamber
	E. Video wide-angle camera						

Table 7-3 (page 8 of 12)

Remarks	Sealed adiabatic chamber.	Used for circadian rhythm study. Ideal duration of biorhythm experiments would be 90 days or more. Connections will be required for electric power and telemetry system.	Experiment package will require space-radiator service from the carrier	Four fish experiments, each requiring aquarium facilities.	(Note: Some of the invertebrate experiments, e.g., Planaria (flatworm) studies also require aquarium facilities.)		Three Daphnia experiment categories exist. All three cylinders of a flight package should be exposed to the same vibration and gravity forces. (Note: Each container is a cylinder with 4-3/4-in. x 1-in, dimensions.)
Volume (cu ft)	{	2		0.5		:	35.5
Weight (1b)	;	70		25		!	99 0
Power (W)	1 1	25		10		10	None
Available Model		Northrup (Dr. Lindberg)		ТВD		TBD	Present state issue of standard 60-ml laboratory vials with screw caps (each vial with cap). (Note: NASA Ames Research Center is developing similar containers.)
Performance Characteristics	Precise heat-production measurements. Ergometrically programmed with timers.	Six animal tubes enclosed in a container with antennas and receivers for each tube. ECS provides atmospheric composition and experiment temperature control for up to 30 days in space. Earth atmosphere of N ₂ and O ₂ provided, with H ₂ O vapor and CO ₂ concentration controlled.		Modified for air volume control in orbit. Must be capable of sustaining small aquatic vertebrate fishes (e. o. o unmies) in a weight-	less environment. (Note: A larger aquarium may be required for the flounder (genetics) experiment.)	Modified for orbital conditions.	Three containers of 60-ml volume each comprise the flight package for one experiment. The cylinders may be placed in line or clustered. Thermal constraints for each cylinder: A. 18°C minimum and 22°C maximum for operating temperature limits; B. 5°C maximum and 25°C maximum and 25°C maximum for storage temperature limits;
Name / Function	54. Mouse calorimetric module Rodent exerciser	unit with ECS, datahandling, and -storage. Battery for interior power for ECS operation during launch, ascent, and OWS setup integrated into common package for unitary mounting and operation.		56. Fish aquarium		Aerator	containers. (Note: Each 60-ml sealed container holds four pregnant Daphnia pulex and their offspring (7 to 30 per container: oxygen-saturated medium No. 39; also Chlamydomonas (algal food).

Table 7-3 (page 9 of 12)

Remarks			See NASA grant NaG-519.	For cockroach behavior, biorhythm, amoeba parasite (or termite embiosis)	ant-behavior experi- ments. Drosophila (vinegar gnat) facility should separate from the general insect chamber.	Special insect chamber for five of the six Drosophila experiments.	The Drosophila circadian rhythm experiment	requires a specialized chamber for the pupal stage of the vinegar gnat,	The facility described, although designed named or managed	studies, can also sub- serve the morphogeo- sensitivity, metabolism, and behavioral experi- ments with modifications for observations of adult specimens.	
Volume (cu ft)			0. 25	;		0.01		1 1	į		;
Weight (1b)			ນາ	:		4	;	; ;	;		
Power (W)			10			None	1 -1 -2	; ;	4		t 1
Available Model			Prototype modified for small aquatic chambers by Derek Engstrom, U of Wash.	1		i	\$ 5 7	: :	1		!
Performance Characteristics	Medium No. 39 contains precipitate CaCO3 concentration of algae 1 x 10 ⁶ cells/ml.	(Note: Oxygen and food supply become the major limiting factors during sealed experiments. The animals will tolerate pO ₂ within the range of 740 to 74 mm Hg.)	pO2, pCO2 and pH must be monitored on controls. Automatic measuring and recording features have been built but require testing.	Growth and reproduction maintained. Visual and photo- graphic observation.	Day-night cycle reversible.	Maintain temperature range of: 15° C to 26° C	Atmosphere: Standard temperature and pressure	Humidity: 40 to 80% Temperature profile required.	;	individual respiratory require- ments for eggs, larvae, and papaean average of 0.004 cu ml/hr/individual.	The grouping in individuals per cell will be 100 eggs, 100 first instar larvae, and 100 pupae.
Name/Function			58. Automatic physiological gas analyzer	59. insect chamber (general)	A. Illumination source with timer.	B. Drosophila (vinegar gnat) facility (excluding circadian rhythm-	vinegar gnat pupal eclosion experiment)		Approximately 96 celle (chambers)	Air to be supplied to the cells by the capsule gas- management system.	Each cell will contain approximately 400 individuals, for a total of 7, 200 organisms.

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Remarks	The circadian rhythm package, as designed, does not require return to Earth.	Extraneous cyclis events should be minimized.	The package can be sectioned into two halves, one being shielded from radiation.	See insect chamber, general.	The amoeba culture containers may also subserve the dino-flagellate and possibly the neurospora (fungus) environmental culture requirements, with some modifications. The human-cell culture experiments have a special package environment.	Three frog-egg flight packages required.	36 modules required.	108 frog-egg units required.	(Note: Unanticipated lengthened hold time	because low temperature will not maintain accept-	able rates of fertilization indefinitely. If	mission longer than 4-1/2 days, all the	embryos can be fixed and returned up to 2 weeks after launch,)
Volume (cu ft)	2.8		o. s	:		1.6			į	!	;	1 	1
Weight (1b)	32		30	f 1 5	۴	40			}	}	! ! }	<u> </u>	}
Power (W)	21		None	;	255	15 (avg)			į	1 † †	;	1 1 1	1
Available Model	:		1	i !	1	General Electric is constructing breadboard.			;	:	š 1	1 3	1
Performance Characteristics	Package consists of four experiment chambers held at 5°C constant darkness until initiation of experiment in space. Two of	the chambers will be brought slowly to $10^{\circ}\mathrm{C}$ and two slowly to $20^{\circ}\mathrm{C}$. Then light cycling will be imposed, and rate of eclosion (emergence) of papae measured.	Temperature: 65° to 80°F. Oxygen to be dissolved in the water. Food: lyophilized sterile liver homogenates suspended in the water medium.	See insect chamber, general.	‡	Each package contains 12 modules	Each module contains three units.	Each unit contains frog eggs and sperm.	20 cc, 100% relative humidity 20 frog eggs (initial)	20 cc capacity	•	(spring water)	:
Name/Function	60. Drosophila (vinegar gnat) biorhythm module		61. Flatwork (Planaria) container	62. Ant growth chambers	63. Amoeba culture container, chemostat, and pump	64. Frog-egg flight package	A. Frog-egg module	B. Frog-egg unit	(1) One egg chamber	(2) One sperm- solution chamber	(3) Two (or more) wash chambers	(4) One media chamber	(5) One fixative chamber

Table 7-3 (page 11 of 12)

Name/Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
(6) One waste chamber	1	1	:	-	\$ 9 1	
Note: Flights longer than 3.25 days require: 12 to 15 sq ft bench space 1 refrigerator 1 microphotographic apparatus 1 sink 1 transport case	Note: Each experiment package to maintain eggs and sperm at 50 to 100°C during hold time, and at 18°C during orbit; temperature readout by thermistor at modular level (four readouts) continuous to ground. Readout range: 5° to 25°C.					
65. Bacteria culture chemostat with transfer device and pump	:	;	25	ĸ	is 0	
66. Bacteriophage flight package	:	į	25	31.5	0.3	This bacteriophage package is a modification of the frog-egg experiment flight package.
 67. Human cell experiment flight package consisting of: A. Microscope-camera system 	;	Woodlawn Wanderer Nine-A, Texas Instruments, Inc.	2, 5, 5, 5, 5, 5, 5, 5, 5, 5, 5, 5, 5, 5,	17.6	0. 45	
(1) Carnera No. 1	40X magnification optics	;	4	;	;	12,000 pictures at five pictures/min.
(2) Camera No. 2	20X magnification optics					12,000 pictures at five pictures/min.
(3) Microscope	Phase contrast optics with adjustable focus.					
B. Biopack system, consisting of four identical biopack subsystems	Each biopack subsystem consists of two epoxy disks, one a "media" and one a "specimen" disk.	;	:	;	;	
C. Temperature- control system (canister is sealed metal enclosure)	Maintains all specimen slides at 98. 6°F (-4°F - 1°F); canister maintains all experiment equipment in a sealed 14. 7-psia air environment.	;	1	;	1	

Table 7-3 (page 12 of 12)

Name/Function	Performance Characteristic	Available Model	Power (W)	Weight (Ib)	Volume (cu ft)	Remarks
D. Human-cell culture chamber storm cellar	Offers radiation protection (37, 5°C)		20	20	0.1	
E. Refrigerated storm cellar	(-40°C)	1	50	42	0. 13	
68. Plant growth chamber with temperature and humidity controls	Must have dark-chamber capabilities. Radiation source and dosimeters may be required within the chamber.	i	50	100	2, 25	
69. Plant growth chamber (radiation-shielded)	Same requirements as unshielded chamber.	-	50	150	2, 25	
70. Seedling culture tank	Lucite; 36.5 x 3.5 x 4.3 in.	1	}	‡ 1	!	
71. Seed culture tank	Lucite; 36.5 x 2.5 x 2.5 in.	1	;	3	:	
A. Nutrient pump		1	15	2	0.10	
B. Air blower	!	; ;	15	2	0.25	
C. Artificial light	•	! !	200	0. 25	0, 25	

Table 7-4 (page 1 of 7)

LIFE-SUPPORT-SYSTEM LABORATORY EQUIPMENT REQUIREMENTS

Remarks	See experiment summary sheets in Appendix A.	Major microbial laboratory equipment required is given in Biomedical List.		Apollo technology.	Commercial unit. A flight type unit ahould be developed to weigh 20 lb and have a volume of 1.5 cu ft and a power of 30 to 50 W.	Development required.	Development required.	Development required.	Development required.	Commercial model miniaturized (SRT).	Available 1968.	Available.	Each battery good for 160 runs: not rechargeable.
Volume (cu ft)	1	:		0.5	12	0.005	0.01	0.05	0, 10	2.0	Unknown	0.5	0.5
Weight (1b)	:	}		12	150	-	2	3.5	'n	100	Unknown	0. 25	10
Power (W)	1	į		10	Unknown See Remarks	-	None	10	.ໝໍ	30	Unknown	2	Battery
Available Model	Experiments 4-1 to 4-58	1		NASA/MSC Unit Built by MELPAR	Bendix "Time of Flight" Mass Spectrometer	None	None	None	None	Beckman or P. E. Instrument	Lockheed USAF/AMD	Flashlight	Miniaturized model constructed by NASA ERC.
Performance Characteristics	See life support experiment summary sheets in Appendix A.	;		Frequent scanning of cabin air for organic contaminants.	Identifies trace gases not identifiable by gas chromatograph; measurements to a fraction of 1 ppm required.	Measures the pH of reclaimed water; range: 1 to 15 pH.	Measures specific conductivity of solution; range: 0 to 10,000 µmhos/cm.	Measure absorption from 3, 000 to 7, 000 Å.	Compares turbidity of solution against standards produced with SiO2; range: 0 to 150 Jackson units.	Measures trace contaminants less than 0.1 ppm.	Range, large masses: 200g to 100 kg Range, small masses: 1µg to 200 g.	Standard three-cell flashlight.	Separates particles in the 0, 2- to 10, 0- μ range into five groups and counts particles in each group.
Name/Function	l. Test specimen	2. Microbial Laboratory	3. Chemical laboratory	A. Gas chromatograph with flame detector	B. Mass spectrometer	C. pH meter	D. Electrical conductivity meter	E. Calorimeter or Coleman spectro- meter, junior model	F. Turbiditimeter	G. IR spectrophotometer nondispersive.	H. Mass-measurement device	I, Illumination device	 J. Aerosol particle analyzer

Table 7-4 (page 2 of 7)

Conclusion of Collecte and Particle sizer;		Name/Function	Performance characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
L. Wyter analysis by Pachalith of water TOC Generated Model 15 25 1.0 Eleman analysis by Pachalith of water TOC Beckman (TOC) M. Total lydrofusershoen achieves 1 pm 10% Mine safety 25 2 2 2 1.5 analyzer, dispersive achieves 1 pm 10% Mine safety 25 2 2 2 1.5 analyzer, dispersive achieves 1 pm 10% Mine safety 35 2 2 1.5 analyzer, dispersive achieves 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10% Mine safety 35 2 2 1.5 and 1 pm 10 mine safety 30 and 100	ᄶ			Miniaturized model constructed by NASA ERC.	Battery operated	10	0.5	Under development.
M. Total hydrocarbon amplication of continuous monitoring of analyzer, dispersive cabbers in any part of the said and purple to less than 1 promise, and are of aircogen with the said handpump 1 promise, and are of aircogen with the said handpump 1 promise, and are of aircogen with the said handpump 1 promise, and are of aircogen with the said handpump 20 to 70 amps at 7 to 55 V. Miss saids of persists and promises and at 7,000 ± 500 V and 20 to 70 amps at 1 to 5 V. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of combinations of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and robar. Miniaturized unit and are of aircogen or helium), agents and are of aircogen or helium, and are of aircogen or helium, and are of aircogen or helium, and are of aircogen or are are are are are are are are are ar	អ		Potability of water TOC range: 2 to 75 mg/1.	Commercial Model Beckman	15	<u>\$</u> 2	1.0	Should be miniaturized.
W. M.	×		Continuous monitoring of cabin; range: I ppm to 10%	MSA, Lira Analyzer	25	20	1.5	Could be miniaturized.
O. Ignition device spark ignition electrodes and continued	Ż		Ammonia, oxides of nitrogen, SO ₂ , sensitive to less than I ppm.	Mine safety appliances	Hand operated	2	0.1	1
P. Fire detector To detect visible and invisible affiniaturized unit a finite from the simple and invisible solution of to Pyralarm, Model Q. Fire-extinguishing Q. Diluent (mitrogen or helium), agents A. Optical pyrometer Portical pyrometer Portical pyrometer Figure measuring range; comparable to those manufactured by Millietron. Work bench for main-personnel. More than one may be used as model of various experiments. A. Log book A. Log book Fireproof paper. Radiation laboratory (See Table 7-1 and 7-3) Tether Tether Tether Tether Tether To detect visible and invisible and invisib	Ö		Spark ignition electrodes and transformers; ranges: 20 to 30 mA at 7,000 ± 500 V and 20 to 70 amps at 1 to 5 V.	Special equipment	140 to 350	20	0.1	Ignition device located in a 75-liter flask.
G. Fire-extinguishing Coly and H2O source (2 lb bothles). R. Optical pyrometer Flamemerasuring range; comparable to those personnel. More than one may be used as model of various experiments. A. Log book but floor for mainte and repair to log feet Table 7-1 and 7-2 be used as model and repair to log feet Table 7-1 and 7-2 be used as model workshop and repair to log feet Table 7-1 and 7-2 be used as model of wardows and repair to log feet Table 7-1 and 7-2 be used as model of wardows and repair to log feet Table 7-1 and 7-2 be used as model of wardows and repair to log feet Table 7-1 and 7-2 be used as model of wardows and repair to log feet Table 7-1 and 7-2 bether trings on crewman suff waist.	្		To detect visible and invisible smoke or products of combustion in cabin.	Miniaturized unit similar in principle to Pyrotronics, Inc., Pyralarm, Model No. DIS-5 5AR5B.	200	κņ	1. 0	Weight, volume and power given for predicted miniaturized unit; 500-W peak required instantaneously to activate relay.
R. Optical pyrometer glame-measuring range; Specially built unit 50 5 00.5 Work bench for main-personnel. More than one may be used, depending on allocation of namy. A. Log book Fireproof paper. Apollo Program None 5 0.5 B. Tools for mainte-foot mainte-and repair fool kits, tools, and repair and repair (See Table 7-1 and 7-2 Experiment No. M-050 (See Table 7-1 and 7-2 (See Table 7-1 and 7-	ä		Diluent (nitrogen or helium), CO ₂ and H ₂ O source (2 lb bottles).	None	None	25.5	0.1	Weight and volume are given for one experiment.
Work bench for main-tenance and repair Restraints for equipment and personnel. More than one may be used, depending on allocation of various experiments. Orbital Workshop leads as model of nany. 50 40 80 A. Log book Fireproof paper. Apollo Program None 5 0.5 B. Tools for mainte-nance and repair Hand tool kits, tools, and nance and repair Special Equipment None 18 0.6 Radiation laboratory (See Table 7-1 and 7-2) Orbital Workshop Experiment No. M-050 15 30 5 Ergometer 60-ft long, to be attached to rings on crewman suit waist. Apollo Block II Type applicable Not 5 0.2	ਲੰ		Flame-measuring range; 900° to 5,500°F.	Specially built unit comparable to those manufactured by Milletron.	50	uo-	0.5	Development required for flight-type unit.
A. Log book Fireproof paper. Apollo Program None 5 0.5 B. Tools for maintenance and repair Hand tool kits, tools, and named and repair Special Equipment None 18 0.6 Radiation laboratory (See Table 7-1 and 7-2) Orbital Workshop Experiment No. M-050 15 30 5 Ergometer 60-ft long, to be attached to rings on crewman suit waist. Apollo Block II Type applicable 5 0.2		ork bench for main- mance and repair	Restraints for equipment and personnel. More than one may be used, depending on allocation of various experiments.	Orbital Workshop Experiment No. M50B, to be used as model or one of many.	50	40	80	Part of vehicle design,
B. Tools for maintenant tool kits, tools, and special Equipment None 18 0.6 nance and repair tool tethers. Radiation laboratory (See Table 7-1 and 7-2 Experiment No. M-050 Tether follows to be attached to Apollo Block II Type Not 5 0.2 rings on crewman suit waist.	, k		Fireproof paper.	Apollo Program	None	ហ	0.5	Applied Apollo technology.
Radiation laboratory (See Table 7-1 and 7-3) Ergometer (See Table 7-1 and 7-2 Orbital Workshop 15 30 5 Experiment No. M-050 Tether 60-ft long, to be attached to Apollo Block II Type Not 5 0.2	ជាំ		Hand tool kits, tools, and tool tethers.	Special Equipment	None	18	9.0	Development required.
Ergometer (See Table 7-1 and 7-2 Orbital Workshop 15 30 5 Experiment No. M-050 Tether 60-ft long, to be attached to Apollo Block II Type Not 5 0.2		adiation laboratory	(See Table 7-1 and 7-3)		1	;	;	. 1
Tether 60-ft long, to be attached to Apollo Block II Type Not 5 0.2 rings on crewman suit waist.		rgometer	(See Table 7-1 and 7-2	Orbital Workshop Experiment No. M-050	15	30	νņ	Flight unit under develop- ment,
		ether	60-ft long, to be attached to rings on crewman suit waist.	Apollo Block II Type	Not applicable	re .	0.5	Available from Apollo and AAP.

Table 7-4 (page 3 of 7)

	Name / Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
αi	Umbilicals	60-ft long, 1 in, diam, delivers O2/hr at 3.69 ps; includes water, O2, and electrical lines.	Apollo Block II type	Not applicable	33	e.	Available from Apollo and AAP.
6 ,	9. EC/LS back-pack	Provides 3, 8-psia O2, removes CO2, and circulates cooling water to under-garment for 4 hr. Also includes control equipment.	Apollo/Hamilton Standard	240 W	4 8	4 <u>i</u> :ru	Apollo technology.
10.	Special clothing	0. 25, 0.5, 0.75 and 1.0 Clo values.	Custom-made	Not applicable	1 to 3	0,3	Existing technology.
11.	11. Space suits						
	A. Soft suit	3. 7-psia suit/5 to 7-psia cabin.	Apollo Pressure Suit	Not applicable	61	80 12.	Existing technology.
	B. Hard suit	3. 0- to 7. 0-psia suit/7-to 14. 7-psia cabin.	NASA Ames AX-I or AX-2	Not applicable	59	8.5	Adaptation of existing technology.
12.	Biomedical monitoring equipment	See Table 6-1.		;	;	;	Development required.
13.	Metabolic measuring device	Register 0 to 6, 61 at 0 to 80 1/min.; pO ₂ : 0 to 2, 5 psi; CO ₂ : 0 to 1. 5 psi; core tem- persture sensitivity: 98 to 101°F; skin: 50° to 115°F; EKG: 300 to 5, 000 µ V; band width: 0, 5 to 50 H _Z .	(See Table 7-2)	None	4	-	Development required.
7	14. Leak detector	Gas: 1 to 10 x 10 ⁻⁶ lb/hr at 50 to 150 psig. Idquid: 1 to 10 x 10 ⁻⁴ lb/hr at 20 to 60 psig.	None	κ	រហ	4.0	One method proposed: monitoring usage rate.
15.	Wattmeter	Ranges: 0 to 100, 0 to 500, 0 to 1,000, 0 to 3,000 W. Accuracy ±1% of full scale.	Specially miniaturized equipment	None	0,5	0.1	Development required.
16.	Voltmeter	Ranges: 0 to 10, 0 to 50, 0 to 1000 to 100, 0 to 200, 0 to 500 V. Accuracy ±1% of full scale.	Specially miniaturized equipment	None	0.5	0.1	Development required.
17.	17. Ammeter	Ranges: 0 to 1,000 mA and 0 to 5, 0 to 10, 0 to 50 amps. Accuracy ±1% of full scale.	Specially ministurized equipment	None	0.5	0.1	Development required,

Table 7-4 (page 4 of 7)

Name / Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
18. Pressure sensor	Gas: 0 to 0. 1 psia to 0 to 200 psia Liquid: 0 in. Hg to 30 in. Hg up to p to 5,000 psig.	Rosemount PN 830A Robinson-Halpern PN P61	1 1	0.77	0.01	Applied Apollo tech- nology.
19. Temperature sensor	Gas: 20° to 90°F.	Apollo PN 836950	<u>{</u>	0.2	0.001	Requires amplifier: Apollo PN 836050, 0.3 1b, 0.005 cu ft.
	-445° to 500°F.	Rosemount PN 150MA	:	0. 25	0, 001	Requires amplifier: Rosemount 510AF, 0.55 1b, 0.004 cu ft.
	Liquid: -50° to 100°F.	Apollo PN 836058	1	9.4	0,002	Requires amplifier: PN 836056 (above).
	-445° to 930°F.	Rosemount PN 177MA	;	0.25	0, 001	Requires amplifier: PN510AF (above).
20. Flowmeter	Gas Flow: 0.04 to 0.3 lb/hr	Apollo Series 5000	None	0, 38	0,0025	Requires converter,
	up to 0.3 to 3.0 to/nr Gas Flow. 0.36 to 6.4 lb/hr	Apollo Series 5000	None	0.165	0.015	which weighs 0.52 lb,
	up to 10 to 100 10/11. Liquid Flow: 50 to 375 lb/hr up to 300 to 4, 750 lb/hr	Apollo Series 5000	None	0, 38	0,0025	
21. Humidity sensor	1 to 1,000 ppm by volume range, equivalent to -101 to -50F. Accuracy: ±5% of full scale.	CEC type 26-350	10	35	eri	Commercial unit, should be made flight type to weigh about 2 to 3 lb; power approximately 1 to 2W.
22. Dew point meter	-100° to 0°F range	Cambridge Model 92	088	09	4	Commercial unit (refrigeration included)
	0° to 100°F range	Technology, Inc., Model 707 Thermo- electric	120	18	0.5	Commercial unit, Flight- type unit should weigh about 10 lb and require less than 30 W when developed.
23. Timer	0 to 24 hr; 0.01-sec accuracy sweep second hand; time out; portable; capable of hand or electronic pulse control.	MDA C-COMPARE	1 1 1	1.0	0.05	Capable of intermittent use without reset.
24. Movie camera	1 to 200 fps; single frame exposure; range finder; 16 mm or larger; zoom, wide angle, and telephoto lens; cartridge load; automatic exposure selectric; pop-up external viewfinder; film remaining indicator.	Bolex	200	3.0	0.2	All controls should be operable in the pressurized pressure garment assembly (PGA). Light source must be included.

Table 7-4 (page 5 of 7)

	Name / Function	Performance Characteristics	Available Model	Power (W)	Weight (1b)	Volume (cu ft)	Remarks
25.	Still camera	35 mm or larger (single-lens reflex); automatic exposure select, wide-angle, telephoto and close-up lenses; one-hand operation.	Hasselblad	None	1.5	0.2	Include strobe-type flash and variety of film: color, black and white, high- contrast, high-speed, etc.
26.	Pumps	Liquid Pump: 200 lb/hr at specific gravity = 1:100°F temperature and 30-psi pressure rise. Liquid/solid pump: flow rate 20 to 200 lb/hr.	Apollo PN 850024-5	36	2 10	0, 01	Adaptation of Apollo technology.
27.	Fans	30 to 150 cfm at 7 psia and 0.4 to 0.15 in. H ₂ O pressure rise.	Apollo PN 826010-2 and 826070-3	19	1.5	0.01	Adaptation of Apollo technology.
28.	Blowers	20 to 35 cfm at 7 psia, 10- to 12-in. H ₂ O pressure rise.	Apollo PN 826000-2	82	κ	0.10	Adaptation of Apollo technology.
29.	Accelerometer	Measurement of small accelerations of the vehicle, on-board crewman and masses,	None	Not applicable	4	9.0	Applied Apollo technology.
30.	30. Velocity meter	0 to 50 fps gas velocity, accuracy $\pm 1\%$ of full scale.	Heated thermister anemometer		0.1	0,001	Development required.
31.	Liquid gaging unit	0 to 1, 0 to 10 and 0 to 200 lb/hr at specific gravity = 1, accuracy $\pm 1\%$ of full scale.	Rotary displacement meter	None	-	0.01	Development required.
32.	Radioisotope shielding	Included as part of test specimen and/or power unit.	1	}	}	. j . j . j	50 rem maximum total dosage allowable to crewmen.
33,	Gas compressor	Multistage compressor with a pressure range of 0 to 3,000 psi.	1	200	20	1.0	Development required.
34.	Accumulator	Spherical stainless-steel tanks	1	None	2	1.0	Applied Apollo-Gemini technology.
35.	Holding tanks	Two bladder tanks, nitrogen or helium pressurizing fluid, l-cu ft capacity each,	1	None	ស	2.0	1
36.	Special space radiator	Required for such experiments as the reversed Brayton cycle reliquefaction system. Thermal energy to be rejected about 500 W.	-	20	30	1	Liquid/vapor problem in null gravity. Radiator area32 sq ft.

Table 7-4 (page 6 of 7)

				2000	Weight	Waliana	
	Name/Function	Performance Characteristics	Available Model	(W)	(Ib)	(cn ft)	Remarks
37.	Cryogenic supply	5-1b capacity, liquid nitrogen, special insulation, liquid delivery of cryogen.	i i	, ro	2	1. 0	Applied AAP technology.
38.	CO ₂ sensor	0 to 20-mm Hg range, accuracy ±1% of full scale.	Beckman CO ₂ partial pressure sensor	 .	2	0.1	Available technology.
39.	Isotope heater	Operating temperature ranges: $200^{\circ} \pm 20^{\circ} F$, $600^{\circ} \pm 50^{\circ} F$ and $1,400^{\circ} \pm 200^{\circ} F$; energy levels: 100 ± 5 thermal watts per capsule.	AEC, Division of Isotopes Development	1	80/kW	0, 2/kW	Approximate weight and volume based on use of 238 PuO ₂ isotope. Power System waste heat is required.
40.	Emergency backup EC/LS	(See experiment requirements summary for Experiments 4-1 and 4-2 in Appendix A.)	Apollo, Gemini, etc.	.) .)	}	1 2 1	Must be flight-qualified hardware.
41.	Reflectometer	Reflectance ρ_{λ} 0.8 accuracy = $\pm 3^{\sigma}$ = $\pm 3\%$ Reflectance ρ_{λ} 0.1 accuracy = $\pm 1^{\sigma}$ = $\pm 1\%$ Wavelength $\rho < 2\mu$;	0 50	10	s. 'o	Development of a miniaturized spaceflight unit is required. Assumed weight volume and power are for flight equipment.
4.2.	Optical density sensor	Optically measures density of cryogenic fluid. Device includes a light beam source and a means of measuring absorption of light passing through the cryogenic fluid. Range of absorption: 0.0001 to 1.0, with accuracy of ±0.1%.	1	20	70	0 .	A special flight-type unit must be developed.
43.	Test container with bladder	Stainless-steel cryogenic container, bladder compatible with cryogenic temperatures. Pressurizing inert gas and gas source included.	;	None	1.5	0.5	Hardware feasibility based on AAP technology. Additional development required.
44.	Pressure controller	Inlet pressure: 150 to 1,000 psig. flow: 10 to 100 cc/min. of nitrogen.	;	None	1.0	0.01	Based on Apollo tech- nology.
45.	Plastic squeeze container	Container volume 0, 5 cu ft. pressurization requirement = 20 psig ±2 psi.		None	0.2	0.5	Commercial-type units, if available, may be modified and used for this purpose.
46.	46. Heating element	Electrical heating element, supplied with a variac to control the electric energy input to the field.	1	500	1.5	0.5	Modified commercial hardware may be used in developing this unit.

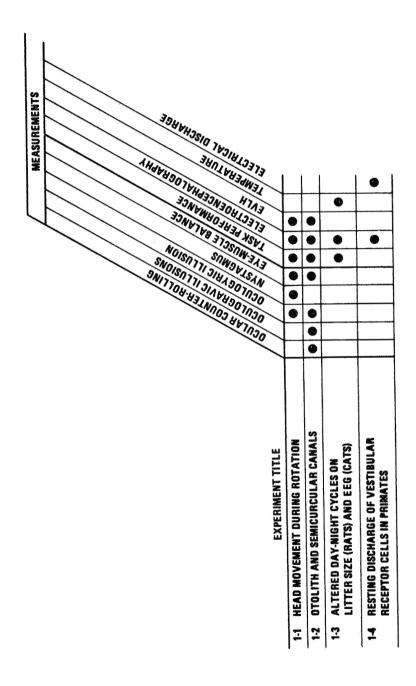
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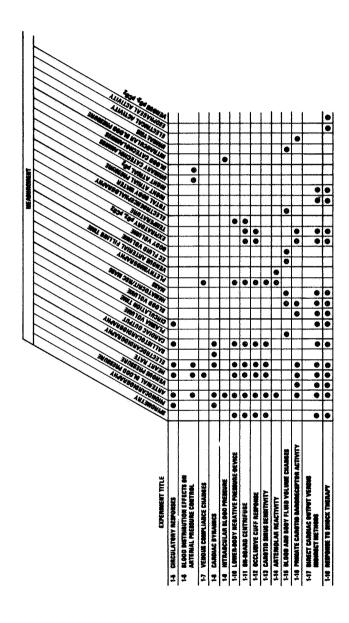
	Name / Function	Performance Characteristics	Available Model	Power (W)	Weight (Ib)	Volume (cu ft)	Remarks
47.	Orientation device	Artificial horizon gyroscope, with readout showing position error in degrees, accuracy ±10.	1	50	ĸ	0.1	A modified Gemini artificial horizon may be used for this purpose.
8.	Photo cell	Photocell-amplifier-relay assembly, 500 ± 10 counts per minute, 2 in. object interrupting light meter.	†	10 to 500	 R	0.05	Power of 500 W needed intermittently to energize relay. Unit assumed to be miniaturized version of presently available devices (example Worner Models M-26, M-36 and 67).
49.	Oscilloscope	Laboratory-type 5-in. oscilloscope built-in voltage regulator, recurrent sweep. Narrow-band sensitivity 10 mV/in rms; rise time: 0.07 ± 0.02 sec.	Requires miniaturization of commercially available units.	25	10	0.1	Assumed weight, volume, and power for flight-type unit. (Commercial hardware Hickok model 770A laboratory oscilloscope.)
50.	Comfort simulator	Specially constructed two-man chamber; men placed on net hammocks, blower circulates air from plenum through diffusers to chamber and returns through double walls; temperature-control switches for each subject.	MDAC design	200	150	160	Development for flight- type qualification is required.
51.	Pretreatment unit	A chemical-solution dispenser dispenses a preset amount (1. 0 cc ±5% when actuated. Includes a mechanical counter.	MDAC design	None	1.5	0.2	Unit dry weight given, requires development for flight-type qualification.
52.	Debris disposal	Incinerator utilizes electrical energy for ignition; unit capacity 1. 0 lb.	Design by General American Research Division, Niles, Ill.	300	ις	1.5	Prototypes of similar units have been made. Development of unit is required.
53.	Vacuum cleaner	Battery-operated conventional unit.	Design by Whirlpool Corp., Systems Division, St. Joseph, Mich.	ហ	м		Power given for recharging batteries; unit needs development for qualification.
4.	Solid/gas separator	Separates solid particles from gas stream. Uses centrifugal action; gas exits through center, solids go to circumference. Flow: 0 to 50 cfm of gas; particle sizes 10 to 1/4-in.	1	20	1. 0	0.01	Similar to centrifugal gas/liquid separators but requires development regarding disposal of solids from periphery of unit.

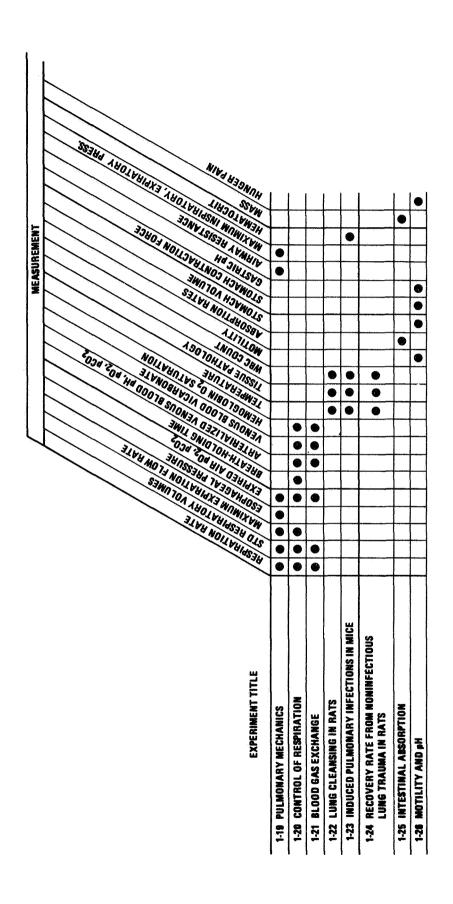
Appendix MATRIX CHARTS

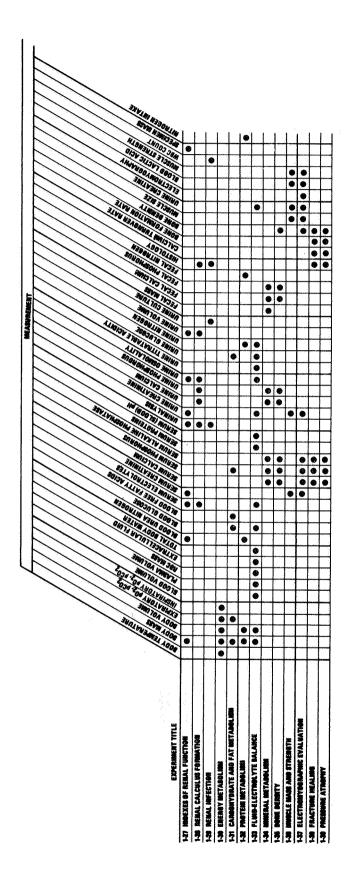
A-1	Biomedical Experiments and Measurements	139
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BIOMEDICAL EXPERIMENTS AND MEASUREMENTS









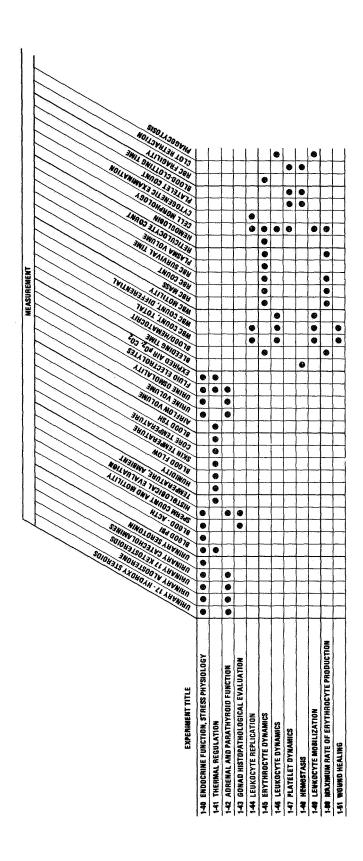


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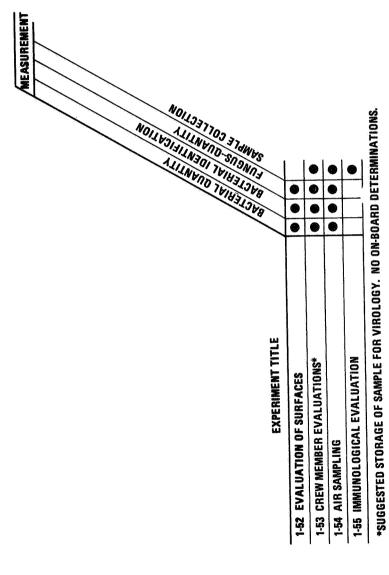
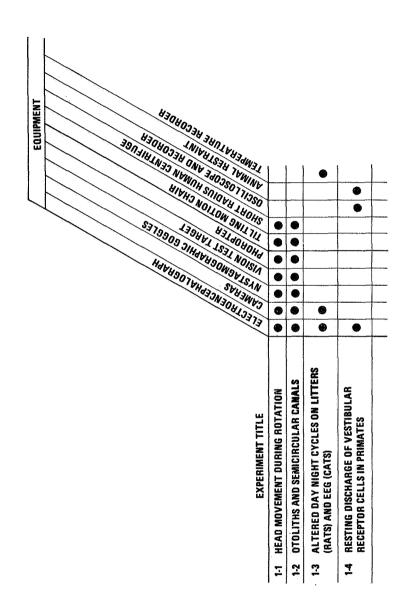
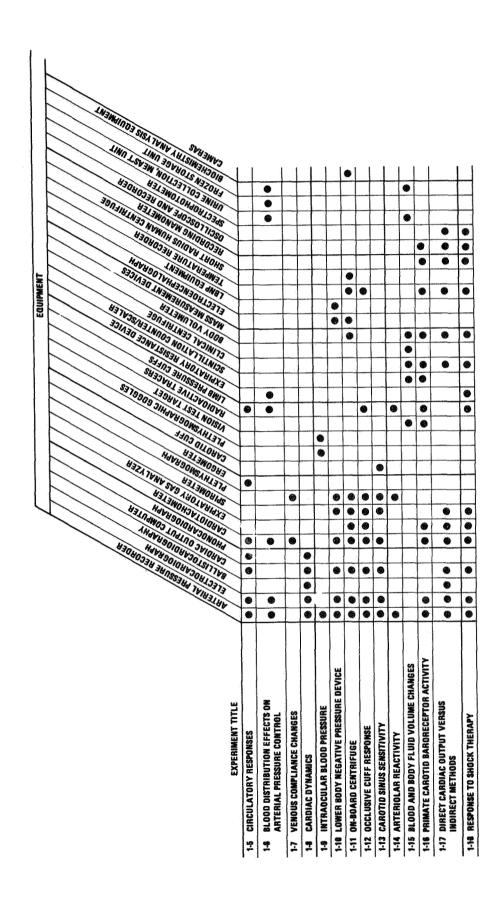
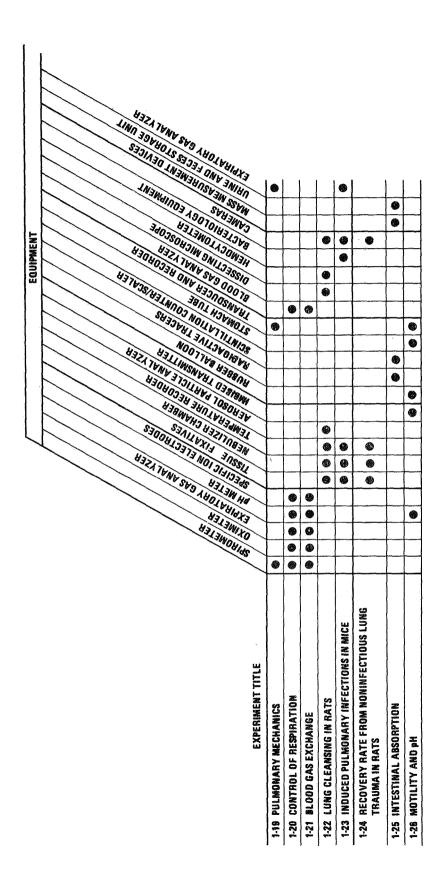
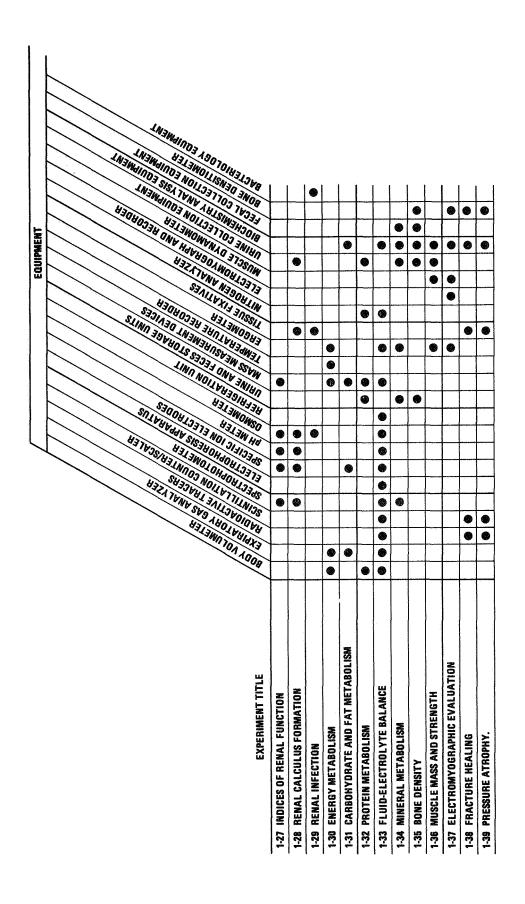


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BIOMEDICAL RESEARCH EQUIPMENT









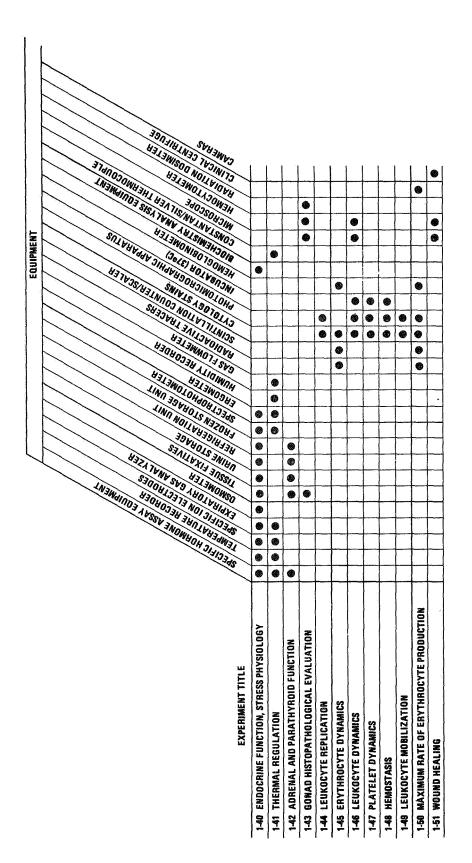
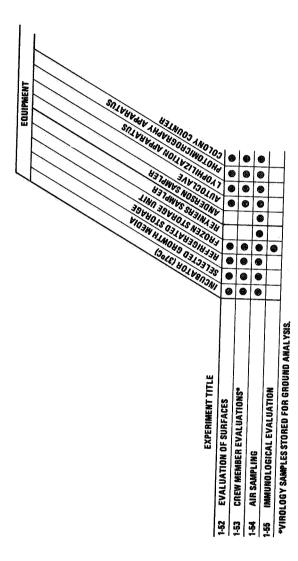


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2-2 RESTRAINT AND FINE-FORCE GENERATION
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2-4 PERSONNEL TRANSLATION
2-5 MASS TRANSLATION
2-5 MASS TRANSLATION
2-7 COGNEUMICATIONS AND RECORDING
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Table A-4
BEHAVIORAL RESEARCH EQUIPMENT

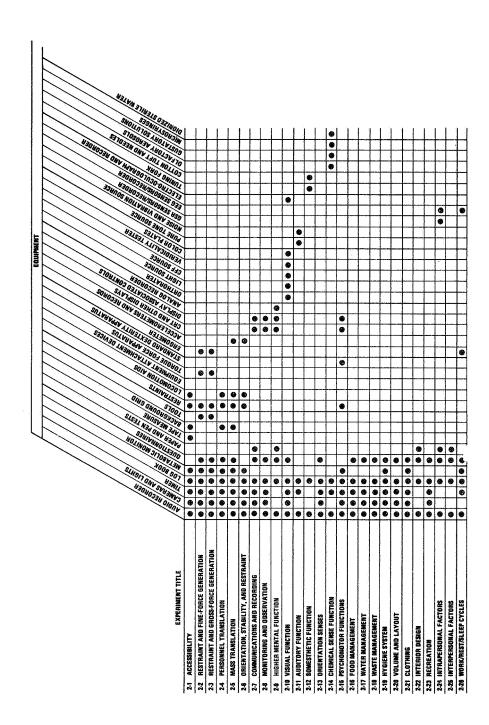


Table A-5 BIOSCIENCE EXPERIMENTS AND MEASUREMENTS

Table A-6 (page 1 of 2) BIOSCIENCE RESEARCH EQUIPMENT

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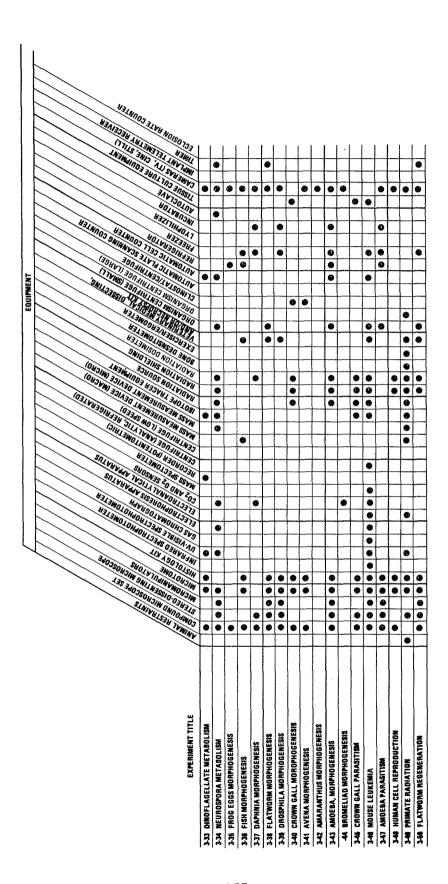


Table A-7 (page 1 of 2)
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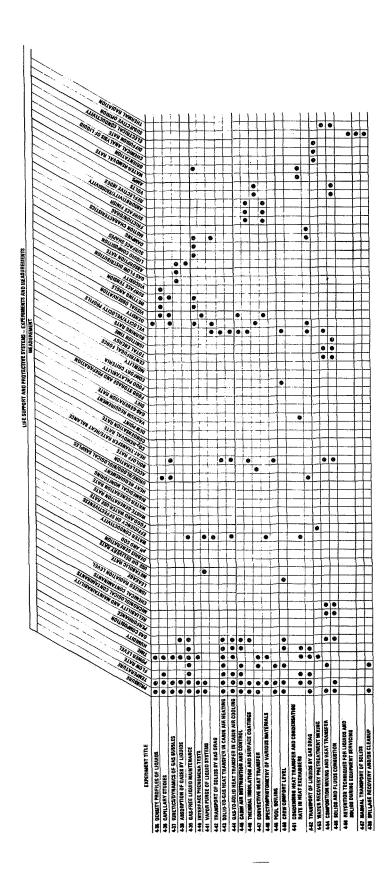
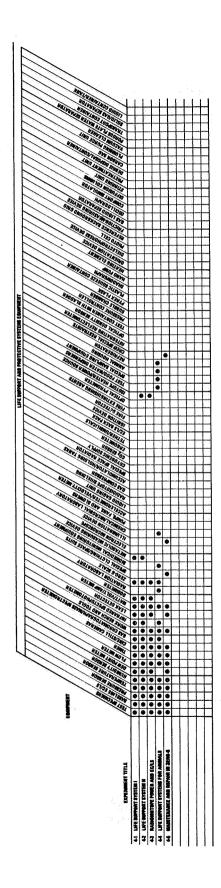
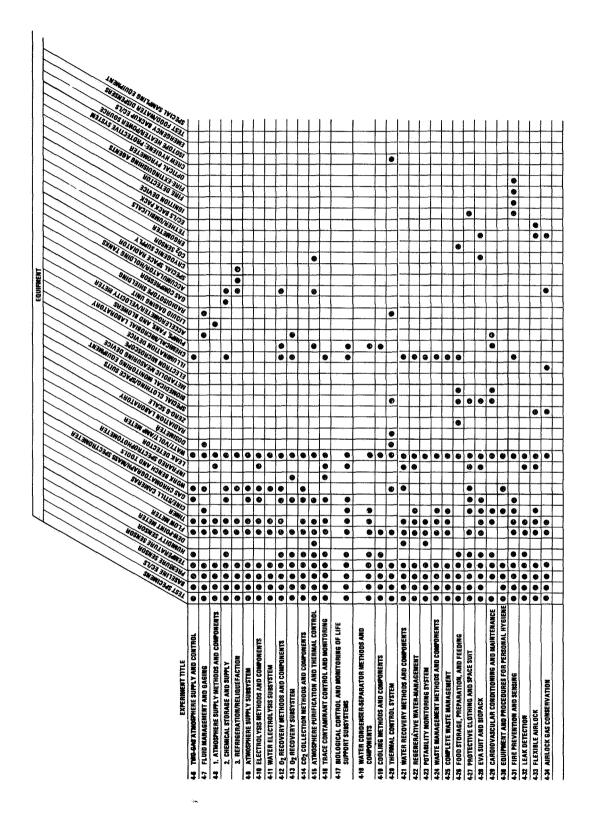
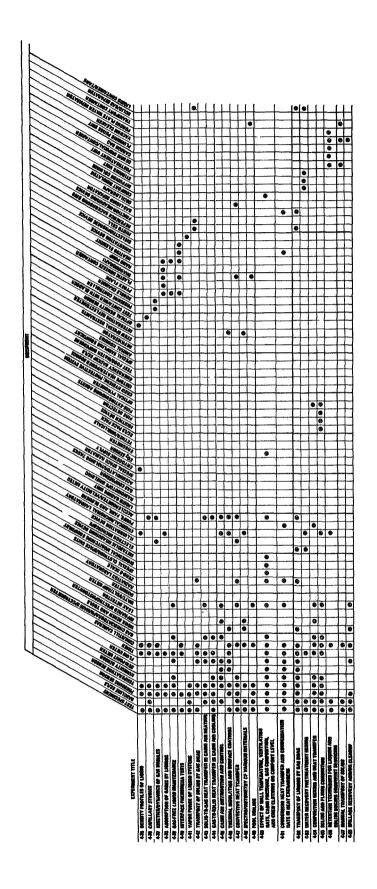


Table A-8 (page 1 of 3) LIFE SUPPORT AND PROTECTIVE SYSTEMS EQUIPMENT







FIRST CLASS MAIL



POSTMASTER:

If Undeliverable (Section Postal Manual) Do Nor Re

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-National Aeronautics and Space Act of 1958

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